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BOTTOM REFLECTION OF UNDERWATER
EXPLOSION SHOCK WAVES, COMPUTER
PROGRAM

By
James R. Britt
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30 JULY 1971

NOL

NAVAL ORDNANCE LABORATORY, WHITE OAK, SILVER SPRING, MARYLAND

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Explosions Research Department
Naval Ordnance Laboratory
White Oak, Silver Spring, Maryland

NOLTR 71-110

30 July 1971

BOTTOM REFLECTION OF UNDERWATER EXPLOSION SHOCK WAVES, COMPUTER PROGRAM

This report is part of a continuing study of the interaction of the underwater explosion shock wave with the ocean bottom. The computer program described in this paper calculates the bottom reflection and generates plots of the pressure history. The calculations of this program are being used in the bottom reflection study to assess the potential danger to ships delivering nuclear underwater weapons posed by various bottom materials.

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ROBERT ENNIS
Captain, USN
Commander

(Signature)

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By direction

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BOTTOM REFLECTION OF UNDERWATER
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1. INTRODUCTION

The bottom reflection of the underwater explosion shock wave is of interest to the Navy because of the danger it presents for self-damage to ships delivering nuclear ASW weapons. The theory presently being used to describe the reflection is a linear spherical wave theory originally developed by L. Cagniard (1) for the calculation of the reflection at an interface between two elastic solids. On the basis of Cagniard's theory, Rosenbaum (2) derived equations which describe the bottom reflection of underwater explosion shock waves. Britt (3) has greatly extended and generalized Rosenbaum's work. Britt's report should be consulted when using the computer program described here.

This report describes a computer program, BOTREF, written in FORTRAN IV for the NOL CDC 6400 computer. The code calculates the pressure history of the bottom reflection of incident exponential pulses reflected from plane, homogeneous, elastic bottoms using the spherical wave theory. Major portions of this program were written by the second author. The first author later brought this program into its present versatile form and used it successfully in practical applications.

The program has options for calculating the spherical wave reflection in two ways: (1) using real arithmetic and equations derived using contour integration (referred to as the Cagniard-Rosenbaum method) and (2) using the "complex arithmetic method". The first method is generally faster, but both usually take less than 30 seconds of central processor time on the CDC 6400 for calculating a complete pressure history. Also included in the program is an option for calculating the bottom reflection using the plane wave theory of Arons and Yennie (4). For both the plane wave and the spherical wave, the calculations include corrections for the non-linear changes of the shock wave peak pressure and time constant with the distance from the charge.

The code generates a CALCOMP plot tape of the total pressure history including the incident, bottom reflected, and acoustic surface reflected waves. The print out, in addition to the pressure history, includes information such as the incident angle, the plane wave reflection coefficient and phase shift, critical angles, arrival times, impulses, and energy flux.

The output of the bottom reflection program can be directly transferred to the PTV Program (NOLTR 71-65) which is then used as a subroutine. This program calculates the peak translational velocity (PTV) of a cylindrical target. This velocity can be used as an index for damage.

The equations used in the BOTREF code are described in Section 2 and references are made as to the location in the program where each equation is used. In Section 3 a detailed description is given of the program organization, inputs, outputs, and other important

symbols. The appendices contain a complete FORTRAN listing of the program, sample output, and a CALCOMP plot.

The code contains many comment cards so that most of the inputs and outputs and much of the organization is explained in the program listing.

Comments on Terminology. In the acoustic literature reflectors are called either solids or fluids, depending on whether they have a shear-strength or not. We prefer the terms non-rigid or rigid, because some solids, for instance, sand, have such a low shear strength that the theory for a non-rigid bottom yields sufficiently accurate results, in spite of the fact that the material is a solid. We hope that our terminology will lead to less misunderstandings than the conventional one or the previously used term "liquid bottom".

Rigidity should be understood as the resistance of a body to a change in shape at constant volume. It is equivalent to shear strength and is measured either by the Poisson Ratio or, as in this paper, by the propagation velocity of the shear wave. The shear velocity is zero for a non-rigid material. Compressibility is the resistance to a change in volume at constant shape and can be represented by the propagation velocity of a compression wave, i.e., the sound velocity.

The word rigid often has the connotation of a material having infinite rigidity. We use it in the sense of a material having a finite, non-vanishing rigidity.

2. THEORY USED TO CALCULATE THE BOTTOM REFLECTION

2.1 Theory of the Bottom Reflection of a Spherical Wave

The theory used in the computer program described in this report has been derived by Rosenbaum (2). Britt (3) has reviewed, explained, and greatly extended Rosenbaum's work. A semi-linear theory is used which describes all phenomena of interest with adequate accuracy. The notation used in this section is essentially that of Britt's report. The following exceptions are to be noted. We denote the excess pressure by p instead of P . Britt and Rosenbaum denote the time by τ ; we use t for the time and τ_m for Rosenbaum's reduced time (compare with Equation (2.2.2)). The program calculates the step wave response $P_m = \tau_m P_1$ which corresponds to one reflection from the bottom. Multiple reflections between the surface and the bottom are not included. (Multiple reflections are of minor importance to underwater explosion phenomena that lead to damage processes. When a strong pressure wave is reflected at the water surface, most of the wave energy is left near the surface and does not propagate down into the water because of cavitation and spray formation.)

We denote $\tau_m P_1$ by P_r , the bottom reflection slant range $\tau_m R_1$ by R_r , the incident or direct wave range by R_i , and the surface reflection range by R_g . We also drop the subscripts n and m except in τ_m and K_m (Equation 2.2.18).

The geometry of the bottom reflection is shown in Figure 1. The water depth is H . The depths of the charge and gauge are d and d_g . The horizontal distance between charge and gauge is r . The incident angle of the bottom reflection is θ . From this figure

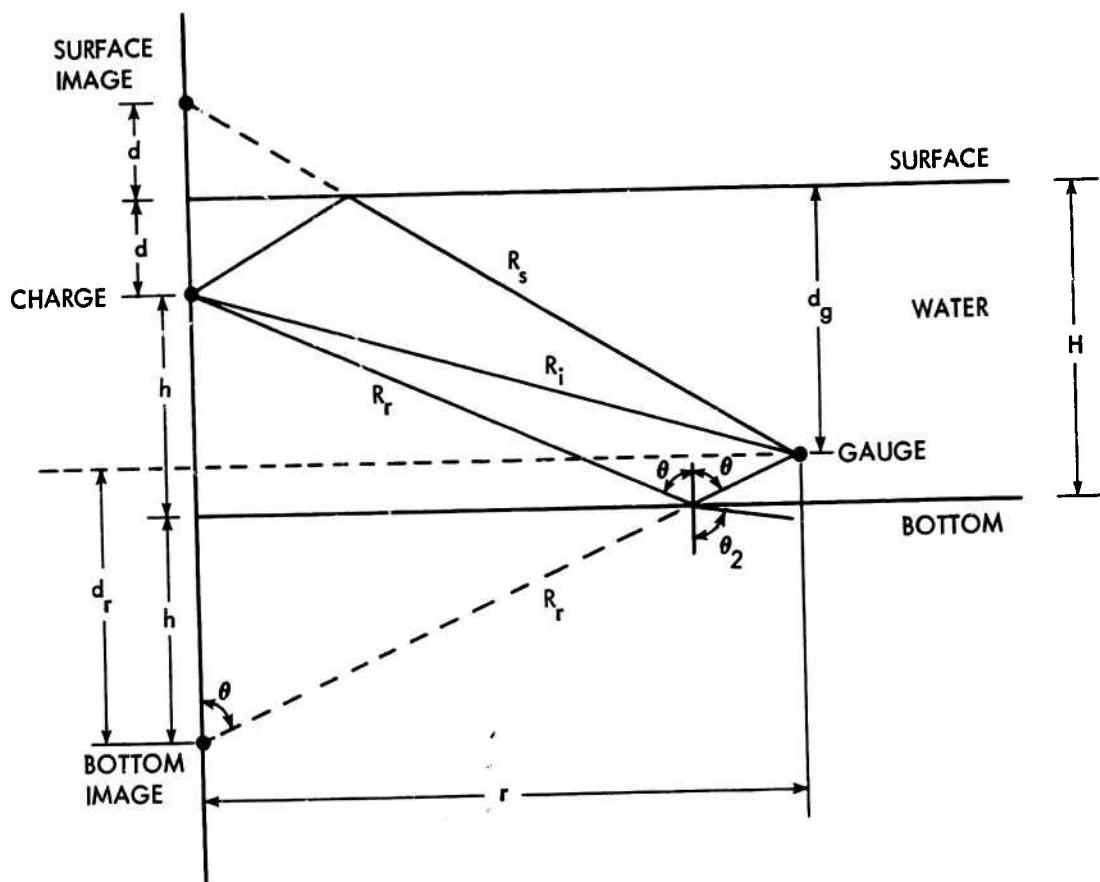


FIG. 1 BOTTOM REFLECTION GEOMETRY

we see that the slant ranges are given by the equations

$$R_i = \left[(d - d_g)^2 + r^2 \right]^{1/2} \quad (\text{slant range of incident wave}) \quad (2.1.1)$$

$$R_r = \left[d_r^2 + r^2 \right]^{1/2} \quad (\text{slant range of wave reflected at bottom}) \quad (2.1.2)$$

and

$$R_s = \left[(d_g + d)^2 + r^2 \right]^{1/2} \quad (\text{slant range of the wave reflected at the water surface}), \quad (2.1.3)$$

where $d_r = 2H - d_g - d$ is the depth of the "image" below the gauge.

Further, we have

$$\cos \theta = d_r/R_r \quad (2.1.4)$$

and $\sin \theta = r/R_r. \quad (2.1.5)$

In the water the sound velocity is denoted by c_1 , and the density by ρ_1 . Similarly, the sound velocity in the bottom material is c_3 , the shear wave propagation velocity is c_4 , and the density is ρ_3 . (Britt denoted the sound velocity and density of a rigid bottom by c_3 and ρ_3 .)

2.1.1 Critical Angles. For an incident angle θ , which is also the reflected angle, the refracted or transmitted ray into the bottom makes an angle θ_3 (see Figure 1) given by Snell's law

$$\sin \theta = \frac{c_1}{c_3} \sin \theta_3. \quad (2.1.6)$$

The angle θ_3 is that angle at which the pressure wave enters the bottom. Similarly, the angle θ_4 of the shear wave in the bottom is defined

$$\sin \theta = \frac{c_1}{c_4} \sin \theta_4. \quad (2.1.7)$$

When $c_3 > c_1$ or $c_4 > c_1$ the angles θ_3 and θ_4 become 90° at incident angles θ_{cr} and θ_{crs} defined by

$$\sin \theta_{cr} = c_1/c_3 \quad (2.1.8)$$

$$\sin \theta_{crs} = c_1/c_4. \quad (2.1.9)$$

θ_{cr} is called the critical angle of the compression wave, and θ_{crs} is called the critical angle of the shear wave. These angles are important for calculating and interpreting the bottom reflection pressure history.

2.1.2 The Incident Pulse. The computer program assumes an exponential incident pulse $p_i(t)$ given by

$$\begin{aligned} p_i(t) &= p_F(R_i) \exp \left[-(t - R_i/c_1)/G \right] \quad \text{for } t \geq R_i/c_1 \\ p_i(t) &= 0 \quad \text{for } t < R_i/c_1, \end{aligned} \quad (2.1.10)$$

where G is the time constant (usually denoted by τ) and $p_F = p_F(R_i)$ is the peak pressure of the incident shock wave. A reduced notation is used in the machine program utilizing the incident slant range R_i (Equation 2.1.1) as the characteristic length. The reduced time is $\bar{t} = t c_1 / R_i$. (It is denoted by T in the program). The reduced arrival time of the front of the direct wave is thus $\bar{t} = 1$. The incident pulse is then given by

$$\begin{aligned} p_i(\bar{t}) &= p_F(R_i) \exp \left[-(\bar{t} - 1)/\bar{G} \right] \quad \text{for } \bar{t} \geq 1 \\ p_i(\bar{t}) &= 0 \quad \text{for } \bar{t} < 1, \end{aligned} \quad (2.1.11)$$

where $\bar{G} = c_1 G / R_i$ is the reduced time constant.

For the time constant G and the peak pressure p_F the relations for the actual underwater explosion shock waves (high amplitude waves) are used which when used together with the wave equation comprise the "semi-linear" theory. The shock wave parameters are obtained from the similitude equations

$$G = C_G W^{1/3} (W^{1/3}/R_i)^{n_G} \quad (2.1.12)$$

$$p_F = C_p (W^{1/3}/R_i)^{n_p}, \quad (2.1.13)$$

where C_G , C_p , n_G , and n_p are constants for a given explosive. W is the charge weight in pounds, or, with appropriate constants, the yield in kilotons. G and p_F are calculated in the main program in Cards BOTR160-167.

Examples of the constants are

Explosive	C_p	n_p	C_G	n_G
TNT	21600	1.13	0.052	-0.23
HBX-1	23800	1.15	0.049	-0.29
Nuclear (W = Yield in kt)	$4.291 \cdot 10^8$	1.13	2.242	-0.22
	$4.380 \cdot 10^8$	1.13	2.274	-0.22

The values for nuclear explosions of the upper row are the most recent ones. Those in the lower row are generally quoted in the literature. The constants C_p and C_G are given in psi and milliseconds.

2.1.3 The Surface Reflection. The surface reflection $p_s(t)$ calculated from the simple acoustic equation is

$$p_s(t) = -p_F(R_s) \exp \left[-(t - R_s/c_1)/G_s \right] \quad \text{for } t \geq R_s/c_1 \quad (2.1.14)$$

$$p_s(t) = 0 \quad \text{for } t < R_s/c_1,$$

where $G_s = G(R_s)$. In reduced notation this becomes

$$p_s(\bar{t}) = -p_F(R_s) \exp \left[-(\bar{t} - \bar{R}_s)/\bar{G}_s \right] \quad \text{for } \bar{t} \geq \bar{R}_s$$

$$p_s(\bar{t}) = 0 \quad \text{for } \bar{t} < \bar{R}_s, \quad (2.1.15)$$

where $\bar{G}_s = c_1 G_s / R_i$ and $\bar{R}_s = R_s / R_i$. These equations are coded in Cards BOTR218, 704, and 882.

The surface reflection is a tension wave and its pressure is to be subtracted from the pressure of the incident and the bottom reflected wave.

Equations (2.1.14 and 15) ignore cavitation which in sea water does not let pressures drop substantially below the vapor pressure. In the machine program this is taken into account by a test that makes sure that the total pressure does not fall below zero absolute (Cards BOTR713 and 884).

For a very oblique incidence the acoustic treatment of the surface reflection breaks down and must be replaced by the anomalous surface reflection described in NOLTR 70-31. The machine program described here does not include this mode of the surface reflection. This problem will be treated in another machine program that describes the shock wave propagation in shallow water.

2.1.4 The Convolution Integral. The theory of the bottom reflection yields the reflected wave for an incident step wave. This step wave response, denoted by $p_r(t)$, is the crucial point of the analysis and will be discussed in detail later. It has the dimension of $(\text{Length})^{-1}$. The pressure history of the bottom reflected wave for an exponential incident wave $p_i(t)$ is obtained from the convolution integral:

$$p_r(t) = p_F'(R_r) \left[p_r(t) - 1/G_r \int_{\delta}^t \exp[-(t-z)/G_r] p_r(z) dz \right]$$

for $t \geq \delta$
 $p_r(t) = 0$ (2.1.16)
 for $t < \delta$.

This equation is explained in Appendix D of Britt's report. The scale factor p_F' and the time constant G_r are given by

$$p_F' = R_r p_F(R_r) \quad (2.1.17)$$

$$G_r = G(R_r), \quad (2.1.18)$$

where R_r is the slant range of the reflected wave, Equation (2.1.2). The factor R_r of the reduced pressure scale factor p_F' stems from the definition of the reduced step wave response $p_r(t)$ which includes R_r^{-1} as a factor.

The reduced form of the convolution integral is readily obtained by introduction of $\bar{t} = t c_1 / R_i$, $\bar{\delta}$, \bar{z} , and $\bar{G}_r = c_1 G_r / R_i$.

The symbol δ in Equation (2.1.16) denotes the arrival time of the bottom reflection.

For subcritical incidence, $\theta < \theta_{cr}$, we have

$$\delta = t_c = R_r / c_1 \quad (2.1.19)$$

and the reduced form is

$$\bar{\delta} = c_1 \delta / R_i = R_r / R_i. \quad (2.1.20)$$

In this case δ is the arrival time t_c of the peak of the reflected wave.

For supercritical incidence, $\theta > \theta_{cr}$, the precursor of the bottom reflection arrives before $t = t_c$, namely at

$$\delta = r/c_2 + d_r (c_1^{-2} - c_2^{-2})^{1/2} \quad (2.1.21)$$

or in the dimensionless form

$$\bar{\delta} = r c_1 / c_2 R_i + d_r \left[1 - (c_1 / c_2)^2 \right]^{1/2} / R_i. \quad (2.1.22)$$

The convolution integral is calculated in the BOTREF program Cards BOTR556, 589, 597, 635, 643, and 673 using Simpson's rule for small intervals with three equally spaced points.

For an exponential incident pulse the integral need not be recalculated from $t = \delta$ for each time step because of

$$\exp(t + \Delta t) = \exp(t) \exp(\Delta t).$$

The algorithm used to calculate the integral in Equation (2.1.16), which we call F_I , is as follows:

$$F_I(t) = \exp(-2\Delta t/G_r) F_I(t - 2\Delta t) + \{ \left[P_r(t - 2\Delta t) \exp(-\Delta t/G_r) + 4P_r(t - \Delta t) \right] \exp(-\Delta t/G_r) + P_r(t) \} \Delta t / 3. \quad (2.1.23)$$

This relation permits a convenient step-by-step quadrature of the integral using its value for a time $2\Delta t$ earlier. The expression is readily transformed into a reduced form by the introduction of \bar{t} , $\Delta\bar{t}$, and \bar{G}_r . F_I has the dimension time/length.

For supercritical incidence $P_r(t)$ has a logarithmic singularity at $t = t_c$. Since $P_r(t)$ is a rapidly changing function of t near t_c , a smaller time increment, $\Delta t' \approx \Delta t/8$, is used in the code for the interval $(t_c - \alpha\Delta t, t_c + 4\Delta t)$. The code calculates α so that there are enough points in the bottom reflection pressure-time history (before and after the time increment change) to execute the impulse

and energy flux integrations. The usual range is $2.1 < \alpha < 6.1$. Because these integrations are performed with Simpson's rule on equally spaced points, each integration step is completed on an odd-numbered point.

Further, in the time range $t_c - 2\Delta t' < t < t_c + 2\Delta t'$ we change the integration variable in the convolution integral F_I to

$$\begin{aligned} v^2 &= t_c^2 - z^2 & \text{for } z \leq t_c \\ u^2 &= z^2 - t_c^2 & \text{for } z \geq t_c. \end{aligned}$$

The step wave response $P_r(t)$ behaves near the singularity like

$$\lim_{t \rightarrow t_c} P_r(t) = C \ln(|t_c - t|).$$

The change of variables v and u transforms the last two factors of Equation (2.1.16) as follows:

$$\begin{aligned} P_r(z) dz &= -\frac{v}{z} P_r(z) dv & z \leq t_c \\ &= \frac{u}{z} P_r(z) du & z \geq t_c. \end{aligned}$$

Then we obtain

$$\lim_{z \rightarrow t_c} -\frac{v}{z} P_r(z) = -C \lim_{z \rightarrow t_c} \frac{(t_c^2 - z^2)^{1/2}}{z} \ln(t_c - z) = 0 \quad z \leq t_c$$

$$\lim_{z \rightarrow t_c} \frac{u}{z} P_r(z) = C \lim_{z \rightarrow t_c} \frac{(z^2 - t_c^2)^{1/2}}{z} \ln(z - t_c) = 0 \quad z \geq t_c.$$

This means the integrands vanish at the singularity of P_r , and thus makes numerical integration possible.

Equation (2.1.24) below illustrates the variable change.

$$\begin{aligned}
 p_r(t) = & p_F' \left[p_r(t) - \frac{1}{G_r} \int_{t_c-2\Delta t'}^t \exp \left[-(t-z)/G_r \right] p_r(z) dz \right. \\
 & - \int_{v(t_c)}^{v(t_c-2\Delta t')} \exp \left[-(t-z)/G_r \right] p_r(z) \frac{v}{z} dv \\
 & + \int_{u(t_c)}^{u(t_c+2\Delta t')} \exp \left[-(t-z)/G_r \right] p_r(z) \frac{u}{z} du \\
 & \left. + \int_{t_c+2\Delta t'}^t \exp \left[-(t-z)/G_r \right] p_r(z) dz \right]. \tag{2.1.24}
 \end{aligned}$$

Up to time $t_c - 2\Delta t'$ and after time $t_c + 2\Delta t'$ the integration variable is z and the algorithm of Equation (2.1.23) is used to perform the quadrature. Around the singularity Simpson's rule on equally spaced intervals of v and u , instead of z or time, is used for the integration.

Using the algorithms described below, $F_I(t)$ and $p_r(t)$ are evaluated in two steps before and after the singularity. When $t_c - 2\Delta t' < t \leq t_c$, the following variables are used:

$$t_1 = t_c - 2\Delta t' \tag{2.1.25}$$

$$v_1 = v(t_1) = (t_c^2 - t_1^2)^{1/2} \tag{2.1.26}$$

$$t_2 = [t_c^2 - (3v_1/4)^2]^{1/2} \tag{2.1.27}$$

$$t_3 = [t_c^2 - (v_1/2)^2]^{1/2} \tag{2.1.28}$$

$$t_4 = [t_c^2 - (v_1/4)^2]^{1/2} \quad (2.1.29)$$

The fifth time used here is t_c . However, $P_r(t_c)$ does not appear in the equations for F_I because the transformed integrand vanishes. The value of F_I at $t = t_3$ is obtained from

$$F_I(t_3) = F_I(t_1) \exp[-(t_3 - t_1)/G_r] + \{ P_r(t_1) \exp[-(t_3 - t_1)/G_r] v_1/t_1 + 3P_r(t_2) \exp[-(t_3 - t_2)/G_r] v_1/t_2 + P_r(t_3) v_1/2t_3 \} v_1/12. \quad (2.1.30)$$

This equation is coded in reduced notation in Card BOTR589. For the next step $F_I(t_c)$ is calculated using

$$F_I(t_c) = F_I(t_3) \exp[-(t_c - t_3)/G_r] + \{ P_r(t_3) \exp[-(t_c - t_3)/G_r] v_1/2t_3 + P_r(t_4) \exp[-(t_c - t_4)/G_r] v_1/t_4 \} v_1/12. \quad (2.1.31)$$

This equation is coded in reduced notation (Card BOTR597).

Similarly, after the singularity we define the following variables:

$$t_5 = t_c + 2\Delta t' \quad (2.1.32)$$

$$u_1 = u(t_5) = (t_5^2 - t_c^2)^{1/2} \quad (2.1.33)$$

$$t_2 = [t_c^2 + (u_1/4)^2]^{1/2} \quad (2.1.34)$$

$$t_3 = [t_c^2 + (u_1/2)^2]^{1/2} \quad (2.1.35)$$

$$t_4 = [t_c^2 + (3u_1/4)^2]^{1/2} \quad (2.1.36)$$

Here t_1 is the time of the singularity t_c , but $P_r(t_c)$ is not needed

since the transformed integrand vanishes. The value of $F_I(t_s)$ is then given by

$$F_I(t_s) = F_I(t_c) \exp[-(t_s - t_c)/G_r] + \left\{ P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / t_s + P_r(t_s) u_1 / 2t_s \right\} u_1 / 12. \quad (2.1.37)$$

This equation is converted to reduced notation and coded in Card BOTR635. Then the last step using the special integration variables is

$$F_I(t_s) = F_I(t_s) \exp[-(t_s - t_s)/G_r] + \left\{ P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / 2t_s + 3P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / t_s + P_r(t_s) u_1 / t_s \right\} u_1 / 12. \quad (2.1.38)$$

This equation in reduced form is coded in Card BOTR643.

2.1.5 The Impulse and Energy Flux. The impulse I and energy flux E_F are calculated in the main program Cards BOTR717-766. These calculations are made only if the spherical wave bottom reflection is used. The impulse in psi-sec is evaluated from the equation

$$I = \int_{t_0}^t p(t) dt,$$

where $p(t) = p_i(t) + p_r(t) + p_s(t)$ is the total pressure of the incident, bottom reflected, and surface reflected waves and t_0 is the time of the beginning of the pressure pulse $p(t)$.

The energy flux E_F in in-psi is found from the equation

$$E_F = \left\{ \int_{t_0}^t |p| p dt \right\} / (2.3066 \rho_1 c_1),$$

where 2.3066 is a conversion factor necessary for E_F to be in units in-psi when p is in psi, time in seconds, ρ_1 in gm/cm³, and c_1 in ft/sec.

Away from the singularity of $p_r(t)$ of $t = t_c$ and for sub-critical bottom reflections the integrals are determined using Simpson's rule on equally spaced points as a function of time. Near the singularity the change of integration variables is made to v and u as for the convolution integral. This change of variables is made in Cards BOTR738-755.

Also calculated in the same section of the program is the "positive impulse" which is simply the impulse of the positive part of the total pressure $p(t)$. If the full output option is used (see the input Z5 in Section 3.1 and the sample outputs of Appendix B), the magnitudes reduced impulse $I/W^{1/2}$, reduced positive impulse, and reduced energy flux $E_F/W^{1/2}$ are calculated in Cards BOTR793-797.

2.2 The Cagniard-Rosenbaum Method for Calculating the Step Wave

Response

In this section the Cagniard-Rosenbaum equations are listed, and forms of these equations similar to the FORTRAN notation are given. This method is faster than the complex arithmetic method which will be discussed in Section 2.3, but it has the disadvantage that separate equations are required for the precursor and the main wave and for each type of bottom (determined by the ordering

of c_1 , c_2 , and c_4). However, in the coding we were able to take advantage of certain common factors and terms and hence reduce the number of statements that would otherwise be required.

2.2.1 Non-Rigid Bottom Precursors. A fast non-rigid bottom ($c_2 > c_1$) for which $\theta > \theta_{cr}$ has a step wave response at times $\delta \leq t < t_c$ expressed by the following equation (Britt (2-1.10)):

$$P_r(t) = \frac{b(\sigma - M)}{R_r} \int_{-1}^1 \frac{\omega(\sigma + \omega)^{1/2} (1 - \sin \pi t/2) dt}{[(1 - b^2)\omega^2 + \sigma^2 b^2]^{1/2} (\omega - N)^{1/2}}, \quad (2.2.1)$$

where $\omega = (\sigma + M)/2 + [(\sigma - M)/2] \sin \pi t/2$, $(2.2.2)$

$b = \rho_1/\rho_2$, $\tau_m = t/R_r$, $\sigma = (c_1^{-2} - c_2^{-2})^{1/2}$, $M = \tau_m \cos \theta + (c_1^{-2} - \tau_m^2)^{1/2} \sin \theta$, $N = \tau_m \cos \theta - (c_1^{-2} - \tau_m^2)^{1/2} \sin \theta$, $\sin \theta = r/R_r$, and $\cos \theta = d_r/R_r$.

In the program the integration variable $x = \pi t/2$ is used. We also set $w = c_1 \omega$. Then after rearranging, Equation (2.2.1) can be put into the form which is coded

$$R_i P_r(t) = \frac{2\sqrt{2} b R_i}{\pi R_r} \int_{-\pi/2}^{\pi/2} \frac{F_x w dx}{w^2 + b^2 (c_1^{-2} \sigma^2 - w^2)}, \quad (2.2.3)$$

where

$$F_x = (1 - \sin x) \{ \left[(c_1 \sigma + w)(c_1 \sigma - c_1 M) \right] / [1 + \sin x + 4(1 - c_1^2 \tau_m^2)^{1/2} \sin \theta / (c_1 \sigma - c_1 M)] \}^{1/2}. \quad (2.2.4)$$

$$= (1 - \sin x) \{ [(\cos \alpha + w) P(1)] / [1 + \sin x + P(2)] \}^{1/2},$$

with $\cos \alpha = c_1 \sigma = [1 - (c_1/c_2)^2]^{1/2}$,

$$P(1) = \cos \alpha - c_1 M,$$

$$P(2) = 4(1 - c_1^2 \tau_m^2)^{1/2} \sin \theta / P(1).$$

The variables $\cos \alpha$, $P(1)$, and $P(2)$ are calculated in Cards BOTR238, STPA022, and 23.

The integrand above is evaluated in FUNCTION ONE. The variable F_x is coded in Card ONE023, and the value of the integrand is ONE in Card ONE055. The factor outside the integral is calculated in Card STPA025. The integration for this and all other precursors is controlled by SUBROUTINE STPWA which uses the Gaussian quadrature of FUNCTION FGI to evaluate the integral. The value of $R_i P_r(t)$, called STPW in Card STPA027, is returned to the main program BOTREF where the convolution integral is executed.

2.2.2 Rigid Bottom Precursor, Case $c_2 > c_1 > c_4$. The precursor integrands for a rigid bottom are also evaluated in FUNCTION ONE. For the case $c_2 > c_1 > c_4$ (slow shear) the following equation (Britt (4-1.6)) is used

$$P_r(t) = \frac{b(\sigma-\omega)}{4R_r c_4^4} \int_{-1}^1 \frac{(\sigma+\omega)^{1/2} A (1-\sin \pi t/2) dt}{(\omega-\eta)^{1/2} [A^2 + (B + C)^2]} , \quad (2.2.5)$$

where

$$A = \omega(c_4^{-2}/2 - c_1^{-2} + \omega^2)^2 \quad (2.2.6)$$

$$B = \omega(c_1^{-2} - \omega^2)(\sigma^2 - \omega^2)^{1/2} |\omega^2 + c_4^{-2} - c_1^{-2}|^{1/2} \quad (2.2.7)$$

$$C = bc_4^{-4}(\sigma^2 - \omega^2)^{1/2}/4 . \quad (2.2.8)$$

In a manner similar to the non-rigid bottom, Equation (2.2.5) can be rearranged to obtain the program form

$$R_i P_r(t) = \frac{2\sqrt{2} b R_i}{\pi R_r} \int_{-\pi/2}^{\pi/2} \frac{F_x A_x dx}{[A_x^2 + (B_x + C_x)^2]} , \quad (2.2.9)$$

where $w = c_1 \omega$, $\cos \alpha = c_1 \sigma$, and

$$A_x = 4c_1 c_4^4 A = w [1 - 2(c_4/c_1)^2 (1-w^2)]^2, \quad (2.2.10)$$

$$B_x = 4c_1 c_4^4 B = 4w(c_4/c_1)^2 (1-w^2) [(\cos^2 \alpha - w^2) |(c_4/c_1)^2 (w^2-1) + 1|]^{1/2}, \quad (2.2.11)$$

$$C_x = 4c_1 c_4^4 C = b(\cos^2 \alpha - w^2)^{1/2}. \quad (2.2.12)$$

As in the previous case F_x and the factor outside the integral are calculated in Cards ONE023, and STPA025. The variables A_x , B_x , and C_x are coded in Cards ONE043, 044, and 050. The value of the integrand is ONE in Card ONE051, and as before SUBROUTINE STPWA controls the integration.

2.2.3 Rigid Bottom Precursor, Case $c_3 > c_4 > c_1$. The precursor for $c_3 > c_4 > c_1$ (fast shear) is based on the following equation (Britt (4-2.8))

$$P_r(t) = \frac{b(\sigma-M)}{4R_r c_4^4} \int_{\psi_1}^1 \frac{(\sigma+\omega)^{1/2} A (1-\sin \pi \psi/2)}{(\omega-N)^{1/2} [A^2 + (B+C)^2]} d\psi + \frac{b(\sigma-M)}{4R_r c_4^4} \int_{-1}^{\psi_1} \frac{(\sigma+\omega)^{1/2} (A-B) (1-\sin \pi \psi/2)}{(\omega-N)^{1/2} [(A-B)^2 + C^2]} d\psi, \quad (2.2.13)$$

where $\psi_1 = \frac{2}{\pi} \arcsin \left[\frac{2(c_1^{-2} - c_4^{-2})^{1/2} - \sigma - M}{\sigma - M} \right]$.

(In Britt's paper the magnitude B in the second integral is denoted by B_2 , a precaution unnecessary if the definition Equation (2.2.7) is used.) Equation (2.2.13) can then be written in the form used in the program

$$R_i P_r(t) = \frac{2\sqrt{2} b R_i}{\pi R_r} \int_{-\pi/2}^{\pi/2} F_x F_k dx, \quad (2.2.14)$$

where

$$F_k = (A_x - B_x) / [(A_x - B_x)^2 + C_x^2] \text{ for } \omega^2 + C_x^{-2} - C_1^{-2} < 0 \quad (2.2.15)$$

$$F_k = A_x / [A_x^2 + (B_x + C_x)^2] \quad \text{for } \omega^2 + C_x^{-2} - C_1^{-2} \geq 0. \quad (2.2.16)$$

The variables A_x , B_x , and C_x are defined in Equations (2.2.10), (2.2.11), and (2.2.12). In the first case the integrand is coded in Card ONE047 and the second case in Card ONE051.

2.2.4 Step Wave Response at $t = t_c$. At the peak of the bottom reflection at $t = t_c = R_r/c_1$, the step wave response $P_r(t_c)$ is calculated in the main program BOTREF. For supercritical incidence, $\theta > \theta_{cr}$, $P_r = \pm\infty$ where the sign depends on the phase shift ϕ explained in Section 2.5.1. The treatment of this case is discussed in Section 2.1.4. For subcritical incidence, $\theta < \theta_{cr}$, P_r remains finite and $P_r = K/R_r$ where K is the plane wave reflection coefficient of Section 2.5.1.

2.2.5 Non-Rigid Bottom Main Wave, Case $c_2 > c_1$. A fast non-rigid bottom ($c_2 > c_1$) has a step wave response at times $t > t_c$ given by the equation (Britt (2-2.10))

$$P_r(t) = \frac{1}{R_r} \frac{1-b}{1+b} + \frac{2b}{\pi R_r} \int_0^{\sigma} \frac{\omega(\sigma^2 - \omega^2)^{1/2}}{[(1-b^2)\omega^2 + \sigma^2 b^2]} \left\{ \left[(\omega - K_m)^2 + L \right]^{-1/2} - \left[(\omega + K_m)^2 + L \right]^{-1/2} \right\} d\omega, \quad (2.2.17)$$

where

$$K_m = \tau_m \cos \theta \quad (2.2.18)$$

$$L = (\tau_m^2 - c_1^2) \sin^2 \theta. \quad (2.2.19)$$

The subscript m has been kept to distinguish it from the reflection coefficient K .

In the code the integration variable is $w = c_1 \omega$, and the form of the equation is similar to that for the precursor:

$$R_i P_r(t) = \frac{(1-b)R_i}{(1+b)R_r} + \frac{2bR_i}{\pi R_r} \int_0^{\frac{F_x w dw}{w^2 + b^2 (c_1^2 \sigma^2 - w^2)}}, \quad (2.2.20)$$

where F_x is now

$$F_x = \left[\frac{c_1^2 \sigma^2 - w^2}{c_1^2 L + (w - c_1 K_m)^2} \right]^{1/2} - \left[\frac{c_1^2 \sigma^2 - w^2}{c_1^2 L + (w + c_1 K_m)^2} \right]^{1/2} \\ = \left[\frac{\cos^2 \alpha - w^2}{P(8) + (w - P(7))^2} \right]^{1/2} - \left[\frac{\cos^2 \alpha - w^2}{P(8) + (w + P(7))^2} \right]^{1/2} \quad (2.2.21)$$

with $\cos \alpha = c_1 \sigma = [1 - (c_1/c_s)^2]^{1/2}$.

The abbreviations $P(7)$ and $P(8)$ are listed in Cards STPB026 and 27.

The function F_x above is calculated in Card ONE032, and the integrand is ONE in Card ONE055. The factor outside the integral is evaluated in Card STPB032. The first term on the right hand side of Equation (2.2.20) is computed in Card STPB038. The integration for this and all other main wave responses is controlled by SUBROUTINE STPWB.

The value of $R_i P_r(t)$ is calculated in Card STPB047.

2.2.6 Non-Rigid Bottom Main Wave, Case $c_1 > c_2$. The step wave response for a slow non-rigid bottom, one with $c_1 > c_2$, is expressed in the equation (Britt (2-3.14))

$$P_r(t) = \frac{1}{R_r} \frac{1-b}{1+b} - \frac{2\sqrt{2}b}{\pi R_r} \int_0^{\bar{\sigma}} \frac{\bar{w}(\bar{\sigma}^2 - \bar{w}^2)^{1/2} \left\{ [(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - F) \right\}^{1/2}}{(1-b^2)\bar{w}^2 + \bar{\sigma}^2 b^2} d\bar{w} \quad (2.2.22)$$

where $\bar{\sigma} = (c_2^{-2} - c_1^{-2})^{1/2}$, (2.2.23)

$$D = \tau_m^2 \cos 2\theta + c_1^{-2} \sin^2 \theta, \quad (2.2.24)$$

$$E = 4(\sin^2 \theta \cos^2 \theta) \tau_m^2 (c_1^{-2} - \bar{\sigma}^2), \quad (2.2.25)$$

and $F = \tau_m^2 - c_1^{-2} \sin^2 \theta$. (2.2.26)

The form used in the program is

$$R_i P_r(t) = \frac{R_i}{R_r} \frac{1-b}{1+b} - \frac{2\sqrt{2}b R_i}{\pi R_r} \int_0^{\bar{\sigma}} \bar{F}_A \bar{F}_B dx, \quad (2.2.27)$$

where $x = c_1 \bar{w}$

$$\bar{F}_A = x(c_1^{-2} \bar{\sigma}^2 - x^2)^{1/2} / [(1-b^2)x^2 + b^2 c_1^{-2} \bar{\sigma}^2] \quad (2.2.28)$$

$$\bar{F}_B = c_1^{-1} \left\{ \frac{[(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - F)}{(\bar{w}^2 + D)^2 + E} \right\}^{1/2}. \quad (2.2.29)$$

The integrand is evaluated in FUNCTION TWO Card TWO017, \bar{F}_A is coded in Card TWO013, and \bar{F}_B is coded in Cards TWO014 and 015. The terms corresponding to D, E, and F are denoted by P(11), P(12), and P(13) and are evaluated in Cards STPB029-31.

2.2.7 Rigid Bottom Main Wave, Case $c_2 > c_4 > c_1$. The rigid bottom main wave response for the case $c_2 > c_4 > c_1$ (fast shear) is expressed in Britt's equations (4-4.3), (4-3.14), and (4-3.15)

which are as follows:

$$P_r(t) = \frac{1}{R_r} + \Delta \quad (2.2.30)$$

$$+ \frac{b}{2\pi R_r c_4^4} \int_0^{\sigma_2} \frac{(\sigma^2 - \omega^2)^{1/2} (A-B)}{[(A-B)^2 + C^2]} \left\{ \frac{1}{[(\omega - K_m)^2 + L]^{1/2}} \frac{1}{[(\omega + K_m)^2 + L]^{1/2}} \right\} d\omega$$

$$+ \frac{b}{2\pi R_r c_4^4} \int_{\sigma_2}^{\sigma} \frac{A(\sigma^2 - \omega^2)^{1/2}}{[A^2 + (B+C)^2]} \left\{ \frac{1}{[(\omega - K_m)^2 + L]^{1/2}} \frac{1}{[(\omega + K_m)^2 + L]^{1/2}} \right\} d\omega$$

where $\sigma_2 = (|c_4^{-2} - c_1^{-2}|)^{1/2}$ and

$$\Delta = -\frac{\sqrt{2} k}{R_r g_1} \left\{ \frac{(a^2 + f)^{1/2} - a}{a^2 + f} \right\}^{1/2} r \quad (2.2.31)$$

with $\Gamma = \left\{ g_1 \left[\left(\frac{c_4^{-2}}{2} - k^2 \right)^2 - k^2 g_3 g_4 \right] - \frac{b g_3}{4 c_4} \right\} / \left\{ \frac{k}{g_1} \left[\left(\frac{c_4^{-2}}{2} - k^2 \right)^2 - k^2 g_3 g_4 \right] - g_1 k \left[4 \left(\frac{c_4^{-2}}{2} - k^2 \right) + 2 g_3 g_4 + k^2 \left(\frac{g_4}{g_3} + \frac{g_3}{g_4} \right) \right] + \frac{b k}{4 g_3 c_4} \right\} .$

Here, c_{st} is the propagation velocity of the Stonley wave, $k = 1/c_{st}$, $g_1 = (k^2 - c_1^{-2})^{1/2}$, $g_3 = (k^2 - c_3^{-2})^{1/2}$, $g_4 = (k^2 - c_4^{-2})^{1/2}$, $a = r_m^2 - (k^2 - c_1^{-2} \cos^2 \theta)$, and $f = 4r_m^2 g_1^2 \cos^2 \theta$.

The Stonley wave propagation velocity c_{st} is calculated in SUBROUTINE STONL. The equation for c_{st} used in the program is described in Section 2.4.

The above equation is coded in the form

$$R_i P_r(t) = \frac{R_i}{R_r} + R_i \Delta + \frac{2bR_i}{\pi R_r} \int_0^{C_1 \sigma} F_x F_k dw . \quad (2.2.32)$$

F_x and F_k have been defined in Equations (2.2.21), (2.2.15), and (2.2.16).

The first two terms of Equation (2.2.32) are calculated in Cards STPB056-71 for all solid bottom main waves, and the result is stored in the variable TERML. The integrand is determined in FUNCTION ONE in Cards ONE047 and 051 in the same way as for the precursor. However, the function F_x and the factor in front of the integral are here calculated in Cards ONE032 and STPB032 as they were for a fast fluid bottom main wave.

2.2.8 Rigid Bottom Main Wave, Case $c_2 > c_1 > c_4$. The main wave response for the rigid bottom case $c_2 > c_1 > c_4$ (slow shear) is given by the following equation (Britt (4-3.13))

$$\begin{aligned} P_r(t) = & \frac{1}{R_r} + \Delta \\ & + \frac{b}{2\pi R_r c_4^4} \int_0^{\sigma} \frac{A(\sigma^2 - \omega^2)^{1/2}}{A^2 + (B+C)^2} \left\{ \frac{1}{[(\omega - K_m)^2 + L]^{1/2}} - \frac{1}{[(\omega + K_m)^2 + L]^{1/2}} \right\} d\omega \\ & - \frac{\sqrt{2} b}{2\pi R_r c_4^4} \int_0^{\sigma_2} \frac{(\bar{w}^2 + \omega^2)^{1/2} \bar{B}}{(\bar{A} + C)^2 + \bar{B}^2} \left\{ \frac{[(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - F)}{(\bar{w}^2 + D)^2 + E} \right\}^{1/2} d\bar{w} \end{aligned} \quad (2.2.33)$$

where $\bar{A} = \bar{w} [c_4^{-2}/2 - c_1^{-2} - \bar{w}^2]^{1/2}$, $(2.2.34)$

$$\bar{B} = \bar{w} (c_1^{-2} + \bar{w}^2) (\sigma^2 + \bar{w}^2)^{1/2} (c_2^{-2} - \bar{w}^2)^{1/2}, \quad (2.2.35)$$

$$\bar{C} = \frac{b}{4c_4^4} (\sigma^2 + \bar{w}^2)^{1/2}. \quad (2.2.36)$$

The above equation is coded in the form

$$R_i P_r(t) = \frac{R_i}{R_r} + R_i \Delta + \frac{2bR_i}{\pi R_r} \left\{ \int_0^{c_1 \sigma} F_x F_k dw - \left(\frac{c_1}{c_4} \right)^2 \frac{\sqrt{2}}{4} \int_0^{c_1 \sigma_2} \bar{F}_A \bar{F}_B dx \right\} \quad (2.2.37)$$

where $x = c_1 \bar{w}$, F_x and F_k are defined in Equations (2.2.21), (2.2.15), and (2.2.16),

$$\bar{F}_A = \frac{(\bar{w}^2 + \sigma^2)^{1/2} \bar{B}}{c_1^4 [(\bar{A} + \bar{C})^2 + \bar{B}^2]} = \frac{(x^2 + \cos^2 \alpha)^{1/2} \bar{B}_x}{(\bar{A}_x + \bar{C}_x)^2 + \bar{B}_x^2}, \quad (2.2.38)$$

$$\bar{F}_B = c_1^{-1} \left\{ \frac{[(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - E)}{(\bar{w}^2 + D)^2 + E} \right\}^{1/2}, \quad (2.2.39)$$

$$\cos \alpha = c_1 \sigma = [1 - (c_1/c_4)^2]^{1/2},$$

$$\bar{A}_x = c_1^8 \bar{A} = x [(c_1/c_4)^2/2 - 1 - x^2]^2,$$

$$\bar{B}_x = c_1^8 \bar{B} = x (1 + x^2) \{ [\cos^2 \alpha + x^2] [(c_1/c_4)^2 - 1 - x^2] \}^{1/2},$$

$$\bar{C}_x = c_1^8 \bar{C} = b (c_1/c_4)^4 (\cos^2 \alpha + x^2)^{1/2} / 4.$$

The first three terms of Equation (2.2.37) are calculated using the same cards as for the fast shear case. The integrand of the second integral is computed in FUNCTION ONE1. \bar{F}_A and \bar{F}_B are expressed in Cards ONE1019, 20, and 21. \bar{A}_x , \bar{B}_x , and \bar{C}_x are calculated in Cards ONE1015-17. The terms corresponding to D, E, and F are denoted by P(11), P(12), and P(13) and are evaluated in Cards STPB029-31. The value of the integrand is stored in the variable ONE1 in Card ONE1023. The response $STPW = R_i P_r(t)$ is then determined in Cards STPB079 and 80.

2.3 The Complex Arithmetic Method for Calculating the Step WaveResponse

A second option for calculating the step wave response is provided by the complex arithmetic method. This procedure is based on the equation (Britt (5-2.12))

$$P_r(t) = \frac{2}{\pi} \int_{y_1}^{y_2} \operatorname{Re} \left\{ \frac{u}{\alpha_1 y} (K - K_1) \right\} dy + \frac{2}{\pi R_r} \cdot \operatorname{Im} \left\{ K_1 \log \left[\frac{R_r \omega_2 - d_r t / R_r}{f(\omega_1)} \right] \right\} \quad (2.3.1)$$

where $u = x + iy$ and for $t < t_c = R_r/c_1$

$$x = 0 \\ y_1 = c_2^{-1} \\ y_2 = R_r^{-2} [tr - d_r (c_1^{-2} R_r^2 - t^2)^{1/2}] \quad (2.3.2)$$

For times $t > t_c$ these variables are

$$x = R_r^{-2} d_r (t^2 - c_1^{-2} R_r^2)^{1/2} \\ y_1 = 0 \\ y_2 = R_r^{-2} tr. \quad (2.3.3)$$

The reflection coefficient K for a solid bottom is defined

$$K = \frac{\alpha_1 [(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4] - b \alpha_2 c_4^{-4}}{\alpha_1 [(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4] + b \alpha_2 c_4^{-4}} \quad (2.3.4)$$

where $\alpha_i = (c_i^{-2} + u^2)^{1/2}$ for $i = 1, 2, 4$.

For a fluid bottom $c_4 = 0$, and the equation for K reduces to

$$K = (\alpha_1 - b\alpha_2)/(\alpha_1 + b\alpha_2). \quad (2.3.5)$$

K_1 is the value of K at $u = x + iy_2$. The other variables used above are as follows:

$$\gamma = [u^2 r^2 + (t - d_r \alpha_1)^2]^{1/2} \quad (2.3.6)$$

$$\omega_1 = [c_1^{-2} + (x + iy_1)^2]^{1/2} \quad (2.3.7)$$

$$\omega_2 = [c_1^{-2} + (x + iy_2)^2]^{1/2} \quad (2.3.8)$$

$$f(\omega_1) = [R_r^2 \omega_1^2 - 2d_r t \omega_1 + (t^2 - c_1^{-2} r^2)]^{1/2} + \omega_1 R_r - d_r t / R_r. \quad (2.3.9)$$

The form of Equation (2.3.1) which is coded is

$$R_i P_r(t) = \left\{ A_2 + \int_{c_1 y_1}^{c_1 y_2} \operatorname{Re} [F \cdot (K - K_1)] dz \right\} / \left(\frac{\pi R_r}{2 R_i} \right) \quad (2.3.10)$$

where $z = c_1 y$,

$$A_2 = \operatorname{Im} \left[K_1 \log \left\{ \left[c_1 \omega_2 - c_1 \tau_m \cos \theta \right] / \left[(c_1^2 \omega_1^2 - 2c_1 \tau_m \cos^2(c_1 \omega_1) + (c_1^2 \tau_m^2 - \sin^2 \theta))^{1/2} + c_1 \omega_1 - c_1 \tau_m \cos \theta \right] \right\} \right], \quad (2.3.11)$$

and

$$F = \frac{c_1 u}{c_1 \alpha_1 (c_1 \gamma / R_r)} = \frac{u R_r}{c_1 \alpha_1 \gamma}. \quad (2.3.12)$$

As in the Cagniard-Rosenbaum method, the response $STPW = R_i P_r(t)$ is calculated in SUBROUTINE STPWA for the precursor ($t < t_c$) and in SUBROUTINE STPWB for the main wave ($t > t_c$) using the Gaussian quadrature of FGI to evaluate the integral. The last factor in Equation (2.3.10) is calculated in Cards STPA039 and STPB097. The integrand and A_2 are coded in FUNCTION SEVEN, Cards SEVN035 and 045. The value of $A_2(t)$ is obtained from STPWA by a call to SEVEN with

$z = c_1 y_2$. The function $K_1 = K(x+iy_2)$ is evaluated using the same equations as for $K(u)$ in the integral, namely, RCOE in Cards SEVN022 and 029. In FUNCTION SEVEN the variables brought over by COMMON statements are calculated in the main program, and members of the P array are determined in Cards STPA035-38 for the precursor and in Cards STPB093-96 for the main wave.

2.4 The Stonley Wave Propagation Velocity

The Stonley wave propagation velocity c_{st} is defined as the zeroes $u = \pm i/c_{st} = \pm ik$ of the denominator of the solid bottom reflection coefficient expressed in Equation (2.3.4). Thus $u^2 = -c_{st}^{-2}$ is the solution of the equation

$$\alpha_1 \left[(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4 \right] + b \alpha_2 c_4^{-4} = 0, \quad (2.4.1)$$

where $\alpha_1 = (c_1^{-2} + u^2)^{1/2}$,

$$\alpha_2 = (c_2^{-2} + u^2)^{1/2},$$

and $\alpha_4 = (c_4^{-2} + u^2)^{1/2}$.

To obtain the form of Equation (2.4.1) which is used in the program, first note that the square roots α_1 , α_2 , and α_4 are imaginary since c_{st} is known to be smaller than c_1 , c_2 , and c_4 . Next replace u^2 by $-c_{st}^{-2}$, multiply through by $ic_1 c_2 c_4^{-4} c_{st}^5$, and set $y_2 = c_{st}^2$ to obtain

$$(c_1^{-2} - y_2)^{1/2} \left\{ c_2 (y_2 - 2c_4^{-2})^2 - 4c_4^{-3} [(c_2^{-2} - y_2)(c_4^{-2} - y_2)]^{1/2} \right\} \\ + bc_1 y_2^2 (c_2^{-2} - y_2)^{1/2} = 0. \quad (2.4.2)$$

This equation is solved for y_2 in SUBROUTINE STONL by iteration using the secant method. The variable y_2 is denoted by the FORTRAN symbol $Y2$. Then c_{st} , called CSTON in the code, is the square root of y_2 .

2.5 Theory of the Plane Wave Bottom Reflection

In cases where the plane wave bottom reflection is adequate for one's needs or when one wishes to compare these results with the spherical wave reflection, the plane wave option of the BOTREF program can be used. The reflection geometry, incident and critical angles, and the incident pulses are the same as for the spherical wave in Section 2.1; and, unless otherwise noted, the notation is the same.

2.5.1 The Plane Wave Reflection Coefficient and Phase Shift.

The plane wave reflection coefficient K and phase shift ϕ for a non-rigid bottom are calculated from the following equations. For subcritical angles of incidence K and ϕ are

$$K = (A_T - 1)/(A_T + 1) \quad (2.5.1)$$

and

$$\phi = 0$$

where $A_T = \cos \theta / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}]$. $(2.5.2)$

At the critical angle θ_{cr} these expressions reduce to $K = 1$ and $\phi = 0$. At supercritical incidence we have

$$|K| = 1$$

and $\phi = 2 \arctan [b(\sin^2 \theta - \sin^2 \theta_{cr})^{1/2} / \cos \theta]$. $(2.5.3)$

The above equations are coded in the main program Cards BOTR260-274, 307. The FORTRAN variables CR and E2 denote K and ϕ . If K is complex, then $CR = |K|$.

For a rigid bottom K and ϕ are determined from the equations below. At subcritical incidence, $\theta < \theta_{cr} < \theta_{crs}$, we have

$$\begin{aligned} K &= (A_T + B_T - 1)/(A_T + B_T + 1) \\ \phi &= 0 \end{aligned} \quad (2.5.4)$$

where

$$A_T = \cos \theta [1 - 2\sin^2 \theta / \sin^2 \theta_{crs}]^{1/2} / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}] \quad (2.5.5)$$

and

$$B_T = 4\cos \theta \sin^2 \theta (\sin^2 \theta_{crs} - \sin^2 \theta) / [b \sin^4 \theta_{crs} (\sin^2 \theta_{crs} - \sin^2 \theta)^{1/2}] \quad (2.5.6)$$

At the critical angle $\theta = \theta_{cr}$ the equations simplify to $K = 1$ and $\phi = 0$. For an incident angle in the range $\theta_{cr} < \theta < \theta_{crs}$ the reflection coefficient is complex. Its modulus is

$$|K| = \left\{ [A_{TA}^2 + (B_T - 1)^2] / [A_{TA}^2 + (B_T + 1)^2] \right\}^{1/2} \quad (2.5.7)$$

and the phase shift is

$$\phi = \arctan[(1-B_T)/A_{TA}] + \arctan[(1+B_T)/A_{TA}] \quad (2.5.8)$$

where

$$A_{TA} = \cos \theta [1 - 2\sin^2 \theta / \sin^2 \theta_{crs}]^{1/2} / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}] \quad (2.5.9)$$

At the critical angle of the shear wave θ_{crs} the equations reduce to

$$|K| = 1$$

and $\phi = 2 \arctan(1/A_{TA})$. (2.5.10)

For angles of incidence $\theta > \theta_{crs}$ we have

$$|K| = 1$$

and $\phi = 2 \arctan[1/(A_{TA} + B_{TA})]$ (2.5.11)

where $B_{TA} = 4 \cos \theta \sin^2 \theta (\sin^2 \theta_{crs} - \sin^2 \theta) / [b \sin^4 \theta_{crs} (\sin^2 \theta - \sin^2 \theta_{crs})^{1/2}]$ (2.5.12)

These equations for the solid bottom reflection coefficient and phase shift are coded in Cards BOTR280-307. As for the fluid bottom, K and ϕ are denoted by CR and E2; and if K is complex, $CR = |K|$.

2.5.2 The Plane Wave Bottom Reflection Pressure History.

The plane wave bottom reflection pressure history $p_r(t)$ is calculated from the following equations:

when $\theta \leq \theta_{cr}$,

$$p_r = 0 \quad \text{for } t < t_c = R_r/c_1$$

$$p_r = p_F(R_r) K \exp[-(t-t_c)/G_r] \quad \text{for } t \geq t_c \quad (2.5.13)$$

when $\theta > \theta_{cr}$,

$$p_r = p_F(R_r) \frac{|K|}{\pi} \exp[-(t-t_c)/G_r] E_1[(t_c-t)/G_r] \sin \phi \quad \text{for } \delta \leq t < t_c$$

$$p_r = \pm \infty \quad \text{with the sign of } \phi \quad \text{for } t = t_c \quad (2.5.14)$$

$$p_r = p_F(R_r) |K| \exp[-(t-t_c)/G_r] \left\{ \cos \phi - \frac{1}{\pi} Ei[(t-t_c)/G_r] \sin \phi \right\} \quad \text{for } t > t_c \quad (2.5.15)$$

Note that the plane wave theory has been modified to use $p_F(R_r)$ and $G_r = G(R_r)$ which account for non-linear changes of the shock wave peak pressure and time constant with distance. Also the arrival times of the main wave and precursor have been changed to conform to the spherical wave situation. In the strict plane wave theory

the precursor begins at $t = -\infty$, and the incident wave and the reflected peak arrive simultaneously.

The functions $E_1(x)$ and $Ei(x)$ are the exponential integrals defined for $x > 0$ as

$$E_1(x) = \int_x^{\infty} \frac{\exp(-y)}{y} dy \quad (2.5.16)$$

$$\begin{aligned} Ei(x) &= - \int_{-x}^{\infty} \frac{\exp(-y)}{y} dy = -E_1(-x) \\ &= \int_{-\infty}^x \frac{\exp(y)}{y} dy. \end{aligned} \quad (2.5.17)$$

The function $E_1(x)$ is evaluated using the following approximate formula (see for example Abramowitz and Stegun (5))

$$0 \leq x < 1$$

$$E_1(x) \approx a_0 + a_1 x + a_2 x^2 + a_3 x^3 + a_4 x^4 + a_5 x^5 - \log(x) \quad (2.5.18)$$

$$a_0 = -.57721566$$

$$a_3 = .05519968$$

$$a_1 = .99999193$$

$$a_4 = -.00976004$$

$$a_2 = -.24991055$$

$$a_5 = .00107857$$

$$x \geq 1$$

$$x \exp(x) E_1(x) \approx \frac{x^4 + a_1 x^3 + a_2 x^2 + a_3 x + a_4}{x^4 + b_1 x^3 + b_2 x^2 + b_3 x + b_4} \quad (2.5.19)$$

$$a_1 = 8.5733287$$

$$b_1 = 9.5733223$$

$$a_2 = 18.059017$$

$$b_2 = 25.632956$$

$$a_3 = 8.6347609$$

$$b_3 = 21.0996531$$

$$a_4 = .26777373$$

$$b_4 = 3.9584969$$

The function $Ei(x)$ is evaluated for $x \leq .5$ using the formula (Reference (5))

$$Ei(x) \approx \gamma + \log(x) + \sum_{n=1}^7 \frac{x^n}{nn!} \quad (2.5.20)$$

where $\gamma = .57721566\dots$ is Euler's constant. For $x > .5$, $Ei(x)$ is obtained from

$$\exp(-x)Ei(x) = \exp(-x)Ei(1) + \int_1^x \frac{\exp(y-x)}{y} dy, \quad (2.5.21)$$

where $Ei(1) = 1.8951178$. The integral is then evaluated using the Gaussian quadrature of FUNCTION FGI.

The reflected pressure $p_r = PBOT$ is calculated in the main program in Cards BOTR861-879. The exponential integrals E_1 and Ei are calculated in the subprograms EXEL and EXEI, and the integrand of Equation (2.5.21) is coded in FUNCTION EXPO.

3. THE BOTTOM REFLECTION COMPUTER CODE

The Bottom Reflection Code has been programmed in FORTRAN IV for use on the CDC 6400 computer at NOL. The code is made up of a main program called BOTREF and the following bottom reflection related subprograms: STONL, STPWA, STPWB, ONE, ONE1, TWO, SEVEN, EXE1, EXEI, EXPO, FGI, PLOT1, and SCAL. In addition, the NOL general purpose plotting program CALCM1 must be included for the generation of a tape to be plotted on CALCOMP incremental plotters. For NOL users CALCM1 is available on the subroutine library tape. The control cards which are included in the program listing of Appendix A contain the statements necessary for using CALCM1 from the library tape. For programmers outside of NOL information on the plotting programs may be obtained from the NOL Mathematics Department (Code 330).

The basic organization of the bottom reflection code is as follows. The main program BOTREF handles all of the input and output and calculates the shock wave peak pressure and time constant and other time independent magnitudes. It performs the time incrementation of the pressure-time histories and calculates the convolution integral, impulse, and energy flux for the spherical wave bottom reflection.

The spherical wave step wave response $P_r(t)$ is obtained by calls from BOTREF to STPWA for the precursor and to STPWB for the main wave. These subroutines in turn set up the integration for $P_r(t)$ using the Gaussian quadrature in FGI. The various integrands described in Sections 2.2 and 2.3 are calculated in subprograms ONE, ONE1, TWO, and SEVEN. The Stonley wave propagation velocity

c_{st} for rigid bottoms is computed in SUBROUTINE STONL on a call from the main program.

The plane wave bottom reflection is also calculated in the main program. Calls to SUBROUTINES EXE1 and EXEI obtain the exponential integrals E_1 and E_i which are used to determine the bottom reflection in Equations (2.5.13), (2.5.14), and (2.5.15).

SUBROUTINES PLOT1 and SCAL set up the CALCOMP plots of the pressure-time history. PLOT1 calls SCAL to scale the plot, calls CALCM1 for plotting the axes and the pressure-time curves, and then calls SUBROUTINES SYMBL4 and NUMBR, which are part of the CALCM1 program, to write additional information on the plots.

The Bottom Reflection Program also has an option for calculating the peak translational velocity (PTV) induced in submerged or floating targets by the bottom reflected pulse. Either of the spherical or plane wave reflection theories may be used. The targets are approximated by an infinitely long cylinder of a specified radius, and the PTV Program described in Reference (6), is used to calculate the peak translational velocity. This program uses the additional subroutines PTV, FV, F1, XMAX, VTAB, and PTAB. The PTV is calculated by calling SUBROUTINE PTV (Cards BOTR813L and 813M).

The cards in the main program which are necessary for PTV calculations are denoted by card numbers followed by letters A, B, C, etc. If the bottom reflection program is not to be used for PTV calculations, these cards and the subroutines of the PTV Program may be omitted.

In the following paragraphs the most important FORTRAN symbols of each subprogram are described, and the locations in the program are given where each symbol is calculated.

3.1 FORTRAN Symbols of the Main ProgramProgram Input

The input data is read in Statements 3 and 4, Cards BOTR041, 42, and 89, and in Card BOTR101I using the format 8F10.5. These inputs are explained in comment Cards BOTR011-39, 72-87, and 101B-101G. The inputs and their units are as follows:

First Data Card, Statement 3

WCH charge weight W in pounds or KT
 CWATER sound velocity c_1 of water in ft/sec
 CBOT sound velocity c_2 of the bottom material in ft/sec
 CSHEAR a double purpose input expressing the rigidity of the bottom.
 If $CSHEAR > .5$, it is the shear wave propagation velocity
 c_4 of the bottom in ft/sec. If $CSHEAR \leq .5$, it is the dimensionless Poisson ratio from which the shear velocity c_4 is
 calculated in Card BOTR062. Values of $c_4 \leq .5$ can be neglected.
 RHOWAT density ρ_1 of water in gm/cm³
 RHOBOT density ρ_2 of the bottom material in gm/cm³
 PRECOE coefficient C_p of the pressure similitude equation in psi.
 PRECOE depends on whether W is in pounds or KT.
 Z5 a control parameter. Z5 greater than zero results in a
 shorter print out for the spherical wave reflection.
 See Appendix B to compare the short and long print out.

Second Data Card, Statement 3

PREEXP exponent n_p of the pressure similitude equation
 THECOE coefficient C_G of the time constant similitude equation in
 seconds. This variable also depends on the units of W .

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THEEXP exponent n_G of the time constant similitude equation
STEPS number of points in the pressure-time history for one time
constant G. STEPS = 20. is usually sufficient to obtain
a smooth, detailed pressure history. In many cases,
STEPS = 10. or 5.0 is adequate.
DURAT duration of pressure-time history in multiples of the time
constant G. If negative, its absolute value is the duration
after the arrival of the bottom reflection peak at $t = t_c$.
If positive, it is the duration after the direct wave
arrival. DURAT = -3.0 is generally sufficient for calculating
the significant parts of the bottom reflection.
X1 CALCOMP plot scaling parameter for the Y-axis in psi per
inch of graph. The X-axis is always drawn three inches
above the bottom of the graph. The length of the Y-axis
is nine inches. Thus the maximum pressure plotted is
6 * X1, and the minimum is -3 * X1. Pressures outside of
this range are plotted at the maximum or minimum, whichever
is applicable.
X2 scaling parameter for the X-axis in microseconds per inch
of graph. If $X2 \leq 0$., SUBROUTINE SCAL calculates an
appropriate value of X2.
SLOPE slope of the bottom from charge to gauge in degrees. If
the slope is not zero, the internal computing geometry is
changed in Cards BOTR170-183. SLOPE must be zero if the
geometry changing options of Z2 and THOVAL are used.

Third Data Card, Statement 4

BIGH water depth H at the charge in feet. BIGH is also used as
a control parameter. After completion of each bottom

reflection pressure-history, the program control returns to Statement 4 to read a new set of data. If BIGH = 0., the program stops. If positive, computation continues with the new geometry. If negative, program control transfers to Statement 3 where a new set of charge, physical constants, etc., are read.

D depth d of the charge below the water surface in feet

DGAU depth d_g of the gauge in feet

SMALLR horizontal range r between charge and gauge in feet

THOVAL desired ratio between the bottom reflection incident angle θ and the critical angle θ_{cr} . The variables D and DGAU are changed in Cards BOTR137-142 to obtain this ratio. SMALLR is not changed. If THOVAL ≤ 0 the geometry is not changed. See Appendix C for a discussion of this option.

Z1 parameter which selects the theory. When Z1 = 0. the spherical wave Cagniard-Rosenbaum method is used. When Z1 = 1.0, the Arons-Yennie plane wave theory is used. And for Z1 = 3.0, the complex arithmetic method is used to calculate the spherical wave bottom reflection. Cards BOTR389-443 make the theory selection and write out the appropriate headings.

Z2 arrival time difference between the bottom reflection peak (at $t = t_c$) and the direct wave in microseconds. If Z2 ≤ 0 ., the geometry is not changed. When the geometry is changed, D and DGAU are varied to obtain the desired arrival time difference. SMALLR is not changed, and the change in D is the negative of the change in DGAU so that the incident

angle θ is also unchanged. This geometry change is performed in Cards BOTR121-127. See Appendix C for a discussion of this option.

Z3 plot control parameter. A CALCOMP plot tape is generated if Z3 = 0.

Fourth Data Card (BOTR101I), For PTV Calculation

RADIUS cylinder radius in feet. This is the draft or cross-sectional radius of the target vessel. If RADIUS $\leq 0.$, the PTV is not calculated.

APRINT controls printing in SUBROUTINE PTV. If APRINT $\leq 0.$, the translational velocities calculated in the iteration for the PTV are printed. An example of this printout is given in Table B.1 following the pressure-time history. If APRINT $\geq 0.$, the variables TIME1, PTV1, and PTV2 described below are printed from the main program (Card BOTR813N).

Program Output

Appendix B contains examples of the full print out and the shorter print out for the spherical wave reflection and a print out for a plane wave reflection. Most of the variables in the output are self-explanatory; others which are not so well defined are described below.

SMALLH height $h = H - d$ of the charge above the bottom

DEZFRO height $d - d_g$ between the charge and gauge depths

D2 reduced height d_r/R_i from image charge to gauge

COSAL $c_1 \sigma$

COSTH $\cos \theta$

SINTH $\sin \theta$

DT	increment Δt of the reduced time \bar{t}
EDT	$\exp(-\Delta t/G_r)$
T	reduced time \bar{t}
STPW	$R_i P_r(t)$
FI/THETA	$R_i F_I(t)/G_r$
PD	incident pressure p_i in psi
TIME	time in seconds relative to the direct wave arrival time
PBOT	bottom reflected pressure p_r in psi
PS	surface reflected pressure p_s in psi
P	total pressure $p = p_i + p_r + p_s$ in psi. Negative pressures are cut off so that $P + \text{hydrostatic} \geq 0$.
FIMP	total impulse I in psi-sec calculated from the equation
	$I = \int_{t_o}^t p \, dt, \text{ where } t_o \text{ is the minimum of } \delta \text{ and } R_i/c_i$
EFLUX	energy flux E_F in in-psi defined by the equation
	$E_F = \left(\int_{t_o}^t p \, p \, dt \right) / (2.3066 \rho_i c_i)$
VMID	value of STPW at $t - \Delta t$, $R_i P_r(t - \Delta t)$
PRE	value of STPW at $t - 2\Delta t$
RESID	R_i^Δ
RFIMP	reduced impulse $I/W^{1/3}$
REFLUX	reduced energy flux $E_F/W^{1/3}$
POSIMP	impulse of the positive part of the total pressure pulse $p(t)$
RPOSIM	reduced positive impulse, $POSIMP/W^{1/3}$
TIME1	time in seconds of the PTV, where time is taken to be zero at the beginning of the bottom reflection
PTV1	the PTV in ft/sec induced by the bottom reflection in a submerged target

PTV2 the PTV in ft/sec induced by the bottom reflection in a target at the surface

Time Independent FORTRAN Symbols

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
B	$b = \rho_1 / \rho_2$	BOTR050
POISR	Poisson ratio $\bar{\sigma} = (.5c_2^2 - c_4^2) / (c_2^2 - c_4^2)$	59
CSHEAR	shear velocity c_4 calculated from $\bar{\sigma}$	62
CSTON	Stonley wave velocity c_{st}	69
SMALLH	h for zero slope	147
RACTU	R_i	151
PH	negative of the hydrostatic pressure at depth d_g	154
RS	reduced surface reflection arrival time	158
W13R	$W^{1/3} / R_i$	162
REDR	$R_i / W^{1/3}$	163
THETA	\bar{G}	164
PACT	$P_F(R_i)$	165
TACT	characteristic time R_i / c_1	166
THET	G in milliseconds	167
A	bottom slope in radians	172
D2ACTU	d_r	187
R2ACTU	R_r	188
CTWO	$\sin \theta_{cr} = c_1 / c_2$	213
R2	reduced bottom reflection slant range R_r / R_i	214
THETAR	$\bar{G}_r = \bar{G}(R_r)$	215
THETR	G_r in milliseconds	216

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PACTC	$(R_r/R_i)^{n_G-1}$	BOTR217
R1	$(R_s/R_i)^{n_G}$	218
THETSR	\bar{G}_s	219
SINTH	$\sin \theta$	221
COSTH	$\cos \theta$	222
D2R2	$c_1 \delta/R_i$ for supercritical reflection	226
	$c_1 \delta/R_i$ for subcritical reflection	228
SINAL	$\sin \theta_{cr}$	235
COSAL	$\cos \theta_{cr} = c_1 \sigma$	238
SINBE	$\sin \theta_{crs}$	243
THE	incident angle θ in degrees	245
CR	plane wave reflection coefficient K	261-304
E2	phase shift ϕ in radians	260-307
EE	phase shift ϕ in degrees	308
ANGA	angle of shear wave in bottom in degrees	312, 315
THONE	angle of compression wave in bottom in degrees	319, 321
ALPHA	θ_{cr} in degrees	352
BETHA	θ_{crs} in degrees	358
SHD2R2	reduced arrival time of critically refracted shear wave	363-365
C2	c_1^2	446
CBOT2	c_2^2	447
CSHR2	c_4^2	448
SINTH2	$\sin^2 \theta$	449
CBSH	$-4c_4^3/c_3$	450

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<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
C2SHR2	$2c_4^2$	BOTR451
C4CB	$c_1^4 b/c_2$	452
<u>Spherical Wave Pressure-Time Calculations</u>		463-816
<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
DT	increment $\Delta\bar{t}$ of reduced time increment $\Delta\bar{t}' \approx \Delta\bar{t}/8$	BOTR476, 500, 652 566, 609
DT1	original value of $\Delta\bar{t}$	478
DTACT	$2\Delta\bar{t}/3$	479, 501
EDT	$\exp(-\Delta\bar{t}/G_r)$	481, 503
N	control parameter for pressure history	721, 804
VMID	$R_i P_r(\bar{t} - \Delta\bar{t})$	542-662
STPW	$R_i P_r(\bar{t})$	520-690
PRE	$R_i P_r(\bar{t} - 2\Delta\bar{t})$	557-691
FI	convolution integral F_I	556-673
NP	number of subintervals to be used in the Gaussian quadrature integration for $P_r(t)$	196, 552-693
V	$v(t)/t_c$ for $t \approx t_c - 2\Delta\bar{t}'$ where $\Delta\bar{t}'$ is approximately $\Delta\bar{t}/8$	573
T1	$\bar{t}(v)$	578
T2	$\bar{t}(.75 v)$	579
T3	$\bar{t}(.5 v)$	580
T4	$\bar{t}(.25 v)$	581
U	$u(t_c + 2\Delta\bar{t}')/t_c$ for $\Delta\bar{t}' \approx \Delta\bar{t}/8$	616
T2	$\bar{t}(.25 u)$	623
T3	$\bar{t}(.5 u)$	624
T4	$\bar{t}(.75 u)$	625
T5	$\bar{t}(u)$	626

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<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PD	incident pressure $p_i(t)$	BOTR702
PS	surface reflected pressure $p_s(t)$	704
PBOT	bottom reflected pressure $p_r(t)$	707
P	total pressure $p = p_i + p_r + p_s$. Negative values of p are cut off at $p + \text{hydrostatic} \geq 0$. 713	

Impulse and Energy Flux Calculations 717-767

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XP	maximum of pressure p and zero	BOTR720
PMID	pressure p of even numbered time $t - \Delta t$	724
XPMID	maximum of PMID and zero	725
PPRE	pressure p at odd numbered time $t - 2\Delta t$	763
XPPRE	maximum of PPRE and zero	764
PEND	pressure $p(t)$ at odd numbered time	757
XPEND	maximum of PEND and zero	758

Variables Used in the PTV Calculation and in Plotting

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XX	storage array for time in microseconds for CALCOMP plot. Here time is zero at the arrival of the direct wave.	BOTR800,889
YY	storage array for the total pressure p for plot	801,890
IPMAX	number of plot points stored in XX and YY arrays	807
QX	the array in which the time in seconds is stored for PTV calculations. This time is zero at the beginning of bottom reflection. and 891E	802E,813H,

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
QY	the array in which the bottom reflection pressure $p_r(t)$ is stored for PTV calculations. If $p_r(t)$ is negative, the value stored in QY is calculated so that $p_r(t) + \text{hydrostatic} \geq 0$	BOTR802F, 813I and 891F
TIMER2	arrival time of the peak or singularity of the bottom reflection. Time in this case is measured from the beginning of the bottom reflection.	813C
XT3	signals the approach of the bottom reflection singularity. The value TIMER2 - $2\Delta t$ is used. The symbol T3 is used for this variable in SUBROUTINE PTV.	813D
XT4	The earliest time at which the translational velocity is to be calculated. The symbol T4 is used instead of XT4 in SUBROUTINE PTV.	813E
XT5	the largest value of time at which the translational velocity is to be calculated. The symbol T5 is used instead of XT5 in SUBROUTINE PTV.	813F
COSA	cosine of the angle which the bottom reflection ray makes with the water surface or a line parallel to the surface if the gauge position is below the surface	813J

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<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PTS	the number of times at which the translational velocity is calculated in the initial search for the PTV. In the call to SUBROUTINE PTV the value PTS = 30. is used.	BOTR813L

Plane Wave Bottom Reflection Variables BOTR819-905

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
SW	direct wave response $p_i(t)/p_F$	BOTR852
PRFL	bottom reflection response $p_r(t)/p_F$	857-877
TBTH	$(t - t_c)/G_r$	863
XE1	$\exp(-TBTH) E_1(-TBTH)$	865
XEI	$\exp(-TBTH) Ei(TBTH)$	874

3.2 FORTRAN Symbols of SUBROUTINE STONL

<u>Symbol</u>	<u>Definition</u>
Y2	$y_2 = c_{st}^2$
FY	Equation (2.4.2) which defines y_2
CK	increment of $y_2 = y_2/1000$
CSTON	Stonley wave velocity c_{st}

3.3 FORTRAN Symbols of SUBROUTINE STPWA

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TR	$c_1 \tau_m$	STPA013
V	$c_1 (c_1^{-2} - \tau_m^{-2})^{1/2}$	14
Cagniard-Rosenbaum Method, CONTR = 0.		
P(9)	0. for precursor	20
XM	$c_1 M$	21

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
P(1)	$c_1 (\sigma - M)$	STPA022
P(2)	$4(c_1^{-2} - \tau_m^2)^{1/2} \sin \theta / (\sigma - M)$	23
P(5)	$c_1 (\sigma + M)$	24
FACTOR	$2 \sqrt{2} b R_i / \pi R_r$	25
STPW	$R_i P_r(t)$	27,43

Complex Arithmetic Method, CONTR = 3.0

P(1)	$c_1 x = 0.$	35
P(2)	$c_1 \tau_m$	36
P(3)	$c_1 / c_2 = c_1 y_1$	37
P(4)	$c_1 y_2$	38
FACTOR	$\pi R_r / 2 R_i$	39
ANS2	A_2	41

3.4 FORTRAN Symbols of SUBROUTINE STPWB

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TR	$c_1 \tau_m$	STPB014
Cagniard-Rosenbaum Method, CONTR = 0.		
P(9)	1.0 for main wave	20
XK	$c_1 K_m$	23
XL	$c_1^2 L$	24
P(7)	XK	26
P(8)	XL	27
P(11)	$c_1^2 D$	29
P(12)	$c_1^4 E$	30
P(13)	$c_1^2 F$	31
FACTOR	$2bR_i / \pi R_r$	32

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TERML	$(R_i/R_r)(1 - b)/(1 + b)$	STPB038
SIGM	$c_1 \bar{\sigma}$	42
STPW	$R_i P_r(t)$	43, 47, 79, 85, 100
XG1	$c_1 g_1$	60
XG3	$c_1 g_3$	61
XG4	$c_1 g_4$	62
XSA	$c_1^2 a/R_i^2$	63
XSF	$c_1^4 f/R_i^4$	64
XNUM	numerator of Γ	65
XDEN	c_1^{-2} times the denominator of Γ	66-67
RESID	R_i^Δ	68-69
TERML	$R_i(1/R_r + \Delta)$	71
SIG2	$c_1(c_4^{-2} - c_1^{-2})^{1/2}$	78

Complex Arithmetic Method, CONTR = 3.0

P(1)	$c_1 x = c_1 \cos \theta (\tau_m^2 - c_1^{-2})^{1/2}$	93
P(2)	$c_1 \tau_m$	94
P(3)	$c_1 y_1 = 0$	95
P(4)	$c_1 y_2 = c_1 \tau_m \sin \theta$	96
FACTOR	$\pi R_r/2R_i$	97
ANS2	A_2	99

3.5 FORTRAN Symbols of FUNCTION ONE

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Precursor Variables		
X	integration variable	ONE 008
W	$c_1 w$	20

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XC2	$c_1^2(\sigma^2 - \omega^2)$	ONE 021,28
FX	F_x defined by Equation (2.2.4)	23
Main Reflection Variables		
W	integration variable $w = c_1 \bar{w}$	27
FX	F_x defined by Equation (2.2.21)	32
Variables for Rigid Bottom Precursors and Main Waves		
FRCS	$c_4^2(w^2 + c_4^{-2} - c_1^{-2})$	41
XA	A_x	43
XB	B_x	44
XC	C_x	50
ONE	rigid bottom integrands defined in Equations (2.2.9), (2.2.14), and (2.2.32) and the integrand of the first integral in Equation (2.2.37).	47,51
ONE	fast non-rigid bottom integrands of Equations (2.2.3) and (2.2.21)	55

3.6 FORTRAN Symbols of FUNCTION ONE1

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
X	integration variable $x = c_1 \bar{w}$	ONE1008
XAB	$c_1^5 \bar{A} = \bar{A}_x$	15
XBB	$c_1^5 \bar{B} = \bar{B}_x$	16
XCB	$c_1^5 \bar{C} = \bar{C}_x$	17
FAB	\bar{F}_A defined by Equation (2.2.38)	19
FBB	\bar{F}_B defined by Equation (2.2.39)	20,21
ONE1	integrand $\bar{F}_A \bar{F}_B$ of the second integral in Equation (2.2.37)	23

3.7 FORTRAN Symbols in FUNCTION TWO

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
X	integration variable $x = c_1 \bar{w}$	TWO 007
FAB	\bar{F}_A defined by Equation (2.2.28)	13
FBB	\bar{F}_B defined by Equation (2.2.29)	14,15
TWO	integrand in Equation (2.2.27)	17

3.8 FORTRAN Symbols in FUNCTION SEVEN

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Z	integration variable $z = c_1 y$	SEVN007
V	$c_1 u$	14
RCOE	non-rigid bottom K defined by Equation (2.3.5)	22
	rigid bottom K defined by Equation (2.3.4)	29
F	F defined in Equation (2.3.12)	34
SEVEN	integrand of Equation (2.3.10)	??
	A_2 defined by Equation (2.3.11)	
RT5	K_1	39
U1	$x + iy_1$	40
U2	$c_1^2 \omega_1^2$	41
U3	$c_1 \omega_1$	42
XB	$-c_1 \tau_m \cos \theta$	43

3.9 FORTRAN Symbols in SUBROUTINE EXE1

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
A	array a_i in Equation (2.5.19)	EXE1008
B	array b_i in Equation (2.5.19)	9
C	array a_i in Equation (2.5.18)	10,11
X	x	12

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
ANS	$\exp(x) E_1(x)$ for $x \geq 1$	EXE1014,15
	$\exp(x) E_1(x)$ for $0 \leq x < 1$	17,18

3.10 FORTRAN Symbols in SUBROUTINE EXEI

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Y	x	EXEI006
A	array of $1/nn!$ for $n = 2, 3, \dots, 7$ in Equation (2.5.20)	9
U	sum of the series in Equation (2.5.20)	11
ANS	$\exp(-x) Ei(x)$ using Equation (2.5.20)	12
ANS1	integral in Equation (2.5.21) evaluated using the Gaussian quadrature of FUNCTION FGI	15
ANS	$\exp(-x) Ei(x)$ using Equation (2.5.21)	16

3.11 FORTRAN Symbols in FUNCTION EXPO

<u>Symbol</u>	<u>Definition</u>
X	integration variable y in Equation (2.5.21)
P(1)	$x = (t - R_r/c_1)/G_r$
EXPO	integrand $\exp(y - x)/y$ in Equation (2.5.21)

3.12 FORTRAN Symbols in FUNCTION FGI

<u>Symbol</u>	<u>Definition</u>
A	lower limit of integration
B	upper limit of integration

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<u>Symbol</u>	<u>Definition</u>
K	number of subintervals into which the integration interval (A,B) is divided. The integral in each subinterval is evaluated using a 4 point Gaussian quadrature.
F	integrand of the integral to be evaluated
P	array used to transfer parameters to the function F
FGI	value of the integral of F between A and B

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APPENDIX A
 FORTRAN IV LISTING OF THE BOTTOM
 REFLECTION PROGRAM BOT REF

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BDCROTR,T600,CM70000.52400311,047,BRITT.
ATTACH(ABC,NOLBIN)
COPYN(0,DEF,ABC)
FTN(L)
LOAD(LGO)
REQUEST,TAPE99,LO,(CALCOMP/RING)
DEF.
' RECORD SEPARATOR =(7-8-9) PUNCH IN COLUMN 1
REWIND(ABC)
CALCM1,13,ABC
' RECORD SEPARATOR =(7-8-9) PUNCH IN COLUMN 1

      PROGRAM BOTREF(INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT,TAPE99)      BOTR001
C      BOTTOM REFLECTION PROGRAM (COG 6400 COMPUTER)      BOTR002
C      DIMENSION XX(1000),YY(1000)      BOTR003
C      COMMON /QXY/QX(1000),QY(1000)      BOTR004
C      COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID      BOTR004A
C      COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2      BOTR005
C      ADATF = DATE(0)      BOTR006
C      ICASF=1      BOTR007
C      PI=3.1415926      BOTR008
C
C      READ INPUT DATA (FORMAT -- 8F10.5)      BOTR009
C
C      WCH----EXPLOSIVE CHARGE WEIGHT (LBS OR KT)      BOTR010
C      CWATER--SOUND VELOCITY OF WATER (FT/SEC)      BOTR011
C      CROT---SOUND VELOCITY OF THE BOTTOM MATERIAL (FT/SEC)      BOTR012
C      CSHEAR--IF CSHEAR GT 0.5, IT IS THE SHEAR WAVE PROPAGATION      BOTR013
C      VELOCITY OF THE BOTTOM (FT/SEC). IF CSHEAR LE 0.5, IT IS      BOTR014
C      THE DIMENSIONLESS POISSON RATIO.      BOTR015
C      RHOJAT--DENSITY OF WATER (GM/CC)      BOTR016
C      RHOBOT--DENSITY OF BOTTOM MATERIAL (GM/CC)      BOTR017
C      PRECOE--COEFFICIENT OF PRESSURE SIMILITUDE EQUATION (PSI)      BOTR018
C      Z5-----CONTROL PARAMETER. Z5 GREATER THAN ZERO RESULTS IN A      BOTR019
C      SHORTER PRINT OUT.      BOTR020
C      PREFXP--EXPONENT OF PRESSURE SIMILITUDE EQUATION      BOTR021
C      THECOE--COEFFICIENT OF TIME CONSTANT SIMILITUDE EQUATION (SEC)      BOTR022
C      THEFXP--EXPONENT OF TIME CONSTANT SIMILITUDE EQUATION      BOTR023
C      STEPS--NUMBER OF POINTS IN P-T CURVE FOR ONE TIME CONSTANT      BOTR024
C      OURAT--DURATION OF PRESSURE TIME HISTORY IN MULTIPLES OF THE      BOTR025
C      TIME CONSTANT. (IF NEGATIVE, ITS ABSOLUTE VALUE IS THE      BOTR026
C      DURATION AFTER THE ARRIVAL OF THE BOTTOM REFLECTION      BOTR027
C      PEAK. IF POSITIVE, IT IS THE DURATION AFTER THE      BOTR028
C      DIRECT WAVE ARRIVAL.)      BOTR029
C      X1-----CALCOMP PLOT SCALING PARAMETER FOR THE Y-AXIS (PSI PER      BOTR030
C      INCH OF GRAPH)      BOTR031
C      X2-----SCALING PARAMETER FOR THE X-AXIS (MICROSECONOS PER      BOTR032
C      INCH OF GRAPH)      BOTR033
C      SLOPE---SLOPE OF BOTTOM FROM CHARGE TO GAUGE (DEGREES)      BOTR034
C
C      ADDITIONAL DATA IS READ IN STATEMENT 4 (CAR0 BOTR089)      BOTR035
C
C

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C   3 READ(5,554)WCH,CWATER,CBOT,CSHEAR,RHOWAT,RHOBOT,PRECOE,Z5      BOTR040
C   READ (5,556) PREEXP,THECOE,THEEXP,STEPS,DURAT,X1,X2,SLOPE      BOTR041
C   FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042      BOTR042
C
C   DURAT IN PRINT OUT IS THE DURATION AFTER THE DIRECT ARRIVAL      BOTR043
C   STORF ORIGINAL DURAT      BOTR044
C
C   XDURAT=DURAT      BOTR045
C
C   R=RHOWAT/RHOBOT      BOTR046
C
C   POISSON RATIO      BOTR047
C
C   IF CSHEAR IS 0.5 FT/SEC OR LESS. THE POISSON RATIO POISR IS SET      BOTR048
C   EQUAL. TO CSHEAR AND THE SHEAR VELOCITY IS CALCULATED.      BOTR049
C
C   IF(CSHEAR.LE.0.) GO TO 39      BOTR050
C   IF(CSHEAR.LE.0.5) GO TO 42      BOTR051
C   44 POISR=(0.5*CHOT**2-CSHEAR**2)/(CHOT**2-CSHEAR**2)      BOTR052
C   GO TO 41      BOTR053
C   42 POISR=CSHEAR      BOTR054
C   CSHEAR=CHOT*SQRT((0.5-POISR)/(1.-POISR))      BOTR055
C   GO TO 41      BOTR056
C   39 POISR=0.4      BOTR057
C
C   STONEY WAVE PROPAGATION VELOCITY      BOTR058
C
C   41 CALL STONL      BOTR059
C
C   THE GEOMETRY IS NOW READ IN (FORMAT -- 8F10.5)      BOTR060
C
C   BIGH----WATER DEPTH AT THE CHARGE (FT). ALSO USED AS A      BOTR061
C   CONTROL VARIABLE. SEE CARDS BOTR094-97 BELOW      BOTR062
C   D-----DEPTH OF THE CHARGE BELOW THE WATER SURFACE (FT)      BOTR063
C   DGAU----DEPTH OF THE GAUGE (FT)      BOTR064
C   SMALLR--HORIZONTAL RANGE BETWEEN CHARGE AND GAUGE (FT)      BOTR065
C   THOVAL--DESIRED RATIO BETWEEN THE BOTTOM REFLECTION INCIDENT AND      BOTR066
C   CRITICAL ANGLES. (IF THOVAL LE 0., THE INPUT GEOMETRY IS      BOTR067
C   NOT CHANGED.) SEE APPENDIX C OF NULTR 71-110.      BOTR068
C   Z1-----PARAMETER WHICH SELECTS THEORY. (SEE CARDS BOTR391-394)      BOTR069
C   Z2-----ARRIVAL TIME OF THE MAIN BOTTOM REFLECTION AFTER THE      BOTR070
C   DIRECT ARRIVAL (MICROSECONDS). GEOMETRY IS UNCHANGED IF      BOTR071
C   Z2 LE 0. SEE APPENDIX C OF NOLTR 71-110.      BOTR072
C   Z3-----PLOT CONTROL PARAMETER. A CALCOMP PLOT TAPE IS GENERATED      BOTR073
C   IF Z3 IS ZERO .      BOTR074
C
C   4 READ(5,554)BIGH,D,DGAU,SMALLR,THOVAL,Z1,Z2,Z3      BOTR075
C   FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042      BOTR076
C   AFTER COMPLETION OF EACH CASE PROGRAM CONTROL RETURNS TO      BOTR077
C   STATEMENT 4. DEPENDING ON BIGH, THE CALCULATIONS ARE CONTINUED AS      BOTR078
C   FOLLOWS      BOTR079
C   IF BIGH = 0 PROGRAM STOPS      BOTR080
C   IF BIGH IS POSITIVE COMPUTATION CONTINUES USING THE PRESENT INPUT      BOTR081
C   IF BIGH IS NEGATIVE PROGRAM TRANSFERS TO STATEMENT 3 WHERE      BOTR082
C   ANOTHER SET OF CHARGE, PHYSICAL CONSTANTS, ETC. ARE READ.      BOTR083
C
C   IF(BIGH)3,1000,6      BOTR084
1000 STOP      BOTR085
C   6 WRITE(6,510)ADATE      BOTR086
C   FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042      BOTR087
C   ADDITIONAL DATA IS READ IN FOR PTV CALCULATION (FORMAT -- 8F10.5)      BOTR088
C

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C   RAOUS--CYLINDER RAOUS IN FEET. THIS IS THE DRAFT OR CROSS-      BOTR101C
C   SECTIONAL RADIUS OF THE TARGET VESSEL.                                BOTR101D
C   APRNT--CONTROLS PRINTING IN SUBROUTINE PTV. THE ITERATIONS TO      BOTR101E
C   OBTAIN THE PTV ARE PRINTED OUT IF APRNT .LE. 0.                      BOTR101F
C   BOTR101G
C   BOTR101H
C   READ(5,554) RADIUS,APRNT
C   IPTV=0
C   A=0,
1006 WRITE(6,550) ICASE
C   WRITE(6,511)RIGH
C   WRITE(6,513)D
C   WRITE(6,512)DGAU
C   FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C   CHANGE OF DEPTH OF EXPLOSION AND OF GAUGE FOR GIVEN ARRIVAL
C   TIME Z2 OF THE BOTTOM REFLECTION PEAK AFTER THE DIRECT
C   ARRIVAL. THE BOTTOM REFLECTION INCIDENT ANGLE AND SMALLR
C   ARE UNCHANGED. SEE APPENDIX C OF NOLTR 71-110.
C   GEOMETRY IS UNCHANGED IF Z2 IS NEGATIVE OR ZERO.
C
C   -D2ACTU- IS THE ORIGINAL TOTAL DISTANCE BETWEEN THE GAGE AND THE
C   IMAGE CHARGE. -R2ACTU- IS THE ORIGINAL TOTAL SLANT DISTANCE FROM
C   THE GAGE TO THE IMAGE CHARGE AND -HG- IS THE NEW VALUE OF THE
C   HEIGHT OF THE GAGE ABOVE THE BOTTOM. NEW VALUES FOR -D- AND
C   -DGAU- ARE CALCULATED.
C
C   IF(Z2.LE.0.) GO TO 2
1  D2ACTU=2.* (BIGH-D)+D-DGAU
R2ACTU=SQRT(SMALLR**2+D2ACTU**2)
DELR=Z2*(1.E-06)*CWATER
C2002=DELR*(2.*R2ACTU-DELR)/D2ACTU**2
HG=0.5*D2ACTU*(1.-SQRT(1.-DR2002))
DGAU=BIGH-HG
D=BIGH-D2ACTU+HG
WRITE(6,553) Z2
WRITE(6,513)D
WRITE(6,512)DGAU
C
C   CHANGE OF GEOMETRY TO OBTAIN THE DESIRED RATIO
C   BETWEEN INCIDENT AND CRITICAL ANGLE=THOVAL
C   GEOMETRY IS UNCHANGED IF THOVAL IS LESS THAN OR EQUAL TO ZERO.
C   SEE APPENDIX C OF NOLTR 71-110.
2 IF(THOVAL.LE.0.) GO TO 5
7 TH=THOVAL* ASIN(CWATER/CBOT)
O2ACTU=SMALLR*COS(TH)/SIN(TH)
O=2.*BIGH-DGAU-D2ACTU
IF(BIGH-O) 8,9,9
8 D = RIGH
DGAU = BIGH - D2ACTU
9 WRITE(6,537)
WRITE(6,512)DGAU
WRITE(6,513)D
C
C   GEOMETRY
5 SMALLR=BIGH-D
C
C   -RACTU- IS THE SLANT DISTANCE BETWEEN CHARGE AND GAGE.
C   RACTU = SQRT((D-DGAU)**2+SMALLR**2)
C
C   CALCULATE HYDROSTATIC PRESSURE -PH-
PH=-14.7*DGAU/33.0-14.7
C

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C  -RS- IS THE REDUCED ARRIVAL TIME OF ACOUSTIC SURFACE REFLECTION.      BOTR156
C  RS=SQRT(SMALLR**2*(D+DGAU)**2)/RACTU                                BOTR157
C  EXPONENTIAL PULSE PEAK PRESSURE AND TIME CONSTANT CALCULATED          BOTR158
C  W13R=WCH**1.0/3.0/RACTU                                              BOTR159
C  REDR=1.0/W13R                                                       BOTR160
C  THETA=THECOE*(W13R)**1.0+THEEXP*CWATER                                BOTR161
C  PACT=PRECOE*(W13R)**PREEXP                                         BOTR162
C  TACT=RACTU/CWATER                                              BOTR163
C  THET=THETA*TACT*1000.                                              BOTR164
C  IF (SLOPE .EQ. 0.0) GO TO 10005                                     BOTR165
C  CHANGE OF GEOMETRY FOR SLOPING BOTTOM                                BOTR166
C  A = SLOPE/57.29578                                              BOTR167
C  HG = BIGH-DGAU                                              BOTR168
C  H1 = SMALLH*COS(A)                                              BOTR169
C  H2 = (HG-SMALLR*TAN(A))*COS(A)                                     BOTR170
C  IF ((H2.LT.0.0).OR.(SMALLR*TAN(A).GT.BIGH)) WRITE(6,555)          BOTR171
C  WRITE (6,514) SMALLR                                              BOTR172
C  WRITE (6,574)                                                       BOTR173
C  SMALLR=SMALLR*COS(A)+(D-DGAU)*SIN(A)                                BOTR174
C  SMALLH = H1                                              BOTR175
C  IF (H2 .GT. H1) D=D+H2-H1                                         BOTR176
C  DGAU = D+H1-H2                                              BOTR177
C  BIGH = D+H1                                              BOTR178
C  WRITE (6,511) BIGH                                              BOTR179
C  WRITE (6,512) DGAU                                              BOTR180
10005 DEZERO = D-DGAU                                              BOTR181
D2ACTU=2.0*SMALLH+DEZERO                                         BOTR182
R2ACTU=SQRT(D2ACTU**2+SMALLR**2)                                     BOTR183
C  INITIALIZATIONS                                              BOTR184
C  1040 FI=0.                                              BOTR185
C  VMID=0.                                              BOTR186
C  PRE=1.0                                              BOTR187
C  IP=1                                              BOTR188
C  NP=4                                              BOTR189
C  ZZDT=4.                                              BOTR190
C  RESIN=0.0                                              BOTR191
C  EFLUX=0.                                              BOTR192
C  FIMP=0.                                              BOTR193
C  POSINP=0.                                              BOTR194
C  IPREQ=1                                              BOTR195
C  PPRE=0.                                              BOTR196
C  XPPRF=0.                                              BOTR197
C  PMID=0.                                              BOTR198
C  XPMIN=0.                                              BOTR199
C  PD=0.                                              BOTR200
C  PS=0.                                              BOTR201
C  PGOT=0.                                              BOTR202
C  BASIC CONSTANTS OF GROUND WAVE                                BOTR203
C  CTWO=CWATER/CBOT                                              BOTR204
C  R2=R2ACTU/RACTU                                              BOTR205
C  THETAR=THETA/R2**THEEXP                                         BOTR206
C  THETD=THET/R2**THEEXP                                         BOTR207
C  PACTC=R2**PREEXP/R2                                         BOTR208
C  R1=R2**PREEXP                                              BOTR209
C  THETR=THETA/R1**THEEXP                                         BOTR210

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D2=0.2*ACTU/RACTU          B0TR220
SINTH=SMALLR/(RACTU*R2)    B0TR221
COSTH=D2/R2                 B0TR222
COSRM=COSTH/R2              B0TR223
IF(CTWO,GE,SINTH) GO TO 2000 B0TR224
GAM=SQRT(1.-CTWO**2)        B0TR225
D2R2=R2*(CTWO*SINTH+GAM*COSTH) B0TR226
GO TO 2005                  B0TR227
2000 D2R2=R2                B0TR228
C                                B0TR229
C      CALCULATION OF NEW DURATION IF DURAT IS READ IN NEGATIVE. B0TR230
C      (DURAT IS EXPLAINED IN CARDS B0TR028-32.) B0TR231
2005 IF(XDURAT,LT,0.) DURAT=(R2-1.)/THETA-XDURAT B0TR232
TSTOP=1.+DURAT*THETA        B0TR233
C                                B0TR234
SINAL=CWATER/CHOT            B0TR235
SIN2AL=SINAL**2              B0TR236
IF(SIN2AL-1.)811,811,812    B0TR237
811 COSAL=SQRT(1.-SIN2AL)    B0TR238
GO TO 813                   B0TR239
812 COSAL=-0.                B0TR240
813 SIN2TH=SINTH**2          B0TR241
IF(CSHEAR)15,15,14          B0TR242
14 SINBE=CWATER/CSHEAR       B0TR243
SIN2BE=SINBE**2              B0TR244
15 THE=57.2958* ASIN(SINTH) B0TR245
B0TR246
C                                B0TR247
C      CALCULATION OF PLANE WAVE REFLECTION COEFFICIENT, PHASE SHIFT, B0TR248
C      AND ANGLE OF S-WAVE. B0TR249
C                                B0TR250
IF(CSHEAR)30,30,50          B0TR251
B0TR252
C      CALCULATION FOR BOTTOM WITH NO SHEAR STRENGTH (NON-RIGID) B0TR253
C                                B0TR254
30 IF(SIN2TH=SIN2AL)33,32,31 B0TR255
B0TR256
C      SUPERCRITICAL REFLECTION (ANGLE OF INCIDENCE GREATER THAN THE B0TR257
C      CRITICAL ANGLE) B0TR258
C                                B0TR259
31 E=ATAN(B*SQRT(SIN2TH-SIN2AL)/COSTH) B0TR260
CR=1.                         B0TR261
IICA=1                         B0TR262
GO TO 88                      B0TR263
32 E2=0.                         B0TR264
CR=1.                         B0TR265
IICA=2                         B0TR266
GO TO 89                      B0TR267
B0TR268
C      SUBCRITICAL REFLECTION (ANGLE OF INCIDENCE LESS THAN THE B0TR269
C      CRITICAL ANGLE) B0TR270
C                                B0TR271
33 E2=0.                         B0TR272
AT=COSTH/(SQRT(SIN2AL-SIN2TH)*B) B0TR273
CR=(AT-1.)/(AT+1.)              B0TR274
IICA=3                         B0TR275
GO TO 89                      B0TR276
B0TR277
C      CALCULATION FOR BOTTOM WITH SHEAR (RIGID) B0TR278
C                                B0TR279
50 CA=COSTH*(1.-2.*SIN2TH/SIN2BE)**2/B B0TR280
CB=4.*COSTH*SIN2TH*(SIN2BE-SIN2TH)/B/SIN2BE**2 B0TR281
IF(SIN2TH-SIN2AL)60,32,51 B0TR282
51 ATA=CA/SQRT(SIN2TH-SIN2AL) B0TR283

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IF(SINTH-SINBE)52,55,57
52 BT=CB/SQRT(SIN2BE-SIN2TH)
CR=SQRT((ATA**2+(BT-1.)**2)/(ATA**2+(BT+1.)**2))
EA=ATAN((1.-BT)/ATA)
EB=ATAN((1.+BT)/ATA)
E2=EA+EB
IICA=4
GO TO 89
55 BTA=0.
GO TO 58
C
57 BTA=CB/SQRT(SIN2TH-SIN2BE)
58 E=ATAN(1./(ATA+BTA))
CR=1.
IICA=6
GO TO 88
C
60 E=0.
AT=CA/SQRT(SIN2AL-SIN2TH)
BT=CB/SQRT(SIN2BE-SIN2TH)
CR=(AT+BT-1.)/(AT+BT+1.)
IICA=7
C
88 E2=2.*E
89 EE=57.2958*E2
IF(CGHEAR,LE,0.) GO TO 92
90 IF(SINTH-SINBE)91,91,92
91 GAMMA= ASIN(SINTH/SINBE)
ANGA=57.2958*GAMMA
GO TO 95
C
92 ANGA=-0.
C
ANGLE OF P-WAVE
95 IF(SINTH-SINAL)293,293,294
293 THONF=57.2958* ASIN(SINTH/SINAL)
GO TO 295
294 THONF=-0.
C
FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS B0TR907-1042
295 WRITE(6,514)SMALLR
WRITE(6,504)WCH
WRITE(6,505)CWATER
WRITE(6,506)CBOT
WRITE(6,546)CSHEAR
WRITE(6,507)RHOWAT
WRITE(6,508)RHOBOT
WRITE(6,515)PREC0E
WRITE(6,503)Z5
WRITE(6,516)PREEXP
WRITE(6,517)THECOE
WRITE(6,518)THEEXP
WRITE(6,519)STEPS
WRITE(6,509)DURAT
WRITE(6,538)THOVAL
WRITE(6,584) X1
WRITE(6,585) X2
WRITE(6,500) SLOPE
WRITE(6,586) Z1
WRITE(6,587) Z2
WRITE(6,588) Z3
WRITE(6,568) RADIUS
WRITE(6,569) APRINT
WRITE(6,520)
B0TR284
B0TR285
B0TR286
B0TR287
B0TR288
B0TR289
B0TR290
B0TR291
B0TR292
B0TR293
B0TR294
B0TR295
B0TR296
B0TR297
B0TR298
B0TR299
B0TR299
B0TR300
B0TR301
B0TR302
B0TR303
B0TR304
B0TR305
B0TR306
B0TR307
B0TR308
B0TR309
B0TR310
B0TR311
B0TR312
B0TR313
B0TR314
B0TR315
B0TR316
B0TR317
B0TR318
B0TR319
B0TR320
B0TR321
B0TR322
B0TR323
B0TR324
B0TR325
B0TR326
B0TR327
B0TR328
B0TR329
B0TR330
B0TR331
B0TR332
B0TR333
B0TR334
B0TR335
B0TR336
B0TR337
B0TR338
B0TR339
B0TR340
B0TR341
B0TR342
B0TR343
B0TR344
B0TR344A
B0TR344B
B0TR345

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      WRITE(6,521)THE          B0TR346
      WRITE(6,573)CSTON        B0TR347
      WRITE(6,545)POISR        B0TR348
      WRITE(6,547)RS          B0TR349
      B0TR350
C      IF(1.=-SINAL)17,16,16  B0TR351
 16  ALPHA=57.2958* ASIN(SINAL) B0TR352
      WRITE(6,522)ALPHA        B0TR353
      GO TO 18                B0TR354
 17  WRITE(6,541)              B0TR355
 18  IF(CSHEAR)49,49,45       B0TR356
 45  IF(1.=-SINBE)47,46,46   B0TR357
 46  BETHA=57.2958* ASIN(SINBE) B0TR358
      WRITE(6,542)BETHA        B0TR359
      B0TR360
C      ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR WAVE B0TR361
C      IF(STNTH,LT,SINBE) .SHD2R2=0. B0TR362
      IF(STNTH,GE,SINBE) 1 SHD2R2=(SMALLR*SINBE+02ACTU*SQRT(1.=-SIN2BE)) B0TR363
 1 /RACTU
      WRITE(6,579) SHD2R2        B0TR364
      B0TR365
C      GO TO 49                B0TR366
 47  WRITE(6,543)              B0TR367
 49  WRITE(6,597)THONE        B0TR368
      WRITE(6,592)CR          B0TR369
      WRITE(6,594)ANGA        B0TR370
      WRITE(6,593)EE          B0TR371
      WRITE(6,523)D2R2        B0TR372
      WRITE(6,533)R2          B0TR373
      WRITE(6,525)RACTU       B0TR374
      WRITE(6,502)REDR        B0TR375
      WRITE(6,526)TACT        B0TR376
      WRITE(6,527)PACT        B0TR377
      WRITE(6,528)THETA       B0TR378
      WRITE(6,539)THET        B0TR379
      WRITE(6,548) THETAR      B0TR380
      WRITE(6,549) THETH      B0TR381
      WRITE(6,535)              B0TR382
      WRITE(6,551)SMALLH,0EZFR0,D2,COSAL,COSTH,SINTH B0TR383
      WRITE(6,532)              B0TR384
      FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS B0TR907-1042 B0TR385
      B0TR386
C      SELECTION OF THE THEORY B0TR387
      B0TR388
      B0TR389
      B0TR390
C      Z1=0. ROSENRAUM METHOD B0TR391
C      Z1=1. PLANE WAVE APPROXIMATION B0TR392
C      Z1=2. NOT USED IN THE PRESENT PROGRAM B0TR393
C      Z1=3. COMPLEX ARITHMETIC METHOD B0TR394
      B0TR395
C      Z1=AQS(Z1)              B0TR396
      IF(Z1=1,)800,801,802 B0TR397
 802  IF(Z1=3,)803,804,805 B0TR398
      B0TR399
C      SPHERICAL WAVE CAGNIARD-ROSENBAUM THEORIES B0TR400
      B0TR401
      B0TR402
      B0TR403
      B0TR404
C      NON-RIGID BOTTOM B0TR405
 820  IF(STNAL=1,)R30,831,832 B0TR406
      B0TR407
C      FAST BOTTOM B0TR408
 830  WRITE(6,560)            B0TR409

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GO TO 11                                BOTR410
C
C   SLOW BOTTOM                         BOTR411
  832 WRITE(6,561)                      BOTR412
  GO TO 11                                BOTR413
C
C   NO REFLECTION                        BOTR414
  831 WRITE(6,599)                      BOTR415
  GO TO 4                                 BOTR416
C
C   RIGID BOTTOM                         BOTR417
  821 IF(SINBE-1.)841,841,840          BOTR418
C
C   FAST SHEARWAVE                       BOTR419
  841 WRITE(6,562)                      BOTR420
  GO TO 11                                BOTR421
C
C   SLOW SHEARWAVE                        BOTR422
  840 WRITE(6,563)                      BOTR423
  GO TO 11                                BOTR424
C
C   PLANE WAVE APPROXIMATION            BOTR425
  801 WRITE(6,565)                      BOTR426
  GO TO 998                               BOTR427
C
C   Z1 = 2. IS NOT NEEDED FOR THE PRESENT PROGRAM
C
  803 WRITE(6,599)                      BOTR428
  GO TO 4                                 BOTR429
C
C   COMPLEX ARITHMETIC METHOD           BOTR430
C
  804 WRITE(6,566)                      BOTR431
C
C   CONSTANTS FOR SUBROUTINE SEVEN      BOTR432
  C2=CWATER**2                           BOTR433
  CBOT2=CAINT**2                         BOTR434
  CSHEAR2=CSHEAR**2                      BOTR435
  SINTH2=SINTH**2                         BOTR436
  CHSH2=4.*CSHEAR**3/CBOT                BOTR437
  C2SHR2=2.*CSHEAR**2                      BOTR438
  C4CB=C2**2/CBOT*B                      BOTR439
  GO TO 11                                BOTR440
C
C   Z1=4. IS NOT NEEDED FOR PRESENT PROGRAM
  805 WRITE(6,599)                      BOTR441
  GO TO 4                                 BOTR442
C
C   *****
C
C   SPHERICAL WAVE CAGNIARD-ROSENBAUM PRESSURE-TIME CALCULATIONS
C
C   PHASES AND TIME STEPS
C
  11 IF(D>R2-D2)10,20,20                BOTR451
C
C   ANGLE OF INCIDENCE GREATER THAN CRITICAL
  10 M=(R2-D2R2)*STEPS/THETA/4.          BOTR452
  IF((D2R2-1.0).GT.0.) M=(R2-1.0)*STEPS/THETA/4.  BOTR453
C
  BOTR454
  BOTR455
  BOTR456
  BOTR457
  BOTR458
  BOTR459
  BOTR460
  BOTR461
  BOTR462
  BOTR463
  BOTR464
  BOTR465
  BOTR466
  BOTR467
  BOTR468
  BOTR469
  BOTR470
  BOTR471
  BOTR472
  BOTR473

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M=4*M+5
C CALCULATE DT, INCREMENT OF REDUCED TIME T
12 DT=(R2-D2R2)/FLOAT(M)/2.
IF((D2R2-1.0).GT.0.) DT=(R2-1.0)/FLOAT(M)/2.
DT1=DT
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETAR)
WRITE(6,536)DT,EDT
WRITE(6,532)
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C -25- IS THE PRINTOUT CONTROL PARAMETER. IF -25- GREATER THAN
C ZERO A SHORTER PRINTOUT RESULTS, IF -25- EQUALS ZERO THE NORMAL,
C LONGER PRINTOUT IS GENERATED.
C
IF (75.GT. 0.0) GO TO 103
WRITE(6,530)
GO TO 100
103 WRITE(6,501)
GO TO 104
C
C ANGLE OF INCIDENCE LESS THAN OR EQUAL TO CRITICAL
C
20 MM=(D2-1.)*STEPS/THETA
MM=2*MM+4
C CALCULATE DT, INCREMENT OF REDUCED TIME T
DT=(D2-1.)/FLOAT(MM)
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETAR)
WRITE(6,536)DT,EDT
WRITE(6,532)
IF(Z5.GT.0.0) GO TO 104
WRITE(6,530)
GO TO 700
104 WRITE(6,501)
GO TO 700
C
C ANGLE OF INCIDENT WAVE LARGER THAN CRITICAL
C R2 LARGER THAN D2R2
C
100 IF(D2R2>0.9999)101,102,102
C
C PRECURSOR ARRIVES BEFORE DIRECT WAVE
C
101 T=D2R2
STPW=0.
N=10
GO TO 72
C
C PRESSURE CALCULATION IF DIRECT WAVE ARRIVES BEFORE PRECURSOR
C
102 T=1.0
STPW=0.
N=1
GO TO 71
110 N=12
114 T=T+2.*DT
IF(T.LT.D2R2) GO TO 71
117 T=D2R2
WRITE(6,534)
N=11
GO TO 71
C

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C      CALCULATION OF THE PRECURSOR                                B0TR538
C
150 N=2                                         B0TR539
152 T=T+DT                                     B0TR540
      CALL STPWA(T,VMID,Z1,NP)                   B0TR541
C      EVERY FIFTH STEP RECALCULATE STPW WITH TWICE THE INTEGRATION B0TR542
C      POINTS,NP, TO CHECK THE INTEGRATION ERROR.                  B0TR543
      IF(MOD(IP,5).NE.0) GO TO 155                 B0TR544
      IF(NP.EQ.16) GO TO 155                         B0TR545
      NP2=2*NP                                     B0TR546
      CALL STPWA(T,VMID2,Z1,NP2)                   B0TR547
      ERROR=ABS((VMID-VMID2)/VMID2)                B0TR548
      IF(ERROR.LT.0.005) GO TO 155                 B0TR549
C      DOUBLE NUMBER OF INTEGRATION POINTS          B0TR550
      NP=NP2                                       B0TR551
      VMID=VMID2                                     B0TR552
155 T=T+DT                                     B0TR553
      CALL STPWA(T,STPW,Z1,NP)                     B0TR554
      FI=FI*EDT**2+((PRE*EDT+4.*VMID)*EDT+STPW)*DST B0TR555
      PRE=STPW                                     B0TR556
      GO TO 70                                     B0TR557
159 IF((IPRES.LT.0).OR.(T.LT.(R2-6.1*DT))) GO TO 150 B0TR558
C      CALCULATION OF PRECURSOR NEAR SINGULARITY          B0TR559
C
200 DT=DT/ZZDT                                 B0TR560
      M=(R2-T)/DT/4.                                B0TR561
      M=4*M+5                                     B0TR562
      DT=(R2-T)/FLOAT(M)/2.                         B0TR563
      DTACT=2.*DT*TACT/3.                           B0TR564
      DST=DT/3.                                    B0TR565
      EOT=EXP(-DT/THETAR)                           B0TR566
      N=9                                         B0TR567
201 IF((IPRES.LT.0).OR.(T.LT.(R2-3.1*DT))) GO TO 152 B0TR568
      TR1=T/R2                                     B0TR569
      V=SQRT(1.-TR1**2)                           B0TR570
      DSV=V/12.*R2                                B0TR571
      TR2=SQRT(1.-(0.75*V)**2)                     B0TR572
      TR3=SQRT(1.-(0.5*V)**2)                     B0TR573
      TR4=SQRT(1.-(0.25*V)**2)                    B0TR574
      T1=T                                         B0TR575
      T2=R2*TR2                                    B0TR576
      T3=R2*TR3                                    B0TR577
      T4=R2*TR4                                    B0TR578
      EDT1=EXP(-(T3-T1)/THETAR)                   B0TR579
      EDT2=EXP(-(T3-T2)/THETAR)                   B0TR580
      EDT3=EXP(-(R2-T3)/THETAR)                   B0TR581
      EDT4=EXP(-(R2-T4)/THETAR)                   B0TR582
C
202 CALL STPWA(T2,VMID,Z1,16)                   B0TR583
      CALL STPWA(T3,STPW,Z1,16)                     B0TR584
      FI=FI*EDT1+((PRE*EDT1*V/TR1+3.*VMID*EDT2*V/TR2+STPW*0.5*V/TR3)*DSV B0TR585
C
      PRE=STPW                                     B0TR586
      T=T3                                         B0TR587
      N=3                                         B0TR588
      GO TO 70                                     B0TR589
C
210 CALL STPWA(T4,VMID,Z1,16)                   B0TR590
      FI=FI*EDT3+((PRE*EDT3*0.5*V/TR3+VMID*EDT4*V/TR4)*DSV B0TR591
      PRE=0.                                         B0TR592
      T=R2                                         B0TR593
      STPW=CR*(1,E+30)*SIGN(1.+ZE)                B0TR594
      WRITE(6,540)                                  B0TR595

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N=4
GO TO 70

C
C
C      CALCULATION OF MAIN BOTTOM REFLECTION NEAR SINGULARITY
C

300 IF(DT.GT.(DT1/ZZDT/2.)) GO TO 301
DT=DT1/ZZDT/2.
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETAR)
NP=16
301 T6=R2+4.*DT1
DTR=DT/R2
U=SQRT((1.+2.*DTR)**2-1.)
DSU=U/12.*R2
C
TR2=SQRT(1.+(0.25*U)**2)
TR3=SQRT(1.+(0.50*U)**2)
TR4=SQRT(1.+(0.75*U)**2)
TR5=1.+2.*DTR
T2=TR2*R2
T3=TR3*R2
T4=TR4*R2
T5=TR5*R2
C
EDT1=EXP(-(T3-R2)/THETAR)
EDT2=EXP(-(T3-T2)/THETAR)
EDT3=EXP(-(T5-T3)/THETAR)
EDT4=EXP(-(T5-T4)/THETAR)
C
302 CALL STPWB(T2,VMID,Z1,16)
CALL STPWB(T3,STPW,Z1,16)
FI=FI*EDT1+(VMID*U*EDT2 /TR2+STPW*0.5*U/TR3)*DSU
PRE=STPW
T=T3
N=5
GO TO 71
C
305 CALL STPWB(T4,VMID,Z1,16)
CALL STPWB(T5,STPW,Z1,16)
FI=FI*EDT3+(0.5*PRE*U*EDT3/TR3+3.*VMID*U*EDT4/TR4+STPW*U/TR5)*DSU
PRE=STPW
T=T5
N=6
GO TO 71
C
308 N=13
DTACT=2.*TACT*DT/3.
310 IF((IFRES.LT.0).OR.(T.LT.T6)) GO TO 410
320 DT=DT1
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETAR)

C
C
C      CALCULATION OF MAIN BOTTOM REFLECTED WAVE
C

400 N=7
410 T=T+DT
CALL STPWB(T,VMID,Z1,NP)
C      EVERY TENTH STEP RECALCULATE STPW WITH HALF THE INTEGRATION
C      POINTS NP.  IF ERROR IS LESS THAN .005 REDUCE NP BY HALF.
IF(MOD(IP,10).EQ.0) GO TO 415

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IF(NP.EQ.4) GO TO 415
NP2=NP/2
CALL STPWB(T,VMID2,Z1,NP2)
ERROR=ABS((VMIL-VMID2)/VMID)
IF(ERROR.LT.0.005) NP=NP2
415 T=T+DT
CALL STPWB(T,STPW,Z1,NP)
FI=FI*EDT**2+((PRE*EDT**4.*VMID)*EDT+STPW)*DST
PRE=STPW
GO TO 71

C
C   CALCULATION OF BOTTOM REFLECTION FOR TIMES LESS THAN OR EQUAL TO
C   THE ARRIVAL TIME OF THE PEAK IF THE ANGLE OF INCIDENCE IS LESS
C   THAN OR EQUAL TO THE CRITICAL ANGLE. THE DIRECT WAVE ARRIVES
C   BEFORE OR TOGETHER WITH THE BOTTOM REFLECTION.
C

700 T=1.0
N=8
707 STPW=0.
GO TO 71
706 N=14
708 T=T+DT*2.0
IF(T+2.0*DT.LE.R2) GO TO 71
710 T=R2
STPW=CR/R2
PRE=0.
WRITE(6,550)
NP=8
N=7
GO TO 71

C
C   CALCULATION OF DIRECT WAVE -PD-, BOTTOM REFLECTION -PBOT-,
C   SURFACE REFLECTION -PS- AND TOTAL PRESSURE -P- .
C

70 IF(T.LT.1.0) GO TO 72
71 PD=PACT*EXP((1.0-T)/THETA)
IF(T.LT.RS) GO TO 72
PS=-PACT/R1*EXP((RS-T)/THETSR)
72 PBOT1=PBOT
FTHETA=FI/THETAR
PBOT=PACT/PACTC*(STPW-FTHETA)
73 TIME=TACT*(T-1.0)
P= P0 + PS + PBOT

C
C   TEST TO INSURE THAT THE ABSOLUTE PRESSURE (P+HYDROSTATIC) .GE. 0.
C
P=AMAX1(P,PH)
IF(T.GT.TSTOP) N=15

C
C   CALCULATION OF IMPULSE= FIMP AND ENERGY FLUX= EFLUX
C   CALCULATION OF POSITIVE IMPULSE= POSIMP
C

6004 XP=AMAX1(P,0.)
GO TO (6030,6007,6020,6030,6040,6050,6007,6030,6007,
1 6030,6010,6007,6007,6007,6060),N
6007 IF(IPRFS.LT.0) GO TO 6070
PMID=P
XPMID=XP
GO TO 6090
6010 XXDT=TACT/3.0*(T-TPRE)
FIMP=3.0*(P+PPRE)*XXDT /2.0+FIMP
POSIMP=3.0*(XP+XPPRE)*XXDT/2.0+POSIMP

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EFLUX=EFLUX+(ABS(P)*P+ABS(PPRF)*PPRE)*XXDT/2./RHOAT/CWATER*3./      B0TR730
1 2.3066      B0TR731
C THE CONVERSION FACTOR 2.3066 IS NECESSARY FOR EFLUX TO HAVE      B0TR732
C UNITS IN-PSI      B0TR733
PPRE=P      B0TR734
XPPRE=XP      B0TR735
IPRES=1      B0TR736
GO TO 6092      B0TR737
6020 DTACT=2.0*DSV*TACT      B0TR738
PPRE=PPRE*V/TR1      B0TR739
XPPRE=XPPRE*V/TR1      B0TR740
PMIN=P*V/2./TR3      B0TR741
XPMID=XP*V/2./TR3      B0TR742
GO TO 6090      B0TR743
6030 PEND=0.      B0TR744
XPEND=0.      B0TR745
GO TO 6072      B0TR746
6040 DTACT=2.0*DSU*TACT      B0TR747
PPRE=0.      B0TR748
XPPRE=0.      B0TR749
PMIN=P*U/2./TR3      B0TR750
XPMID=XP*U/2./TR3      B0TR751
GO TO 6090      B0TR752
6050 PEND=P/TR5      B0TR753
XPEND=XP*U/TR5      B0TR754
GO TO 6072      B0TR755
6060 IF(IPRES,GT,0) GO TO 6010      B0TR756
6070 PEND=P      B0TR757
XPEND=XP      B0TR758
6072 FIMP=FIMP+(PPRE+4.*PMID+PEND)*DTACT      B0TR759
POSIMP=POSIMP+(XPPRE+4.*XPMID+XPEND)*DTACT      B0TR760
EFLUX=EFLUX+(ABS(PPRE)*PPRE+4.*ABS(PMID)*PMID+ABS(PEND)*PEND)*      B0TR761
1 DTACT/RHOAT/CWATER/2.3066      B0TR762
PPRE=P      B0TR763
XPPRE=XP      B0TR764
6090 IPRES=-1*IPRES      B0TR765
6092 IF(IPRES,GT,0) TPRES=T      B0TR766
C WHEN D2R2,LT,1.0 THE BOTTOM REFLECTION IS NOT CALCULATED AT      B0TR767
C THE DIRECT WAVE ARRIVAL TIME T=1.0. IN ORDER TO PLOT THE      B0TR768
C INSTANTANEOUS RISE OF THE DIRECT SHOCK AT T=1.0, THE BOTTOM      B0TR769
C REFLECTION IS OBTAINED BY LINEAR INTERPOLATION. PLOT POINTS ARE      B0TR770
C THEN CALCULATED FOR THE TOP AND BOTTOM OF THE SHOCK FRONT.      B0TR771
IF((IP,EQ,1).OR.(IP,GT,1000)) GO TO 7002      B0TR772
IF((TIME,GT,0.0).AND.(XX(IP-1),LT,0.0)) GO TO 6095      B0TR773
GO TO 7002      B0TR774
6095 XX(IP)=0.      B0TR775
IF(T,NE,R2) PBOTD=PBOT1+XX(IP-1)*(PBOT-PBOT1)/(XX(IP-1)-TIME*1.E6)      B0TR776
IF(T,EG,R2) PBOTD=PBOT1      B0TR777
YY(IP)=AMAX1(PBOTD,PH)      B0TR778
XX(IP+1)=0.      B0TR779
YY(IP+1)=AMAX1(PBOTD+PACT,PH)      B0TR780
WRITE(6,559) YY(IP+1)      B0TR781
IP=IP+2      B0TR782
C PRINT ROUTINE      B0TR783
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS B0TR907-1042      B0TR784
7002 IF (Z5,GT,0.0) GO TO 7003      B0TR785
WRITE(6,551) T,STPW,FTHETA,PD,TIMF,P,FIMP      B0TR786
WRITE(6,598) PROT,PS,VMID,PRE,EFLUX      B0TR787
GO TO 7004      B0TR788
7003 WRITE(6,551) T,PBOT,EFLUX,PD,TIME,P,FIMP      B0TR789
GO TO 7005      B0TR790
C REDUCED IMPULSE      B0TR791
7004 RFIMP=FIMP/W13R/RACTU      B0TR792
B0TR793

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C      REDUCED POSITIVE IMPULSE          B0TR794
      RPOSIM=POSIMP/W13R/RACTU          B0TR795
C      REDUCED ENERGY FLUX           B0TR796
      REFLUX=EFLUX/W13R/RACTU          B0TR797
      WRITE(6,556) RESID,RFIMP,REFLUX,POSIMP,RPOSIM B0TR798
      7005 IF(IP.GT.1000) GO TO 999 B0TR799
      XX(IP)=TIME*1.0E6 B0TR800
      YY(IP)=P B0TR801
      IP=IP+1 B0TR802
C      BOTTOM REFLECTION TIME AND PRESSURE STORED IN QX AND QY FOR B0TR802A
C      PTV CALCULATION.          B0TR802B
      IF(T.LT.D2R2) GO TO 7001 B0TR802C
      IPTV=IPTV+1 B0TR802D
      QX(IPTV)=TACT*(T-D2R2) B0TR802E
      QY(IPTV)=AMAX1(PBOT,PH) B0TR802F
C      7001 GO TO (110,159,210,300,305,308,410,706,201,159,159,114, B0TR803
      1 310,708,999),N B0TR804
C      999 IPMAX=IP=1 B0TR805
C      CALCOMP PLOT TAPE GENERATED IF Z3 = 0. B0TR806
C      IF(Z3.NE.0.) GO TO 997 B0TR807
      CALL PLOT1(XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CBOT,POISR,Z1) B0TR808
      997 ICASE=ICASE+1 B0TR809
C      CALCULATION OF INPUTS FOR SUBROUTINE PTV. B0TR810
      IF(RADIUS.LE.0.) GO TO 4 B0TR811
      1997 TIMER2=TACT*(R2-D2R2) B0TR812
      XT3=TIMER2-2.*DT*TACT B0TR813
      XT4=0.8*TIMER2 B0TR813A
      XT5=QX(IPTV) B0TR813B
      IPTV=IPTV+1 B0TR813C
      QX(IPTV)=1.0E20 B0TR813D
      QY(IPTV)=0. B0TR813E
      COSA=COSTH*COS(A)+SINTH*SIN(A) B0TR813F
C      CALCULATION OF PEAK TRANSLATIONAL VELOCITY (PTV). B0TR813G
      CALL PTV(TIMER2,XT3,XT4,XT5,RADIUS,30.,APRINT,COSA,RHOWAT, B0TR813H
      1 CWATER,TIME1,PTV1,PTV2) B0TR813I
      IF(APRINT.GT.0.) WRITE(6,570) TIME1,PTV1,PTV2 B0TR813J
      GO TO 4 B0TR813K
C      **** B0TR813L
C      PLANE WAVE APPROXIMATION USING EQUATIONS OF ARONS AND YENNIE B0TR813M
C      998 IF(CSHEAR.GT.0.) GO TO 1005 B0TR813N
      WRITE(6,590) B0TR814
      GO TO 1007 B0TR815
      1005 WRITE(6,596) B0TR816
C      FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS B0TR907-1042 B0TR817
C      SELECTION OF TIME STEP B0TR818
C      1007 WRITE(6,591) B0TR819
      IF(D2R2.GE.R2) GO TO 1020 B0TR820
C      CALCULATION OF TIME STEP FOR SUPERCRITICAL REFLECTION B0TR821
      1010 M=(R2-D2R2)*STEPS/THETA/2.+1.0 B0TR822
      M=2*M-1 B0TR823
      IF(M.LE.4) GO TO 1020 B0TR824
      1012 DT=(R2-D2R2)/FLOAT(M) B0TR825

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IF(D2R2-1.)1014,1015,1015          BOTR838
1014 T=D2R2                         BOTR839
    GO TO 1700                      BOTR840
1015 T=1.                           BOTR841
    GO TO 1700                      BOTR842
C
C      CALCULATION OF TIME STEP FOR SUBCRITICAL REFLECTION  BOTR843
1020 DT=THETA/STEPS                BOTR844
    T=1.                           BOTR845
:700 IF(T.GE.1.0) GO TO 1731        BOTR846
1730 SW=0.                          BOTR847
    GO TO 1732                      BOTR848
C
C      INCIDENT (DIRECT) WAVE RESPONSE  BOTR849
1731 SW=EXP(-(T-1.0)/THETA)        BOTR850
1732 XE1=0.                         BOTR851
    XE1=0.                         BOTR852
    IF(E2.NE.0.) GO TO 1738          BOTR853
1733 IF(T.GE.R2) GO TO 1736        BOTR854
1735 PRFL=0.                         BOTR855
    GO TO 1745                      BOTR856
C
C      PRESSURE RESPONSE FOR SUBCRITICAL REFLECTION  BOTR857
1736 PRFL=CR*EXP(-(T-R2)/THETAR)  BOTR858
    GO TO 1745                      BOTR859
1738 TBTH=(T-R2)/THETAR           BOTR860
    IF(TBTH) 1741,1742,1743        BOTR861
1741 CALL EXE1(TBTH,XE1)          BOTR862
C
C      PRESSURE RESPONSE FOR PRECURSOR OF SUPERCRITICAL REFLECTION  BOTR863
PRFL=CR*SIN(E2)*XE1/PI            BOTR864
    GO TO 1745                      BOTR865
C
C      PRESSURE RESPONSE AT SINGULARITY  BOTR866
1742 PRFL=(1.E+30)*SIGN(1.,EE)    BOTR867
    GO TO 1745                      BOTR868
1743 CALL EXEI(TBTH,XE1)          BOTR869
C
C      PRESSURE RESPONSE FOR MAIN SUPERCRITICAL BOTTOM REFLECTION  BOTR870
PRFL=CR*(EXP(-TBTH)*COS(E2)-XE1*SIN(E2)/PI)  BOTR871
1745 PRFL=PRFL/PACTC/R2          BOTR872
    PBOT=PACT*PRFL                BOTR873
1710 TIME=TACT*(T-1.)             BOTR874
    PD=PACT*SW                    BOTR875
    IF(T.GE.RS) PS=-PACT/R1*EXP((RS-T)/THETSR)  BOTR876
C
C      TOTAL PRESSURE (NEGATIVE VALUE LIMITED TO PH)  BOTR877
P=AMAX1(PD+PBOT+PS,PH)           BOTR878
C
C      OUTPUT ROUTINE  BOTR879
    WRITE(6,551)T,PBOT,PD,PS,TIME,P  BOTR880
C
C      IF(IP.GT.1000) GO TO 4000  BOTR881
7100 XX(IP)=TIME*1.0E6            BOTR882
    YY(IP)=P                      BOTR883
    IP=IP+1                      BOTR884
C
C      BOTTOM REFLECTION TIME AND PRESSURE STORED IN QX AND QY FOR  BOTR885
C      PTV CALCULATION.          BOTR886
    IF(T.LT.D2R2) GO TO 7102      BOTR887
    IPTV=IPTV+1                  BOTR888
    QX(IPTV)=TACT*(T-D2R2)        BOTR889
    QY(IPTV)=AMAX1(PBOT,PH)        BOTR890
7102 TDT=T                        BOTR891
    T=T+DT                        BOTR892
    IF(ABS(TDT-R2).LT.1.5*DT) T=TDT+0.2*DT  BOTR893
    IF((TDT.LT.R2).AND.(T.GT.R2)) T=R2        BOTR894
                                                BOTR895

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IF((TDT,LT,1.0),AND,(T,GT,1.0)) T=1.0          BOTR896
IF(T,LT,TSTOP) GO TO 1700                      BOTR897
4000 IPMAX=IP-1                                  BOTR898
C   CALCOMP PLOT TAPE IS GENERATED IF Z3 = 0.    BOTR899
IF(Z3,NE,0.) GO TO 4001                      BOTR900
CALL PLOT1(XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CBOT,POISR,Z1)  BOTR901
4001 ICASE=ICASE+1                            BOTR902
IF(RADIUS,GT,0.) GO TO 1997                  BOTR902A
GO TO 4                                     BOTR903
C   *****
C   *****                                     BOTR904
C   *****                                     BOTR905
C   *****                                     BOTR906
500 FORMAT(10X,26HSLOPE OF BOTTOM IN DEGREES,54X,6HSLOPE=,E16.8 )  BOTR907
501 FORMAT(5X,12HREDUCED TIME,23X,11HENERGY FLUX,26X,4HTIM=,11X,  BOTR908
 18HPRESSURE,10X,7HIMPULSE/10X,1HT,16X,4HPBOT,31X,2HPD,12X,  BOTR909
 27HSECONDS,14X,3HPSI / )                      BOTR910
502 FORMAT(10X,39HREDUCED SLANT DISTANCE (RACTU/WCH**1/3),41X,6HREDR=  BOTR911
 1,1E15.8)                                     BOTR912
503 FORMAT(10X,60HPRINT OUT CONTROL PARAMETER (Z5.GT.0. FOR SHORTER PR  BOTR913
 1INT OUT),20X,3HZ5=,E16.6)                  BOTR914
504 FORMAT(10X,40HWEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT),40X,5HWCH=  BOTR915
 1 E15.8)                                     BOTR916
505 FORMAT(10X,36HVELOCITY OF SOUND IN WATER IN FT/SEC,44X,8HCWATER= ,  BOTR917
 1E15.8)                                     BOTR918
506 FORMAT(10X,37HVELOCITY OF SOUND IN BOTTOM IN FT/SEC,43X,6HCBOT= ,  BOTR919
 1E15.8)                                     BOTR920
507 FORMAT(10X,25HDENSITY OF WATER IN GM/CC,55X,8HRHOWAT= ,E15.8 )  BOTR921
508 FORMAT(10X,26HDENSITY OF BOTTOM IN GM/CC,54X,8HRHOBOT= ,E15.8 )  BOTR922
509 FORMAT(10X,51HDURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA,  BOTR923
 129X,7HDURAT= 1E15.8 )                      BOTR924
510 FORMAT(1H1,52X,17HBOTTOM REFLECTION,10X,4HDATE,2X,1A10 )        BOTR925
511 FORMAT(10X,20HDEPTH OF WATER IN FT,60X,6HBIGH= 1E15.8 )        BOTR926
512 FORMAT(10X,20HDEPTH OF GAUGE IN FT,60X,6HDGAU= 1E15.8 )        BOTR927
513 FORMAT(10X,24HDEPTH OF EXPLOSION IN FT,56X,3HD= 1E15.8 )        BOTR928
514 FORMAT(10X,50HHORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT,3  BOTR929
 10X,8HSMALLR= 1E15.8 )                      BOTR930
515 FORMAT(10X,41HCOEFFICIENT OF SW PRESSURE FORMULA IN PSI,39X,  BOTR931
 1 BHPRECOE= ,E15.8 )                         BOTR932
516 FORMAT(10X,31HEXPONENT OF SW PRESSURE FORMULA,49X,BHPREEXP= ,1E15.  BOTR933
 18 )                                         BOTR934
517 FORMAT(10X,50HCOEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS,  BOTR935
 1 30X,BHTHECOE= ,E15.8 )                      BOTR936
518 FORMAT(10X,36HEXPONENT OF SW TIME CONSTANT FORMULA,44X,BHTHEEXP= ,  BOTR937
 11E15.8 )                                     BOTR938
519 FORMAT(10X,31HNUMBER OF SUBDIVISIONS OF THETA 49X,7HSTEPS= ,1E15.8  BOTR939
 1 )                                         BOTR940
520 FORMAT(1H0,47X,25HCHARACTERISTIC MAGNITUDES / )                BOTR941
521 FORMAT(10X,33HANGLE OF INCIDENT WAVE IN DEGREES,47X,5HTHE= ,1E15.8  BOTR942
 1 )                                         BOTR943
522 FORMAT(10X,45HCRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES,35X,7H  BOTR944
 1ALPHA= 1E15.8 )                           BOTR945
523 FORMAT(10X,33HREDUCED TIME OF PRECURSOR ARRIVAL,47X,6HD2R2= ,1E15.  BOTR946
 18 )                                         BOTR947
524 FORMAT(10X,35HREDUCED TIME OF GROUND WAVE ARRIVAL,45X,4HR2= ,1E15.  BOTR948
 18/ )                                         BOTR949
525 FORMAT(10X,68HSLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERIST  BOTR950
 1IC LENGTH IN FT,12X,7HRACTU = ,1E15.8 )      BOTR951
526 FORMAT(10X44HCHARACTERISTIC TIME=RACTU/CWATER IN SECONDS,36X,6HTAC  BOTR952
 1T= ,1E15.8 )                           BOTR953
527 FORMAT(10X,58HCHARACTERISTIC PRESSURE=FREE WATER SW PEAK PRESSURE  BOTR954
 1IN PSI 22X,6HPACT= ,1E15.8)                 BOTR955
528 FORMAT(10X,38HREDUCED TIME CONSTANT OF INCIDENT WAVE,42X,7HTHETA=  BOTR956
 1,1E15.8 )                           BOTR957
529 FORMAT(1H1)                                BOTR958

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530 FORMAT(5X,12HREDUCED TIME,3X,17HSTEPWAVE RESPONSE,3X,12HCONVOLUTI B0TR959
 10N-,25X,4HTIME,11X,8HPRESSURE,10X,7HIMPULSE/
 2 10X,1HT,16X,4HSTPW,11X,8HFI/THETA , 12X,2HP0,12X,7HSECONDS B0TR960
 3 ,14X,3HPSI,7X,11HENERGY FLUX/6X,10HSECOND ROW, 12X,2HP0,12X,7HSECONDS B0TR961
 4 11X,4HPR0T ,13X,2HPS,15X,4HVMID,14X,3HPRE/ 7X,9HTHIRD ROW,9X, 7H B0TR962
 5RESIDUE,7X,15HREDUCED IMPULSE,3X,13HREDUCED EFLUX,6X,6HPCSIMP, B0TR963
 6 24X,14HREOUCED POSIMP//) B0TR964
 532 FORMAT(1H0) B0TR965
 533 FORMAT(10X,45HREDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE, B0TR966
 1 35X,4HR2= E15.8) B0TR967
 534 FORMAT(50X,20HARRIVAL OF PRECURSOR /) B0TR968
 535 FORMAT(46X,28HCONSTANTS OF THE CALCULATION/ 9X,6HSMALLW,11X,6HDEZE B0TR969
 1R0,13X,2H02,14X,5HCOSAL,12X,5HCOSTH,12X,5HSINTH /) B0TR970
 536 FORMAT(50X,2HDT,14X,3HEDT/ 40X,2E17.7 /) B0TR971
 537 FORMAT(24X,73HINPUT CHANGED SO THAT RATIO BETWEEN INCIDENT AND CR1 B0TR972
 1TICAL ANGLE IS TH0VAL/) B0TR973
 538 FORMAT(10X,49HDESIRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE,31 B0TR974
 1X,8HTH0VAL= 1E15.8) B0TR975
 539 FORMAT(10X,39HACTUAL SW TIME CONSTANT IN MILLISECONDS,40X,6H THET= B0TR976
 1,1E15.8) B0TR977
 540 FORMAT(1H0,39X,10H***** /39X,39HARRIVAL OF GROUNDWAVE PEAK(SI B0TR978
 1NGULARITY) / 40X,10H***** /) B0TR979
 541 FORMAT(10X,44HCRITICAL ANGLE OF COMPRESSION WAVE IMAGINARY) B0TR980
 542 FORMAT(10X,38HCRITICAL ANGLE OF SHEARWAVE IN DEGREES,42X,7HBETHA= B0TR981
 11F7.3) B0TR982
 543 FORMAT(10X,37HCRITICAL ANGLE OF SHEARWAVE IMAGINARY) B0TR983
 545 FORMAT(10X,13HPOISSON RATIO,67X,7HPOISR= ,1E15.8) B0TR984
 546 FORMAT(10X,31HVELOCITY OF SHEARWAVE IN FT/SEC,49X,8HCSHEAR= 1E15.8 B0TR985
 1) B0TR986
 547 FORMAT(10X,34HREDUCED TIME OF SURFACE REFLECTION,46X,4HRS= 1E15.8 B0TR987
 1) B0TR988
 548 FORMAT(10X,46HREDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE ,34X B0TR989
 1 ,8HTHETAR= ,E15.8) B0TR990
 549 FORMAT(10X,51HBOTTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS B0TR991
 1 ,29X,7HTHETR= E15.8 /) B0TR992
 550 FORMAT(53X,11HRUN NUMBER ,115/ 57X,5HINPUT//) B0TR993
 551 FORMAT(1X,7E17.7) B0TR994
 553 FORMAT(24X,59HGEOMETRY CHANGED SO THAT ARRIVAL TIME OF GROUNDWAVE B0TR995
 1IS Z2 = ,E15.8 /) B0TR996
 554 FORMAT(8F10.5) B0TR997
 555 FORMAT(1H0,10H***** /5X,15H** WARNING ** //5X,35HSLOPE AND B0TR998
 1GEOMETRY ARE INCONSISTENT /5X,52HCOMPUTATION CONTINUES BUT RESULTS B0TR999
 2 MAY BE MEANINGLESS //) B0T1000
 556 FORMAT(18X,4E17.7,17X,E17.7 /) B0T1001
 558 FORMAT(39X,26HARRIVAL OF GROUNDWAVE PEAK//) B0T1002
 559 FORMAT(1H0,10X,27HARRIVAL OF DIRECT WAVE P = ,E17.7 //) B0T1003
 560 FORMAT(50X,21HFAST NON-RIGID BOTTOM) B0T1004
 561 FORMAT(50X,21HSLOW NON-RIGID BOTTOM) B0T1005
 562 FORMAT(44X,33HRIGID BOTTOM WITH FAST SHEAR WAVE) B0T1006
 563 FORMAT(44X,33HRIGID BOTTOM WITH SLOW SHEAR WAVE) B0T1007
 C B0T1008
 C B0T1009
 565 FORMAT(48X,24HPLANE WAVE APPROXIMATION) B0T1010
 566 FORMAT(45X,25HCOMPLEX ARITHMETIC METHOD) B0T1011
 567 FORMAT(52X,16HROSENBAUM METHOD) B0T1012
 568 FORMAT(10X,21HCYLINDER RADIUS IN FT ,59X,7HRADIUS= ,E15.8) B0T1013
 569 FORMAT(10X,43HPRINT CONTROL PARAMETER (FULL PRINT OUT IN , B0T1013A
 1 31HSURROUTINF PTV IF APRINT.LE.0.) ,6X,8HAPRINT= E15.8) B0T1013B
 570 FORMAT(1H0,18X,5HTIME1,4X,14HPTV(SUBMERGE0),2X,12HPTV(SURFACE)/ B0T1013C
 1 19X,5H(SEC),7X,8H(FT/SEC),7X,8H(FT/SEC)/11X,3E15.6) B0T1013D
 573 FORMAT(10X,35HVELOCITY OF STONLEY WAVF IN FT/SEC 45X,7HCTSON= B0T1013E
 1E15.8) B0T1014
 574 FORMAT(5X,10H***** ,5X,47HCALCULATION GEOMETRY CHANGED FOR SL B0T1015
 10PING BOTTOM) B0T1016
 B0T1017

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579 FORMAT(10X,5SHREDUCED ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR W BOT1018
 1AVE,25X,8HSHD2R2= E15.8) BOT1019
 584 FORMAT(10X,52HSCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH) BOT1020
 1 28X,4HX1= ,E15.8) BOT1021
 585 FORMAT(10X,61HSCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH BOT1022
 10F GRAPH) 19X,4HX2= ,E15.8) BOT1023
 586 FORMAT(10X,29HPARAMETER THAT SELECTS THEORY,51X,4HZ1= ,E15.8) BOT1024
 587 FORMAT(10X,43HARRIVAL TIME OF GROUND WAVE IN MICROSECONDS 37X,4HZ2 BOT1025
 1= ,E15.8) BOT1026
 588 FORMAT(10X,55HPLOT CONTROL PARAMETER (Z3 = 0, MEANS PLOTS ARE WANT BOT1027
 1ED) ,25X,4HZ3= ,E15.8) BOT1028
 590 FORMAT(43X,39HARONS-YENNIE APPROACH NON-RIGID BOTTOMS /) BOT1029
 591 FORMAT(5X,12HREDUCED TIME,3X,17HBOTTOM REFLECTION,4X,9HSHOCKWAVE, BOT1030
 1 4X,18HSURFACE REFLECTION,5X,4HTIME,8X,14HTOTAL PRESSURE/10X, BOT1031
 2 1HT,15X,4HPBOT,14X,2HPD,15X,2HPS,12X,7HSECONDS,9X, BOT1032
 3 7HP (PSI) /) BOT1033
 592 FORMAT(10X,22HREFLECTION COFFICIENT,58X,4HCR= 1E15.8) BOT1034
 593 FORMAT(10X,30HANGLE OF PHASESHIFT IN DEGREES ,50X,4HEE= 1E15.8) BOT1035
 594 FORMAT(10X,39HANGLE OF SHEARWAVE IN BOTTOM IN DEGREES,41X,6HANGA= BOT1036
 1 1E15.8) BOT1037
 596 FORMAT(37X,47HARONS-YENNIE APPROACH EXTENDED TO RIGID BOTTOMS /) BOT1038
 597 FORMAT(10X,43HANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES,37X,8HTH BOT1039
 1EONE= ,E15.8) BOT1040
 598 FORMAT(18X,4E17.7,17X,E17.7) BOT1041
 599 FORMAT(10X,42HINPUT INCONSISTENT. COMPUTATION SUPPRESSED/) BOT1042
 C END BOT1043
 C END BOT1044

C*****SUBROUTINE STONL*****
 C
 C
 C CALCULATION OF PROPAGATION VELOCITY OF STONLEY WAVE.
 C
 C
 C SUBROUTINE STONL
 C COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID STON001
 C COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2 STON002
 C
 C IF(CSHEAR.LE.0.) GO TO 416 STON003
 4 C12=CWATER**2 STON004
 C32=CBOT**2 STON005
 C42=CSHEAR**2 STON006
 C
 C ITERATION PROCESS STON007
 C IF(CWATER.GT.CSHEAR) GO TO 2 STON008
 1 Y2=CWATER**2 STON009
 GO TO 3 STON010
 2 Y2=CSHEAR**2 STON011
 3 CK=Y2/1000. STON012
 FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(STON013
 1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2) STON014
 Y2=Y2-CK STON015
 400 DO 410 IR=1,999 STON016
 C
 C FS STORED STON017
 FS=FY STON018
 C
 C STONLEY WAVE VELOCITY STON019
 C
 FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(STON020
 1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2) STON021
 C STON022
 STON023
 STON024
 STON025
 STON026
 STON027
 STON028
 STON029
 STON030
 STON031
 STON032
 STON033
 STON034

```

C
  IF(FY)412,415,408
  408 Y2=Y2-CK
  410 CONTINUE
  WRITE(6,401) CWATER,CBOT,CSHEAR,B,Y2,FS,FY
  STOP
C
C   FY IS NEGATIVE
C
C   FALSE POSITION OR SECANT METHOD ITERATION
  412 YS=Y2+CK
  DO 450 I=1,50
  YSS=Y2
  IF(ABS((YS-Y2)/YS).LT.1.0E-7) GO TO 415
  440 Y2=YS+FS*(Y2-YS)/(FS-FY)
  FS=FY
  YS=YSS
C
  FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2)
C
  450 CONTINUE
  WRITE(6,402) CWATER,CBOT,CSHEAR,B,Y2,YS,FS,FY
  STOP
C
C   RESULT
  415 CSTON=SQRT(Y2)
  RETURN
  416 CSTON=-0.
  RETURN
C
  401 FORMAT(20X, 42HFIRST ITERATION FOR CSTON DID NOT CONVERGE//1 30H CWATER,CBOT,CSHEAR,B,Y2,FS,FY // 1P7E16.6)
  402 FORMAT(20X, 43HSECOND ITERATION FOR CSTON DID NOT CONVERGE//1 33H CWATER,CBOT,CSHEAR,B,Y2,YS,FS,FY // 1P8E14.6)
C
  END

```

```

C****SUBROUTINE STPWA****
C
C   PRECURSOR CALCULATION USING CAGNIARD METHOD.
C
C   SUBROUTINE STPWA(T,STPW,CONTR,K)
  DIMENSION P(30)
  COMMON B,COSAL,CC3TH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID
  COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2
  EXTERNAL ONE,SEVEN
  DATA AA,BB,SQ2/-1.57079633,1.57079633,1.41421356/
  TR=T/R2
  V=SQRT(1.-TR**2)
  5 IF(CONTR.EQ.3.) GO TO 100
C
C   CALCULATION OF THE PRECURSOR USING CAGNIARD-ROSENBAUM INTEGRALS.
C
  P(9)=0.
  XM=COSTH*TR*SINTH*V
  P(1)=COSAL-XM
  P(2)=4.*V*SINTH/P(1)
  P(5)=COSAL+XM

```

NOLTR 71-110

```

C FACTOR=SQ2*B/(R2*BB) STPA025
C STPW=FACTOR*FGI(AA,BB,K,ONE,P) STPA026
C RETURN STPA027
C STPA028
C STPA029
C STPA030
C STPA031
C STPA032
C CALCULATION OF THE PRECURSOR USING COMPLEX ARITHMETIC METHOD STPA033
C STPA034
100 P(1)=0. STPA035
P(2)=TR STPA036
P(3)=CWATER/CBOT STPA037
P(4)=SINTH*TR-COSTH*V STPA038
FACTOR=R2*BB STPA039
C ANS2=SEVEN(P(4),P) STPA040
C STPA041
C STPW=(ANS2+FGI(P(3),P(4),K,SEVEN,P))/FACTOR STPA042
C STPA043
C RETURN STPA044
C END STPA045

```

```

IF(CBOT,GT,CWATER) GO TO 11                               STPB040
C SLOW NON-RIGID BOTTOMS                                STPB041
C SIGM=SQRT((CWATER/CBOT)**2-1.)                         STPB042
C STPW=TERM1-SQ2*FACTOR*FGI(0.,SIGM,K,TWO,P)           STPB043
C RETURN                                                 STPB044
C
C FAST NON-RIGID BOTTOMS                                STPB045
11 STPW=TERM1+FACTOR*FGI(0.,COSAL,K,ONE,P)             STPB046
C RETURN                                                 STPB047
C
C RIGID BOTTOMS                                         STPB048
C
C STONELEY POLE RESIDUE                                STPB049
C
12 TERM1=1./R2                                         STPB050
CWS2=(CWATER/CSHEAR)**2                                 STPB051
SK=CWATER/CSTON                                         STPB052
SK2=SK**2                                              STPB053
XG1=SQRT(ABS(SK2-1.))                                 STPB054
XG3=SQRT(ABS(SK2-(CWATER/CBOT)**2))                   STPB055
XG4=SQRT(ABS(SK2-CWS2))                               STPB056
XSA=R2**2*(TR**2-SK2*COSTH**2)                      STPB057
XSF=(2.*R2**2*TR*COSTH*XG1)**2                      STPB058
XNUM=XG1*((CWS2/2.-SK2)**2-SK2*XG3*XG4)-B*XG3*CWS2**2/4. STPB059
XDEN=((CWS2/2.-SK2)**2-SK2*XG3*XG4)/XG1-XG1*(2.*CWS2-4.*SK2+2.*XG3) STPB060
1*XG4+SK2*(XG4/XG3+XG3/XG4))+B*CWS2**2/4./XG3          STPB061
RESID=-SQ2*SQRT(ABS((SQRT(ABS(XSA**2+XSF))-XSA)/(XSA**2+XSF))) STPB062
1)*XNUM/XDEN/XG1                                      STPB063
C TERM1=1./R2+RESID                                     STPB064
C
IF(CSHEAR,GT,CWATER) GO TO 50                           STPB065
C
C SLOW SHEAR WAVE                                       STPB066
C
30 SIG2=SQRT(CWS2-1.)                                 STPB067
STPW=TERM1+FACTOR*(FGI(0.,COSAL,K,ONE,P)-CWS2**2*SQ2/4.* * STPB068
1 FGI(0.,SIG2,K,ONE,P))                               STPB069
RETURN                                                 STPB070
C
C FAST SHEAR WAVE                                       STPB071
C
50 STPW=TERM1+FACTOR*FGI(0.,COSAL,K,ONE,P)             STPB072
C
99 RETURN                                              STPB073
C
C CALCULATION OF THE MAIN REFLECTION USING COMPLEX ARITHMETIC METHOD STPB074
C
100 P(1)=COSTH*SQRT(TR**2-1.)                         STPB075
P(2)=TR                                                 STPB076
P(3)=0.                                                 STPB077
P(4)=SINTH*TR                                         STPB078
FACTOR=R2*BB                                           STPB079
C
ANS2=SEVEN(P(4),P)                                    STPB080
STPW=(ANS2+FGI(P(3),P(4),K,SEVEN,P))/FACTOR          STPB081
C
RETURN                                                 STPB082
END                                                 STPB083

```

```

*****FUNCTION ONE*****
C
C
C     INTEGRAND OF THE CAGNIARD-ROSENBAUM INTEGRAL FOR ALL FAST BOTTOMS, ONE 001
C     EXCEPT FOR PART OF THE MAIN REFLECTION OF A BOTTOM WITH SLOW SHEAR ONE 002
C
C
C     FUNCTION ONE(X,P) ONE 003
C     DIMENSION P(30) ONE 004
C     COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID ONE 005
C     COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2 ONE 006
C
C
C     P ARRAY CALCULATED IN SUBROUTINES STPWA AND STPWB ONE 007
C     TEST FOR PRECURSOR PHASE ONE 008
C     IF(P(9).GT.0.) GO TO 2 ONE 009
C
C     PRECURSOR PHASE ONE 010
C
C     SINX=SIN(X) ONE 011
C     W=(SINX+P(1)+P(5))/2. ONE 012
C     XC2=COSAL**2-W**2 ONE 013
C     IF(XC2.LT.0.) XC2=0. ONE 014
C     FX=(1.-SINX)*SQRT(((COSAL+W)*P(1))/(1.+SINX+P(2))) ONE 015
C     GO TO 3 ONE 016
C
C     MAIN REFLECTED WAVE ONE 017
C
C     2 W=X ONE 018
C     XC2=COSAL**2-W**2 ONE 019
C     IF(XC2.LT.0.) XC2=0. ONE 020
C     RT1=SQRT(XC2/(P(8)+(W-P(7))**2)) ONE 021
C     RT2=SQRT(XC2/(P(8)+(W+P(7))**2)) ONE 022
C     FX=RT1-RT2 ONE 023
C
C     RELATIONS FOR PRECURSOR AND MAIN WAVE ONE 024
C
C     3 IF(CSHEAR.EQ.0.) GO TO 110 ONE 025
C
C     RIGID BOTTOMS ONE 026
C     CSW=CSHEAR/CWATER ONE 027
C     CSW2=CSW**2 ONE 028
C     FRCS=CSW2*(W**2-1.)*1. ONE 029
C
C     XA=W*(1.-2.*CSW2*(1.-W**2))**2 ONE 030
C     XB=4.*W*CSW2*CSW*(1.-W**2)*SQRT(XC2*A8S(FRCS)) ONE 031
C
C     IF(FRCS.GE.0.) GO TO 22 ONE 032
C     ONE=FX*(XA-XB)/((XA-XB)**2+B*W*XC2) ONE 033
C     RETURN ONE 034
C
C     22 XC=B*SQRT(XC2) ONE 035
C     ONE=FX*XA/(XA**2+(XB+XC)**2) ONE 036
C     RETURN ONE 037
C
C     NON-RIGID BOTTOMS ONE 038
C
C     110 ONE=FX*W/(W**2+B*B*XC2) ONE 039
C     RETURN ONE 040
C
C     END ONE 041

```

```

C*****FUNCTION ONE1****          ONE1001
C                                     ONE1002
C                                     ONE1003
C                                     ONE1004
C                                     ONE1005
C                                     ONE1006
C                                     ONE1007
C                                     ONE1008
C                                     ONE1009
C                                     ONE1010
C                                     ONE1011
C                                     ONE1012
C                                     ONE1013
C                                     ONE1014
C                                     ONE1015
C                                     ONE1016
C                                     ONE1017
C                                     ONE1018
C                                     ONE1019
C                                     ONE1020
C                                     ONE1021
C                                     ONE1022
C                                     ONE1023
C                                     ONE1024
C                                     ONE1025
C                                     ONE1026

C
C   INTEGRAND OF THE SECOND CAGNIARD-ROSENBAUM INTEGRAL OCCURRING FOR
C   THE MAIN REFLECTION OF A BOTTOM WITH A SLOW SHEAR WAVE.          ONE1004
C   ONE1005
C   ONE1006
C   ONE1007
C   ONE1008
C   ONE1009
C   ONE1010
C   ONE1011
C   ONE1012
C   ONE1013
C   ONE1014
C   ONE1015
C   ONE1016
C   ONE1017
C   ONE1018
C   ONE1019
C   ONE1020
C   ONE1021
C   ONE1022
C   ONE1023
C   ONE1024
C   ONE1025
C   ONE1026

C
C   FUNCTION ONE1(X,P)          ONE1008
C   DIMENSION P(30)          ONE1009
C   COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID ONE1010
C   COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CR,SINTH2 ONE1011
C
C
C   1 CWS2=(CWATER/CSHEAR)**2          ONE1012
C   SIG22=CWS2-1.          ONE1013
C   XAB=X*(CWS2/2.-1.-X**2)**2          ONE1014
C   XBB=X*(1.+X**2)*SQRT((COSAL**2+X**2)*(SIG22-X**2))          ONE1015
C   XCB=B*CWS2**2*SQRT(COSAL**2+X**2)/4.          ONE1016
C
C   FAB=SQRT(X**2+COSAL**2)*XBB/((XAB+XCB)**2+XBB**2)          ONE1017
C   FBB=SQRT((SQRT((X**2+P(11))**2+P(12))+X**2-P(13))/((X**2+P(11))**2
C   1+P(12)))          ONE1018
C   P ARRAY CALCULATED IN SUBROUTINE STPWB          ONE1019
C   ONE1=FA8*FB8          ONE1020
C
C   99 RETURN          ONE1021
C   END          ONE1022
C
C

```

```

C*****FUNCTION TWO****          TWO 001
C                                     TWO 002
C                                     TWO 003
C                                     TWO 004
C                                     TWO 005
C                                     TWO 006
C                                     TWO 007
C                                     TWO 008
C                                     TWO 009
C                                     TWO 010
C                                     TWO 011
C                                     TWO 012
C                                     TWO 013
C                                     TWO 014
C                                     TWO 015
C                                     TWO 016
C                                     TWO 017
C                                     TWO 018
C                                     TWO 019
C                                     TWO 020

C
C   CAGNIARD-ROSENBAUM INTEGRAND FOR SLOW NON-RIGID BOTTOMS          TWO 003
C
C
C   FUNCTION TWO(X,P)          TWO 007
C   DIMENSION P(30)          TWO 008
C   COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID TWO 009
C   COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CR,SINTH2 TWO 010
C
C   SIGM2=(CWATER/CBOT)**2-1.
C   FA8=X*SQRT(SIGM2-X**2)/((1.-B**2)*X**2+SIGM2*B**2)          TWO 011
C   FBB=SQRT((SQRT((X**2+P(11))**2+P(12))+X**2-P(13))/((X**2+P(11))**2
C   1+P(12)))          TWO 012
C   P ARRAY CALCULATED IN SUBROUTINE STPWB          TWO 013
C   TWO=FA8*FB8          TWO 014
C
C   RETURN          TWO 015
C   END          TWO 016
C
C

```

```

C*****FUNCTION SEVEN****          SEVN001
C                                     SEVN002
C                                     SEVN003
C                                     SEVN004
C                                     SEVN005
C                                     SEVN006
C                                     SEVN007
C                                     SEVN008
C                                     SEVN009
C                                     SEVN010
C                                     SEVN011
C                                     SEVN012

C
C   INTEGRAND OF CAGNIARD INTEGRAL USING COMPLEX ARITHMETIC.          SEVN003
C
C
C   FUNCTION SEVEN(Z,P)          SEVN007
C   DIMENSION P(30)          SEVN008
C   COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID SEVN009
C   COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CR,SINTH2 SEVN010
C   COMPLEX F,RCOE,Y1,Y3,V,RT1,RT2,RT5,W,U1,U2,U3,U4 SEVN011
C   COMPLEX V2,XY1,XW SEVN012


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```

C      P ARRAY CALCULATED IN SUBROUTINES STPWA AND STPWB      SEVN013
V=CMPLX(P(1),Z)      SEVN014
V2=V*V      SEVN015
RT1=CSQRT(1.+V2)      SEVN016
RT2=CSQRT(V2*C80T2+C2)      SEVN017
IF(CSHEAR.GT.0.) GO TO 20      SEVN018
C      SEVN019
C      NON-RIGID BOTTOMS      SEVN020
Y3=B/C80T*RT2      SEVN021
RCOE=(RT1-Y3)/(RT1+Y3)      SEVN022
GO TO 40      SEVN023
C      SEVN024
C      RIGID BOTTOMS      SEVN025
20 XY1=C2SHR2*V2+C2      SEVN026
Y1=HT1*(XY1*CSH*V2*RT2*CSQRT(V2*CSH2+C2))      SEVN027
Y3=C4CB*RT2      SEVN028
RCOF=(Y1-Y3)/(Y1+Y3)      SEVN029
40 IF(Z.EQ.P(4)) GO TO 50      SEVN030
C      INTEGRAND CALCULATION      SEVN031
XW=P(2)-COSTH*RT1      SEVN032
W=V2*SINTH2+XW*XW      SEVN033
F=V/RT1/CSQRT(W)      SEVN034
SEVEN=REAL(F*(RCOE-RT5))      SEVN035
RETURN      SEVN036
C      SEVN037
C      CALCULATION OF ANS2      SEVN038
50 RT5=RCOE      SEVN039
U1=CMPLX(P(1),P(3))      SEVN040
U2=1.+U1*U1      SEVN041
U3=CSQRT(U2)      SEVN042
XB=-P(2)*COSTH      SEVN043
U4=RT5*CLOG((RT1+XB)/(CSQRT(U2+2.*XR*U3+P(2)**2-SINTH2)+U3+XB))      SEVN044
SEVEN=AIMAG(U4)      SEVN045
RETURN      SEVN046
END      SEVN047

```

```

C*****SUBROUTINE EXE1****      EXE1001
C      EXE1002
C      EXE1003
C      CALCULATION OF EXPONENTIAL INTEGRAL E1 TIMES EXP(-Y) , NEGATIVE Y      EXE1004
C      EXE1005
SUBROUTINE EXE1(Y,ANS)      EXE1006
DIMENSION A(4),B(4),C(6)      EXE1007
DATA A/ 8.5733287,18.059017,8.6347609,0.26777373 /      EXE1008
DATA B/ 9.5733223,25.632956,21.0996531,3.9584969 /      EXE1009
DATA C/ -0.57721566,0.99999193,-0.24991055,0.05519968,-0.00976004      EXE1010
1 .0.00107857 /      EXE1011
X=-Y      EXE1012
IF(X.LT.1.0) GO TO 10      EXE1013
ANS=(A(4)+X*(A(3)+X*(A(2)+X*(A(1)+X))))/(B(4)+X*(B(3)+X*(B(2)+      EXE1014
1 X*(B(1)+X))))/X      EXE1015
RETURN      EXE1016
10 ANS=EXP(X)*(C(1)+X*(C(2)+X*(C(3)+X*(C(4)+X*(C(5)+X*C(6))))))-      EXE1017
1 ALOG(X)      EXE1018
RETURN      EXE1019
END      EXE1020

```

```

C****SUBROUTINE EXEI****
C
C
C   CALCULATION OF EXPONENTIAL INTEGRAL EI TIMES EXP(-Y) , POSITIVE Y
C
C   SUBROUTINE EXEI(Y,ANS)
C   DIMENSION P(10),A(6)
C   EXTERNAL EXP0
C   DATA A/.25,.05555556,.01041667,.00166667,.00023148,.00002834 /
C   IF(Y.GT.0.5) GO TO 10
C   U=Y*(1.+Y*(A(1)+Y*(A(2)+Y*(A(3)+Y*(A(4)+Y*(A(5)+Y*A(6)))))))
C   ANS=(0.57721566+ALOG(Y)+U)*EXP(-Y)
C   RETURN
10  P(1)=Y
    ANSI=FGI(1.,Y,4,EXP0,P)
    ANS=ANS1+1.8951178 *EXP(-Y)
    RETURN
    END

EXEI001
EXEI002
EXEI003
EXEI004
EXEI005
EXEI006
EXEI007
EXEI008
EXEI009
EXEI010
EXEI011
EXEI012
EXEI013
EXEI014
EXEI015
EXEI016
EXEI017
EXEI018

C****FUNCTION EXP0****
C
C
C   INTEGRAND OF EXPONENTIAL INTEGRAL
C
C   FUNCTION EXP0(X,P)
C   DIMENSION P(10)
C   EXP0=EXP(X-P(1))/X
C   P(1) IS THE ARGUMENT OF THE EXPONENTIAL INTEGRAL
C   RETURN
C   END

EXP0001
EXP0002
EXP0003
EXP0004
EXP0005
EXP0006
EXP0007
EXP0008
EXP0009
EXP0010
EXP0011

C****FUNCTION FGI****
C
C
C   THIS SUBPROGRAM INTEGRATES THE FUNCTION F BETWEEN THE LIMITS
C   A AND B USING A FOUR-POINT GAUSSIAN QUADRATURE IN EACH OF THE
C   K SUBINTERVALS. P IS AN ARRAY USED TO TRANSFER PARAMETERS TO THE
C   FUNCTION F.
C
C   FUNCTION FGI(A,B,K,F,P)
C   DIMENSION V(4),W(2),SUM(4),P(1)
C   DATA V/ -.861136311594053,-.339981043584856,
1 -.339981043584856,.861136311594053 /
C   DATA W/.347854845137454,.652145154862546 /
C   SUM(1)=0.0
C   SUM(2)=0.0
C   SUM(3)=0.0
C   SUM(4)=0.0
C   H=(B-A)/FLOAT(K)
C   H2=H/2.
C   AA=A+H2
C   DO 20 L=1,K
C   DO 10 I=1,4
C   X=H2*V(I)+AA
10  SUM(I)=SUM(I)+F(X,P)
20  AA=AA+H
    FGI=H2*(W(1)*(SUM(1)+SUM(4))+W(2)*(SUM(2)+SUM(3)))
    RETURN
    END

FGI 001
FGI 002
FGI 003
FGI 004
FGI 005
FGI 006
FGI 007
FGI 008
FGI 009
FGI 010
FGI 011
FGI 012
FGI 013
FGI 014
FGI 015
FGI 016
FGI 017
FGI 018
FGI 019
FGI 020
FGI 021
FGI 022
FGI 023
FGI 024
FGI 025
FGI 026
FGI 027
FGI 028

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```

C*****SUBROUTINE PLOT1*****
C
C PLOTTING SUBROUTINE WHICH GENERATES A PLOT TAPE FOR
C CALCOMP PLOTTER
C
C SUBROUTINE PLOT1 (XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CROT,POISR,Z1)
C DIMENSTON XX(1000),YY(1000),BCDX(2),BCDY(2),TITLE0(2),TITLE1(2),
C 1TITLE2(2),TITLE4(3),TITLE5(2),TITLE6(2),TITLE7(2),TITLE8(2),
C 2TITLE9(2)
C DATA RCDX/10HTIME (MICR,10HOSEC)      /,
C 18CDY/10HPRESSURE (.10HPSI)      /,
C 2TITLE1/10HBTM REF,10HLECTION      /,
C 3TITLE2/10HCAGNIARD-R,10HOSENBAUM      /,
C 4TITLE3/10HPLANE WAVE/,,
C 5TITLE4/10HPLANE WAVE,10H USING CON,10HV. INT.      /,
C 6TITLE5/10HCOMPLEX AR,10HITHMETIC      /,
C 7TITLE6/10HANGLE OF I,10HNCIDENCE      /,
C 8TITLE7/10HCHARGE WEI,10HGHT (LBS)      /,
C 9TITLE8/10HCBOT (FT/S,10HEC)      /
C DATA TITLE9/10HPOISSON RA,10HTIO      /,
C 1TITLE0(1)/10HDATE      /
C 1TITLE0(2)=ADATE
C YMAX=6.*X1
C YMIN=-3.*X1
C YLIN=(YMAX-YMIN)/X1
C XLMAX=90.
C XLMIN=1.
C IYYMAX=0
C DO 4 I=1,IPMAX
C IF(YY(I)-YMIN)1,1,2
C 1 YY(I)=YMIN
C GO TO 4
C 2 IF(YY(I)-YMAX)4,3,3
C 3 YY(I)=YMAX
C IYYMAX=I
C 4 CONTINUE
C XMIN=XX(1)
C XMAX=XX(IPMAX)
C CALL SCAL (X2,XMIN,XMAX,XLMIN,XLMAX,XLIN)
C IF(XMIN)5,10,10
C 5 IF(XMAX)10,10,6
C 6 XS=XMIN
C XN=1.
C 7 IF(ABS(XS)-1.E-36)9,9,71
C 71 IF(XS)8,9,100
C 8 XS=XMIN+X2*XN
C XN=XN+1.
C GO TO 7
C 9 XN=XN-1.5
C YN=5.16
C CALL CALCM1(IPMAX,XX,YY,0,XMIN,XMAX,YMIN,YMAX,XLIN,YLIN,TITLE,0,BC
C 1DX,15,BCDY,0,FLOAT,18) PLOT052
C CALL SYMBL4(XN,YN,.14,BCDY,90,.14) PLOT053
C GO TO 11
C 10 CALL CALCM1(IPMAX,XX,YY,0,XMIN,XMAX,YMIN,YMAX,XLIN,YLIN,TITLE,0,BC PLOT056
C 1DX,15,RCDY,14,FLOAT,18) PLOT057
C 11 IF(IYYMAX)100,12,13 PLOT058
C 12 XS=(XMAX-XMIN)/X2-3.
C GO TO 14
C 13 XS=(XX(IYYMAX)-XMIN)/X2+1.
C 14 CALL SYMBL4 (XS,9.,.14,TITLE1,0.,20) PLOT062
C 15 IF(Z1-1.)17,18,16 PLOT063
C 16 IF(Z1-3.)19,20,100 PLOT064

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```

17 CALL SYMBL4 (XS,8.7,.14,TITLE2,0.,20)          PLOT065
    GO TO 21
18 CALL SYMBL4 (XS,8.7,.14,TITLE3,0.,10)          PLOT066
    GO TO 21
19 CALL SYMBL4 (XS,8.7,.14,TITLE4,0.,30)          PLOT067
    GO TO 21
20 CALL SYMBL4 (XS,8.7,.14,TITLE5,0.,20)          PLOT068
21 CALL SYMBL4 (XS,8.4,.14,TITLE6,0.,20)          PLOT069
    XN=XS+.14*18.+.2
    CALL NUMBR(XN,8.4,.14,THE,0.,2)                PLOT070
    CALL SYMBL4 (XS,8.1,.14,TITLE7,0.,20)          PLOT071
    XN=XS+.14*19.+.2
    CALL NUMBR(XN,8.1,.14,WCH,0.,5)                PLOT072
    CALL SYMBL4 (XS,7.8,.14,TITLE8,0.,20)          PLOT073
    XN=XS+.14*13.+.2
    CALL NUMBR(XN,7.8,.14,CBOT,0.,2)               PLOT074
    CALL SYMBL4 (XS,7.5,.14,TITLE9,0.,20)          PLOT075
    XN=XS+.14*13.+.2
    CALL NUMBR(XN,7.5,.14,POISR,0.,5)              PLOT076
    CALL SYMBL4 (XS,7.2,.14,TITLE0,0.,20)          PLOT077
    CALL CALCM1(0,0.)
    RETURN
100 WRITE(6,22)
    CALL CALCM1(0,0.)
    STOP
22 FORMAT(1H1,10X,34H PLOTTING ERROR IN SUBROUTINE PLOT1 //)
    END

```

```

C*****SUBROUTINE SCAL***#
C
C      SUBROUTINE FOR SCALING PLOTS
C
C      SUBROUTINE SCAL (XSCALE,XMIN,XMAX,XLMIN,XLMAX,XLIN)
C      DIMENSION X(6)
C      SXMAX=XMAX
C      SXMIN=XMIN
C      1 IJ=1
C      2 IR (XSCALE)3,3,6
C
C      DETERMINATION OF SCALE RANGE
C
C      3 M1= ALOG10(ABS(XMIN))
C      M2= ALOG10(ABS(XMAX))
C      IF (IABS(IARS(M1))-IABS(M2))=1)5,4,4
C      4 XM=M1
C      XM=10.**XM
C      GO TO 7
C      5 XM=M1-1
C      XM=10.**XM
C      GO TO 7
C      6 XM=XSCALE
C
C      SCALE FACTORS ALLOWED
C
C      7 X(1)=1.*XM
C      X(2)=2.*XM
C      X(3)=2.5*XM
C      X(4)=5.*XM
C      X(5)=7.5*XM
C      X(6)=10.*XM

```

```

SCAL001
SCAL002
SCAL003
SCAL004
SCAL005
SCAL006
SCAL007
SCAL008
SCAL009
SCAL010
SCAL011
SCAL012
SCAL013
SCAL014
SCAL015
SCAL016
SCAL017
SCAL018
SCAL019
SCAL020
SCAL021
SCAL022
SCAL023
SCAL024
SCAL025
SCAL026
SCAL027
SCAL028
SCAL029
SCAL030
SCAL031
SCAL032
SCAL033

```

```

C IF (XSCALE)8,8,24 SCAL034
C AUTOMATIC SCALING SCAL035
C
C 8 DO 21 I=1,6 SCAL036
C DETERMINATION OF SCALED MINIMUM SCAL037
C
C XMIN=SXMIN SCAL038
C XMAX=SXMAX SCAL039
C NSCALE=XMIN/X(I) SCAL040
C IF (NSCALE)11,9,12 SCAL041
C 9 IF (XMIN)11,13,10 SCAL042
C 10 XMIN=0. SCAL043
C GO TO 13 SCAL044
C 11 XN=NSCALE-1 SCAL045
C XMIN=XN*X(I) SCAL046
C GO TO 13 SCAL047
C 12 XN=NSCALE SCAL048
C XMIN=XN*X(I) SCAL049
C
C DETERMINATION OF SCALED MAXIMUM SCAL050
C
C 13 NSCALE=XMAX/X(I)+1. SCAL051
C XN=NSCALE SCAL052
C XMAX=XN*X(I) SCAL053
C
C LENGTH OF SCALE AXIS CALCULATED AT THIS POINT SCAL054
C
C XLIN=(XMAX-XMIN)/X(I) SCAL055
C IF (XLIN-XLMAX)14,18,17 SCAL056
C 14 IF (XLIN-XLMIN)15,18,18 SCAL057
C 15 IF (I-1)35,16,16 SCAL058
C 16 XM=XM*1.E+01 SCAL059
C IJ=IJ+I SCAL060
C IF (IJ-4)7,7,23 SCAL061
C 17 IF (I-6)21,177,35 SCAL062
C 177 XM=XM*1.E-01 SCAL063
C IJ=IJ+1 SCAL064
C IF (IJ-4)7,7,23 SCAL065
C 18 IF (XSCALE)19,20,35 SCAL066
C 19 IF (ABS (XSCALE)-X(I))22,20,22 SCAL067
C 20 XSCALE=X(I) SCAL068
C GO TO 39 SCAL069
C 21 CONTINUE SCAL070
C 22 XSCALE=ABS (XSCALE) SCAL071
C XMIN=SXMIN SCAL072
C XMAX=SXMAX SCAL073
C GO TO 1 SCAL074
C 23 IF (XSCALE)38,36,35 SCAL075
C
C FIXED SCALING SCAL076
C
C 24 DO 25 I=1,6 SCAL077
C IF (XSCALE-X(I))25,26,25 SCAL078
C 25 CONTINUE SCAL079
C GO TO 37 SCAL080
C
C DETERMINATION OF SCALE MINIMUM SCAL081
C
C 26 NSCALE=XMIN/XSCALE SCAL082
C IF (NSCALE)28,27,30 SCAL083
C 27 IF (XMIN)28,31,29 SCAL084

```

```

28 XN=NSCALE-1          SCAL097
  XMIN=XN*XSCALE        SCAL098
  GO TO 31              SCAL099
29 XMIN=0.              SCAL100
  GO TO 31              SCAL101
30 XN=NSCALE            SCAL102
  XMIN=XN*XSCALE        SCAL103
C
C      DETERMINATION OF SCALED MAXIMUM      SCAL104
C
31 NSCALE=XMAX/XSCALE+1.  SCAL105
  XN=NSCALE            SCAL106
  XMAX=XN*XSCALE        SCAL107
C
C      LENGTH OF SCALE AXIS      SCAL108
C
  XLIN=(XMAX-XMIN)/XSCALE  SCAL109
  IF(XLIN-XLMAX)32,39,33   SCAL110
32 IF(XLIN-XLMIN)34,39,39  SCAL111
33 XSCALE=XSCALE*1.E+01    SCAL112
  IJ=IJ+1                SCAL113
  IF(IJ-4)2,2,37          SCAL114
34 XSCALE=XSCALE*1.E-01    SCAL115
  IJ=IJ+1                SCAL116
  IF(IJ-4)2,2,37          SCAL117
35 STOP                  SCAL118
36 WRITE(6,200)XMAX,XMIN,XLIN,(X(I),I=1,6)  SCAL119
  GO TO 35              SCAL120
37 WRITE(6,201)XMAX,XMIN,XLIN,(X(I),I=1,6)  SCAL121
  GO TO 35              SCAL122
38 WRITE(6,202)XMAX,XMIN,XLIN,(X(I),I=1,6)  SCAL123
  GO TO 22              SCAL124
200 FORMAT(42X,36HAUTOMATIC SCALING CANNOT BE ACHIEVED/30X,5HXMAX=1E14  SCAL125
  1.5,2X,5HXMIN=1E14.5,2X,5HXLIN=1E14.5//51X,17HSCALE FACTORS ARE//18  SCAL126
  2X,6E14.5//)           SCAL127
201 FORMAT(45X,32HFIXED SCALING CANNOT BE ACHIEVED/30X,5HXMAX=1E14.5,2  SCAL128
  1X,5HXMIN=1E14.5,2X,5HXLIN=1E14.5//51X,17HSCALE FACTORS ARE//18X,6E  SCAL129
  214.5//)               SCAL130
202 FORMAT(9X,76HAUTOMATIC SCALING CANNOT ACHIEVE DESIRED SCALE FACTOR  SCAL131
  1, WILL TRY FIXED SCALING//30X,5HXMAX=1E14.5,2X,5HXMIN=1E14.5,2X,5HX  SCAL132
  2LIN=1E14.5//51X,17HSCALE FACTORS ARE//18X,6E14.5//)               SCAL133
39 RETURN                SCAL134
  END                   SCAL135

```

```

C      ***** PTV PROGRAM *****
C
C      SUBROUTINE PTV(TIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT,
1 T,V,VS)
C
C      THIS SUBPROGRAM CONTROLS THE ITERATION FOR THE PEAK
C      TRANSLATIONAL VELOCITY, PTV. IT IS THE ONLY SUBROUTINE OF THE
C      PTV PROGRAM WHICH IS CALLED FROM THE MAIN PROGRAM.
C
C      DIMENSION QX(1000),QY(1000),IS(2)
C      DIMENSION G(6)
C      DIMENSION A(50),C(50)
C      COMMON /QXY/QX,QY
C      COMMON /GIS/IS
C
C      IF(OPTION.GT.0.) GO TO 10
C      WRITE(6,580)
C      WRITE(6,600) TIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT
C      WRITE(6,590)
C
C      74.21457 IS A UNITS CONVERSION FACTOR
10 VC=2.*74.21457/RHOW/RAD
N=PTS
T=T4
DT=(T5-T)/FLOAT(N-1)
IF(T.LE.0.) N=N-1
IF(T.LE.0.) T=DT/2.
IS(1)=2
IS(2)=1
G(2)=CWAT/RAD
G(3)=TIMER2
G(5)=SQRT(TIMER2-T3)
G(6)=SQRT(TIMER2)
C
C      INITIAL SEARCH FOR MAXIMUM VELOCITY
DO 40 I=1,N
G(1)=T
V=VC*FV(G)
A(I)=T
C(I)=V
VS=2.*COSA*V
IF(OPTION.LE.0.) WRITE(6,610) T,V,VS
T=T+DT
40 CONTINUE
C
C      ITERATION FOR PTV
C      DETERMINE THE MAXIMUM VELOCITY FROM C ARRAY
CALL XMAX(C,N,M,M1)
A2=A(M1)
C2=C(M1)
A(1)=A(M)
C(1)=C(M)
A(2)=A2
C(2)=C2
DA=DT
T=A(1)-1.8*DA
IF(T.LE.0.) T=DA/5.
DT=DA/2.
DO 45 I=3,10
G(1)=T
V=VC*FV(G)
A(I)=T
C(I)=V
VS=2.*COSA*V
IF(OPTION.LE.0.) WRITE(6,610) T,V,VS

```

```

T=T+DT
45 CONTINUE
N=10
IF (IABS(M-M1).LT.3) GO TO 55
T=A(2)-0.8*DA
IF (T.LE.0.) T=DA/5.
DT=DA/3.
DO 50 I=11,16
G(1)=T
V=VC*FV(G)
A(I)=T
C(I)=V
VS=2.*COSA*V
IF (OPTION.LE.0.) WRITE(6,610) T,V,VS
T=T+DT
50 CONTINUE
N=16
55 CONTINUE
DO 75 JJ=1,6
CALL XMAX(C,N,M,M1)
IF (JJ.LT.3) GO TO 62
IF (ABS((C(M)-C(M1))/C(M)).LT.0.001) GO TO 110
IF (JJ.EQ.6) GO TO 120
62 N=10
T1=A(M)
T2=A(M1)
V1=C(M)
V2=C(M1)
A(9)=T1
A(10)=T2
C(9)=V1
C(10)=V2
DT=ABS(T1-T2)/5.
II=1
DO 70 I=1,8
T=T1+DT*FLOAT((I-10)/2*II)
IF (T.LE.0.0) GO TO 64
G(1)=T
V=VC*FV(G)
VS=2.*COSA*V
GO TO 66
C WHEN T IS LESS THAN ZERO SET TO ZERO.
64 T = 0.0
V = 0.0
66 IF (OPTION.LE.0.0) WRITE (6,610) T,V,VS
A(I)=T
C(I)=V
II=-1*II
70 CONTINUE
75 CONTINUE
110 V=C(M)
T=A(M)
VS=2.*COSA*C(M)
IF (OPTION.LE.0.) WRITE(6,620) A(M),C(M),VS
RETURN
120 V=C(M)
T=A(M)
VS=2.*COSA*C(M)
VS1=2.*COSA*C(M1)
WRITE(6,630) T,V,VS,A(M1),C(M1),VS1
RETURN
C
C
580 FORMAT(1H1,10X,30HTRANSLATIONAL VELOCITY PROGRAM )

```

```

590 FCRRMAT(1H0,5X,45HITERATION FOR PEAK TRANSLATIONAL VELOCITY PTV //  

  1 12X,9HTIME (SEC) ,8X,16HVELOCITY(FT/SEC) ,3X,25HVERTICAL VELOCITY(IF  

  2T/SEC) /29X,16HTARGET SUBMERGED,7X,17HTARGET AT SURFACE )  

600 FORMAT(1H0,5X,23HINPUT TO SUBROUTINE PTV // 10X,  

  1 45HTIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT //1P5E14.5/  

  2 1P5E14.5 )  

610 FORMAT(1P3E22.6)  

620 FORMAT(1H0,6X,20H***** ,9X,3HPTV,19X,3HPTV//  

  1 1P3E22.6)  

630 FORMAT(1H0,42H*** WARNING ITERATION DID NOT CONVERGE *** ,5X,  

  1 35HMAXIMUM AND NEAREST VALUE ARE GIVEN //  

  1 12X,9HTIME (SEC) ,8X,16HVELOCITY(FT/SEC) ,3X,25HVERTICAL VELOCITY(IF  

  2T/SEC) /29X,16HTARGET SUBMERGED,7X,17HTARGET AT SURFACE /  

  3 (1P3E22.6))

```

```

C
  END

```

FUNCTION FV(G)

```

C
C  THIS SUBPROGRAM SETS UP THE INTEGRATION FOR
C  THE TRANSLATIONAL VELOCITY V
C
C  DIMENSION G(6)
C  EXTERNAL F1
C  DATA N/18/
C
C  NN=FLOAT(N)*G(1)*G(2)/8.
C  NN=MAX0(NN,8)
C  NN=MIN0(NN,N)
C  X=G(1)-8./G(2)
C  IF(X,GT.,G(3)) GO TO 43
C  Z1=G(6)
C  IF(X,GT.,0.) Z1=SQRT(G(3)-X)
C  IF(G(1),GT.,G(3)) GO TO 40
C  G(4)=-1.0
C  Z2=SQRT(G(3)-G(1))
C  INTEGRATION FOR T .LE. TIMER2
C  FV=-FGI(Z1,Z2,NN,F1,G)
C  RETURN
40 Z2=0.
C  Z3=SQRT(G(1)-G(3))
C  IF(G(3),EQ.0.) GO TO 45
C  G(4)=-1.0
C  N1=Z1/(Z1+Z3)*FLOAT(NN)+2.0
C  NNN=Z3/(Z1+Z3)*FLOAT(NN)+2.0
C  INTEGRATION FOR INTERVAL WHICH INCLUDES TIMER2
C  V1=-FGI(Z1,Z2,N1,F1,G)
C  G(4)=1.0
C  V2=FGI(Z2,Z3,NNN,F1,G)
C  FV=V1+V2
C  RETURN
43 Z2=SQRT(X-G(3))
C  Z3=SQRT(G(1)-G(3))
C  G(4)=1.0
C  INTEGRATION FOR T LARGER THAN TIMER2 BUT THE
C  INTERVAL DOES NOT INCLUDE TIMER2.
C  FV=FGI(Z2,Z3,NN,F1,G)
C  RETURN
C

```

```

FUNCTION F1(Z,G)
C
C THIS SUBPROGRAM CALCULATES THE PRODUCT INCIDENT PRESSURE *
C REDUCED STEP WAVE ACCELERATION BY CALLING THE INTERPOLATION
C PROGRAMS VTAB AND PTAB.
C
C DIMENSION QX(1000),QY(1000),IS(2)
C DIMENSION G(6),QQX(120),QQY(120)
C COMMON /QXY/QX,QY
C COMMON /QIS/IS
C
C REDUCED STEP WAVE ACCELERATION OF A CYLINDER
C
DATA (QQX(I),I=1,106) /0.,.0125,.025,.0375,.050,.075,.100,
1 .125,.150,.175,.200,.225,.250,.275,.300,.325,.350,.375,
2 .4000,.425,.450,.475,.500,.525,.550,.575,.600,.625,.650,
3 .675,.700,.725,.750,.775,.800,.825,.850,.875,.900,.925,.950,
4 .975,1.00,1.05,1.10,1.15,1.20,1.25,1.30,1.35,1.40,1.45,
5 1.50,1.55,1.60,1.65,1.70,1.75,1.80,1.85,1.90,1.95,2.00,
6 2.05,2.10,2.15,2.20,2.25,2.30,2.35,2.40,2.45,2.50,2.55,
7 2.60,2.65,2.70,2.75,2.80,2.85,2.90,3.00,3.10,3.2,3.3,3.4,
8 3.5,3.6,3.7,3.8,3.9,4.0,4.2,4.4,4.6,4.8,5.0,5.25,5.50,
9 5.75,6.00,6.25,6.5,7.0,7.5,8.0 /
DATA (QQY(I),I=1,60) / 0.,.198193,.275935,.332694,.378180,
1 .448836,.502189,.544000,.577342,.604111,.625589,.642701,
2 .656143,.666457,.674079,.679365,.682612,.684070,.683955,
3 .682452,.679721,.675904,.671127,.665499,.659120,.652078,
4 .644453,.636315,.627730,.618755,.609444,.599844,.589999,
5 .579949,.569730,.559374,.548913,.538372,.527777,.517151,
6 .506515,.495887,.485284,.464215,.443417,.422977,.402968,
7 .383447,.364460,.346042,.328218,.311008,.294424,.278471,
8 .263152,.248465,.234404,.220960,.208124,.195881 /
DATA (QQY(I),I=61,106) / .184219,.173122,.162573,.152555,
1 .143051,.134041,.125509,.117435,.109801,.102590,.095782,
2 .089361,.083308,.077608,.072242,.067196,.062453,.057999,
3 .053818,.049897,.046221,.039556,.033725,.028637,.024209,
4 .020368,.017044,.014177,.011712,.009599,.007795,.006260,
5 .003863,.002172,.001009,.000230,-0.000267,-0.000619,
6 -0.000774,-0.000804,-0.000767,-0.000696,-0.000606,
7 -0.000430,-0.000297,-0.000206 /
C
IF(G(4),GT.0.) GO TO 20
X=G(3)-Z*Z
GO TO 30
20 X=G(3)+Z*Z
30 XD=-(G(1)-X)*G(2)
IF(Z,GT,G(5)) GO TO 35
P=PTAB(X,QX,QY,IS(2))
GO TO 40
35 P=VTAB(X,QX,QY,IS(2))
40 F1=Z*P*VTAB(XD,QQX,QQY,IS(1))
RETURN
END

```

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```

SUBROUTINE XMAX(B,N,M,M1)

C
C THIS SUBPROGRAM DETERMINES THE LOCATIONS OF THE TWO LARGEST
C ABSOLUTE VALUES OF MEMBERS OF THE B ARRAY.
C
C
DIMENSION B(50)
X=ABS(B(1))
M=1
DO 10 I=2,N
IF(ABS(B(I)).LT.X) GO TO 10
M=I
X=ABS(B(M))
10 CONTINUE
M1=1
IF(M.EQ.1) M1=2
X=ABS(B(M1))
DO 20 I=2,N
IF(ABS(B(I)).LT.X) GO TO 20
IF(I.EQ.M) GO TO 20
M1=I
X=ABS(B(M1))
20 CONTINUE
RETURN
END

```

```

FUNCTION VTAB(X,Y,Z,K)

C THIS SUBPROGRAM PERFORMS A SECOND ORDER LAGRANGIAN INTERPOLATION
C
C THE INDEPENDENT VARIABLE IS STORED IN THE Y ARRAY IN INCREASING
C ORDER. THE DEPENDENT VARIABLE IS STORED IN THE Z ARRAY.
C X IS THE POINT AT WHICH THE FUNCTION IS TO BE EVALUATED.
C K IS THE NUMBER OF THE ELEMENT IN THE Y ARRAY WHICH IS FIRST
C COMPARED WITH X.
C
C
DIMENSION Y(1000),Z(1000)
IF(X.LE.0.) GO TO 50
DO 10 I=K,1000
J=I
IF(Y(I).GT.X) GO TO 20
10 CONTINUE
20 J=MAX0(3,J-1)
DO 30 I=1,1000
IF(Y(J).LT.X) GO TO 40
J=J-1
IF(J.LT.3) GO TO 40
30 CONTINUE
40 K=J+1
IF(Z(J).EQ.Z(K)) GO TO 60
L=J-1
A=(X-Y(K))/(Y(J)-Y(L))
C=(X-Y(L))/(Y(K)-Y(J))
IF((A.LT.-5.0).OR.(C.GT.5.0)) GO TO 60
B=(X-Y(J))/(Y(K)-Y(L))
VTAB=C*(B*Z(K)-A*Z(J))+A*B*Z(L)
RETURN
50 VTAB=0.
RETURN
60 VTAB=Z(J)+(X-Y(J))*(Z(K)-Z(J))/(Y(K)-Y(J))
RETURN
END

```

FUNCTION PTAB(X,Y,Z,K)

C
C
C
C
CTHIS SUBPROGRAM PERFORMS A SECOND ORDER LAGRANGIAN INTERPOLATION
WITH PROVISIONS FOR HANDLING A SINGULARITY.
FUNCTION ARGUMENTS ARE THE SAME AS IN VTAB .

```

DIMENSION Y(1000),Z(1000)
IF(X.LE.0.) GO TO 50
DO 10 I=K,1000
J=I
IF(Y(I).GT.X) GO TO 20
10 CONTINUE
20 J=MAX0(3,J-1)
DO 30 I=1,1000
IF(Y(J).LT.X) GO TO 40
J=J-1
IF(J.LT.3) GO TO 40
30 CONTINUE
40 J=J+1
JJ=J

THE FOLLOWING THREE STATEMENTS PROVIDE FOR EXTRAPOLATION
AROUND A SINGULARITY.

IF(ABS(Z(J)).GT.1.0E20)JJ=J-2
IF(ABS(Z(J-1)).GT.1.0E20)JJ=J+1
IF((JJ.EQ.J).AND.(ABS(Z(J-2)).LT.1.0E20)) JJ=J-1

J=JJ
K=J+1
IF(Z(J).EQ.Z(K)) GO TO 60
L=J-1
A=(X-Y(K))/(Y(J)-Y(L))
C=(X-Y(L))/(Y(K)-Y(J))
IF((A.LT.-5.0).OR.(C.GT.5.0)) GO TO 60
B=(X-Y(J))/(Y(K)-Y(L))
PTAB=C*(B*Z(K)-A*Z(J))+A*B*Z(L)
RETURN
50 PTAB=0.
RETURN
60 PTAB=Z(J)+(X-Y(J))*(Z(K)-Z(J))/(Y(K)-Y(J))
RETURN
END

```

TABLE B.1 FULL OUTPUT FOR A SPHERICAL WAVE BOTTOM REFLECTION

BOTTOM REFLECTION RUN NUMBER 1	DATE 06/29/71
INPUT	

DEPTH OF WATER IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR RT)
VELOCITY OF SOUND IN WATER IN FT/SEC
VELOCITY OF SOUND IN BOTTOM IN FT/SEC
VELOCITY OF SWEAVE IN FT/SEC
DENSITY OF BOTTOM IN GM/CC
COEFFICIENT OF SW PRESSURE FORMULA IN PSI
EXPONENT OF SW PRESSURE FORMULA IN PSI
PRINT OUT CONTROL PARAMETER 175.GT.0. FOR SHORTER PRINT OUT
COEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS
EXPONENT OF SW TIME CONSTANT FORMULA
NUMBER OF SUBDIVISIONS OF THETA
OVERTURE (FT) THAT DIVIDES ARRIVAL IN MULTIPLES OF THETA
DESIRABLE RATIO BETWEEN INCIDENT AND CRITICAL ANGLE
SCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH)
SCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH)
SLOPE OF BOTTOM IN DEGREES
PARAMETER THAT SELLECTS THEORY
ARRIVAL TIME OF GROUND WAVE IN MICROSECONDS
PLOT CONTROL PARAMETER (Z3 = n. MEANS PLOTS APF MANTED)
CYLINDRICAL RADIUS IN FT
PRINT CONTROL PARAMETER (FULL PRINT OUT IN SUBROUTINE PTV IF APHINT.LF.0.)

CHARACTERISTIC MAGNITUDES

ANGLE OF INCIDENT WAVE IN DEGREES
VELOCITY OF STANLEY WAVE IN FT/SEC
POISSON RATIO
REDUCED TIME OF SURFACE REFLECTION
CRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES
ANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES
REFLECTION COEFFICIENT
ANGLE OF SWEAVE WAVE IN BOTTOM IN DEGREES
REDUCED TIME OF PHASESHIFT IN DEGREES
REDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE
SLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERISTIC LENGTH IN FT.
REDUCED SLANT DISTANCE (RACTU/WCH*1/3)
CHARACTERISTIC TIME RACTU/CHATEW IN SECONDS,
CHARACTERISTIC PRESSURE=PRESSURE WATER SW PEAK PRESSURE IN PSI
REDUCED TIME CONSTANT OF INCIDENT WAVE
ACTUAL SW TIME CONSTANT IN MILLISECONDS
REDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE
BOTTOM REFLECTED SW TIME CONSTANT IN MILLISECONDS

CONSTANTS OF THE CALCULATION	COSTH	SINTH
SMALLH	0.106106E+00	0.3347480E+00
D2	0.357106E+00	0.3363364E+00

BIGH= .20000000E+04
D= .70000000E+03
DGAU= .80000000E+03
SMALLR= .70000000E+04
WCH= .10000000E+02
CHATEW= .49000000E+04
CROT= .52000000E+04
CSHEAR= 0.
RJWATE= .10300000E+01
RHOBOTE= .16600000E+01
PRECOR= .43800000E+01
Z5= -0.
PREEXP= .11300000E+01
THEC0E= .22740000E-02
THEEXP= -.22200000E+01
STEPS= .40000000E+01
DURAT= .5040081E+01
TH0VAL= -0.
X1= .10000000E+03
X2= .10000000E+05
SLOPE= -0.
Z1= -0.
Z2= -0.
Z3= .10000000E+01
RADIUS= .22000000E+02
APRINT= 0.

THE= .70346201E+02
CTSON= -0.
POISR= .50000000E+00
RS= .10225972E+01
ALPHA= .70442811E+02
THEONE= .8014453E+02
CR= .863802E+00
ANGA= -0.
EE= 0.
D2R2= .10617537E+01
R2= .10617537E+01
RACTU= .70001142E+04
REDR= .3249437E+04
TACT= .14287172F+01
PACT= .47111824E+03
THETA= .20313067E-01
THE= .29021628E-02
THETR= .2052624E-01
THETR= .29066749E+02

TABLE B.1 CONTINUED

ESTATE PLANNING METHOD

01 02205490E-02 89H3802E+00 EDT

TABLE B.1 CONTINUED

TABLE B.1 CONTINUED

TRANSLATIONAL VELOCITY PROGRAM

INPUT TO SUBROUTINE PTV

TIMER2,T3,T4,T5,HAD,PTS,OPTION,COSA,RHOM,CWAT

0.	-6.30204E-03	0.	6.30204E-02	2.20000E+01
3.00000E+01	3.36336E-01	1.03000E+00	4.90000E+03	

INFLATION FOR PEAK TRANSLATIONAL VELOCITY PTV

TIME (SEC)	VELOCITY (FT/SEC)		VERTICAL VELOCITY (FT/SEC)
	TARGET	SUBMERGED	
1.086559E-03	6.525452E-01	4.389494E-01	
3.25967F-03	2.147045E+00	1.44259E+00	
5.43275F-03	2.922026E+00	1.65567E+00	
7.61591F-03	3.073522E+00	2.067475E+00	
9.77903F-03	2.847512E+00	1.94351E+00	
1.19525F-02	2.592360E+00	1.43610E+00	
1.41252F-02	2.320794E+00	1.561135E+00	
1.62983F-02	2.068548E+00	1.591456E+00	
1.84715F-02	1.831783E+00	1.232191E+00	
2.06446F-02	1.620605E+00	1.089770E+00	
2.281774F-02	1.429563E+00	9.016284E+00	
2.499086F-02	1.259587E+00	8.472898E+01	
2.716398F-02	1.1019541E+00	7.63583E+01	
2.93379E-02	9.757967E-01	6.563919E+01	
3.15102F-02	8.567997E-01	5.763459E+01	
3.368333F-02	7.51192RE-01	5.053070E+01	
3.58565F-02	6.566918E-01	4.417387E+01	
3.80295F-02	5.723656E-01	3.950148E+01	
4.02026F-02	4.972796E-01	3.345064E+01	
4.23758F-02	4.297510E-01	2.890818E+01	
4.45482F-02	3.693076E-01	2.484232E+01	
4.672204F-02	3.152316E-01	2.120477E+01	
4.889516E-02	2.666219E-01	1.793493E+01	
5.10688F-02	2.231784E-01	1.501260E+01	
5.324139F-02	1.842856E-01	1.239639E+01	
5.54145F-02	1.493711E-01	1.004779E+01	
5.75876F-02	1.162309E-01	7.953062E+00	
5.976075F-02	9.038491E-02	6.07947E+00	
6.193367F-02	6.546200E-02	4.03451E+00	
3.69410F-03	2.363885E-00	1.993640E+00	
4.78086F-03	2.767755E+00	1.590121E+00	
5.867419F-03	2.992797E+00	1.061656E+00	
6.953978E-03	3.075552E+00	2.013173E+00	
8.040537E-03	3.054987E+00	2.068840E+00	
9.127096F-03	2.963759E+00	2.055006E+00	
1.021364F-02	2.831359E+00	1.993640E+00	
1.13002F-02	2.682011E+00	1.804116E+00	
6.432430F-03	3.050738E+00	2.052148E+00	
7.475526F-03	3.076577E+00	2.069530E+00	
6.562817F-03	3.059309E+00	2.057914E+00	
7.345139E-03	3.078365E+00	2.00733E+00	
6.693204F-03	3.066342E+00	2.062645E+00	
7.214722F-03	3.078826E+00	2.071043E+00	
6.82359F-03	3.071737E+00	2.066274E+00	
7.084365F-03	3.077909E+00	2.070426E+00	
7.11043F-03	3.078206E+00	2.070625E+00	
7.319062E-03	3.078566E+00	2.070867E+00	
7.136550F-03	3.078445E+00	2.070787E+00	
7.292985F-03	3.078712E+00	2.070966E+00	

TABLE B.1 CONTINUED

7.162597E-03	3.078628E+00	2.070909E+00
7.266907E-03	3.078805E+00	2.071029E+00
7.188675E-03	3.078755E+00	2.070999E+00
7.200830E-03	3.078843E+00	2.071054E+00
*****	PTV	PTV
7.240830E-03	3.078843E+00	2.071054E+00

TABLE B-2 SHORT OUTPUT FOR A SPHERICAL WAVE BOTTOM REFLECTION

HOTTON REFLECTION
RUN NUMBER
2
INPUT

DEPTH OF WATER IN FT
DEPTH OF EXPLOSION IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
***** CALCULATION GEOMETRY CHANGED FOR SLOPING BOTTOM
DEPTH OF WATER IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT)
VELOCITY OF SOUND IN WATER IN FT/SEC
VELOCITY OF SOUND IN BOTTOM IN FT/SEC
VELOCITY OF SHEARWAVE IN FT/SEC
DENSITY OF WATER IN GM/CC
DENSITY OF BOTTOM IN GM/CC
COEFFICIENT OF SW PRESSURE FORMULA IN PSI
PRINT OUT CONTROL PARAMETER (25.GT.0 FOR SHORTER PRINT DUTY)
EXPONENT OF SW PRESSURE FORMULA
COEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS
EXPONENT OF SW TIME CONSTANT FORMULA
NUMBER OF SUBDIVISIONS OF THETA
DURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA
DESIRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE
SCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH)
SCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH)
SLOPE OF BOTTOM IN DEGREES
PARAMETER THAT SELECTS THEORY
ARRIVAL TIME OF GROUND WAVE IN MICROSECONDS
PLOT CONTROL PARAMETER (23 = 0. MEANS PLOTS ARE WANTED)
CYLINDER RADIUS IN FT
PRINT CONTROL PARAMETER (FULL PRINT OUT IN SUBROUTINE PTV IF APRINT.LE.0.)

CHARACTERISTIC MAGNITUDES

ANGLE OF INCIDENT WAVE IN DEGREES
VELOCITY OF STONLEY WAVE IN FT/SEC
POISSON RATIO
REDUCED TIME OF SURFACE REFLECTION
CRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES
CRITICAL ANGLE OF SHEARWAVE IN DEGREES
REDUCED ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR WAVE
ANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES
REFLECTION COEFFICIENT
ANGLE OF PHASESHIFT IN DEGREES
REDUCED TIME OF PRECURSOR ARRIVAL
REDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE
SLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERISTIC LENGTH IN FT.
REDUCED SLANT DISTANCE (RACTUWCH*1/3)
CHARACTERISTIC TIME=RACTU/CWATER IN SECONDS
CHARACTERISTIC PRESSURE=FREE WATER SW PEAK PRESSURE IN PSI
REDUCED TIME CONSTANT OF INCIDENT WAVE
ACTUAL SW TIME CONSTANT IN MILLISECONDS
REDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE
BOTTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS
CONSTANTS OF THE CALCULATION
SMALL DEZERO 02 COSAL COSTH SINTH
3298469E+01 -.318198E+01 .857812E+00 .9642855E+00 .6190160E+00 .5737678E+00

TABLE B.2 CONTINUED

ROSFNAIM METHOD
RIGHT BOTTOM WITH FAST SHEAR WAVE
DT EOT
•1.052095E-02 •9537487E+00

ARRIVAL TIME	POINT	ENERGY FLUX	PI	TIME SECNDS	PRESSURE PSI	IMPULSE
T						
0.	0.	0.	0.	-1109371E-04	0.	2497786E+03
•9463454E+00	•2659025E+01	•2019874E-08	0.	•3284164E-05	•2459025E+01	•2814175E-04
•9884496E+00	•3420554E+01	•2019874E-08	0.	•7674613E-05	•320556E+01	•8010188E-05
•9905434E+00	•2812631E+00	•2019874E-08	0.	•4565602E-05	•3393006E+01	•8010188E-05
•9926474E+00	•3035064E+01	•5402794E-08	0.	•425512E-05	•3035006E+01	•1942298E-04
•9947421E+00	•2553954E+01	•5402794E-08	0.	•2555961E-05	•2553975E+01	•1942298E-04
•9964464E+00	•2049203E+01	•7398414E-08	0.	•03364102E-06	•2049203E+01	•2814175E-04
ARRIVAL OF DIRECT WAVE P = .2624485E-03						
•1001175E+01	•1569338E+01	•7398414L-08	•2482092E+03	•8731405E-06	•2497786E+03	•2814175E-04
•1003179E+01	•1138297E+01	•1521177E-04	•2255710E+03	•2882691E-05	•2267093E+03	•7278453E-03
•1005179E+01	•766901E+00	•1521177E-04	•4922414E+03	•7674645E+03	•7278453E-03	•7278453E-03
•1007179E+01	•4616681E+00	•2812631E+00	•1863005E+03	•6001793E-05	•1867607E+03	•1432481E-02
•1009491E+01	•2194444E+00	•2812631E+00	•1693088E+03	•7711343E-05	•1695282E+03	•1432481E-02
•1011454E+01	•1026314E+01	•3689284E-04	•1736626E+03	•1736626E+03	•2013036E-02	•2013036E-02
•1013700E+01	•5123134E+01	•3689284E-04	•1498333E+03	•1113044E-04	•1397717E-03	•2491698E-02
•1015344E+01	•9948436E+01	•4281077E-04	•1270746E+03	•1284000E-04	•1269801E-03	•2491698E-02
•1017304E+01	•5501643E+01	•4281077E-04	•1154891E+03	•1454955E-04	•1154241E+03	•2491698E-02
•1019179E+01	•1620172E+01	•4691592E-04	•1625193E+03	•1625193E+03	•2886992E-02	•2886992E-02
•1022179E+01	•2505162E+00	•95348320E+02	•1796665E-04	•9563384E+02	•9563384E+02	•2886992E-02
•1024219E+01	•558814E+00	•4971565E-04	•4681367E+02	•1967820E-04	•8724250E-02	•3214532E-02
•1026314E+01	•9903794E+01	•4971565E-04	•7977637E+02	•7977637E+02	•3214532E-02	•3214532E-02
•1028414E+01	•1911349E+01	•5124669E-04	•759260E+02	•2209730E-04	•7320409E+02	•3487805E+02
•1030245E+01	•2464014E+01	•5164494E-04	•6516292E+02	•2400485E-04	•6752694E+02	•3487805E+02
•1032449E+01	•3744911E+01	•5204444E-04	•5912878E+02	•5912878E+02	•3719227E-02	•3719227E-02
•1034472E+01	•4204104E+01	•5411946E-04	•5271558E+02	•2822595E-04	•5917679E+02	•392404E+02
•1036466E+01	•1300464E+02	•5410966E-04	•4403484E+02	•2933550E-04	•5703784E+02	•392404E+02
•1038459E+01	•2464424E+02	•5515045E-04	•4410400E+02	•3164505E-04	•5738554E+02	•392404E+02
•1041154E+01	•5510104E+01	•5510104E+01	•4013300E+02	•3353156E-04	•667842E+02	•412569E+02
•1041172E+01	•4262691E+02	•5510104E+01	•39353594E+02	•341570E-04	•620585E-02	•412569E+02
•1041194E+01	•4261258E+02	•5529314E-04	•38944278E+02	•3430830E-04	•7215536E+02	•4146718E+02
•1041272E+01	•3663349E+02	•5529314E-04	•3915349E+02	•3430830E-04	•7578708E+02	•4146718E+02
•1041286E+01	•4140145E+02	•5544801E-04	•3H16804E+02	•34308200E-04	•8016949E+02	•4173568E+02
•1042147E+01	•4659329E+02	•5544801E-04	•3H3H681E+02	•34233885E-04	•8497967E+02	•4173568E+02
•1043454E+01	•5152624E+02	•5540454E-04	•3H17404E+02	•3451995E-04	•1223517E+03	•4275648E+02
•1043616E+01	•4282314E+02	•5574456E-04	•3763430E+02	•3459255E-04	•1294394E+03	•4318904E+02
•1042579E+01	•4143745E+02	•5593266E-04	•3726390E+02	•3476940E-04	•1016399E+03	•4237497E+02
•1042719E+01	•7075123E+02	•5593266E-04	•3489695E+02	•3494625E-04	•1076502E+03	•4237497E+02
•1043231E+01	•7839704E+02	•56345794E-04	•34533371E+02	•3512310E-04	•1149307E+03	•4275648E+02
•1043454E+01	•4611746E+02	•563574E-04	•3H17404E+02	•3451995E-04	•1223517E+03	•4275648E+02
•1043616E+01	•4362152E+02	•5682744E-04	•3548792E+02	•34547680E-04	•1294394E+03	•4318904E+02
•1042579E+01	•4111254E+02	•5684744E-04	•3465314E+02	•3465314E+02	•1365778E+02	•4318904E+02
•1042719E+01	•1077041E+03	•5741154E-04	•3516116E+02	•3483050E-04	•1428202E+03	•4367159E+02
•1043231E+01	•1127494E+03	•5741154E-04	•347705F+02	•3600735E-04	•1475601E+03	•4367159E+02
•1043454E+01	•1044315E+03	•5804849E-04	•3428154E+02	•3468449E+02	•1487195E+03	•4419140E+02
•1043616E+01	•11039294E+03	•5804849E-04	•3631605E+02	•3631605E+02	•1449190E+03	•4419140E+02

NOT REPRODUCIBLE

TABLE B.2 CONTINUED

ARRIVAL OF GRUNNEN VF. PEAK (SINGULARITY)		ARRIVAL OF GRUNNEN VF. PEAK (SINGULARITY)	
*1.1744E-01	*1.0166E-03	*5H7391E-04	*365379E-02
*1.194519E+01	*4677902E+02	*3375362E+02	*365379E-04
*1.04500E+01	*5915E-04	*3342132E+02	*367147E-04
*1.485678E+02	*5915E-04	*3309230E+02	*368916E-04
*1.045425E+01	*5915E-04	*3309230E+02	*368916E-04
*1.045439E+01	*5925167E-04	*326439E+02	*372431E-04
*1.146141E+01	*5935067E-04	*321245E+02	*374224E-04
*1.1956238E+02	*5935067E-04	*3140828E+02	*375990E-04
*1.2170717E+02	*5936165E-04	*3149511E+02	*377886E-04
*1.92127He+01	*5936165E-04	*3118508E+02	*379527E-04
*1.09752E+03	*5935433E-04	*3118508E+02	*379527E-04
*1.146714E+01	*5935433E-04	*3087400E+02	*3803041E-04
*1.04745E+01	*5936165E-04	*3087400E+02	*3803041E-04
*1.1946472E+03	*5934701E-04	*3087400E+02	*3803041E-04
*1.2797102E+03	*5934701E-04	*3087400E+02	*3803041E-04
*****		*****	
*1.0147147E+01	*5934694E-04	*3127309E+02	*3848356E-04
*1.04733E+01	*5934694E-04	*311427E+02	*3853668E-04
*1.04733E+01	*5934694E-04	*299133E+02	*386945E-04
*1.047433E+01	*5934694E-04	*295571E+02	*389164E-04
*1.044154E+01	*594293E-04	*2920662E+02	*391243E-04
*1.170799E+02	*594293E-04	*2820662E+02	*391243E-04
*1.1148720E+02	*594293E-04	*2851661E+02	*3955172E-04
*1.048425E+01	*5967051E-04	*2851661E+02	*3955172E-04
*1.048425E+01	*5967051E-04	*2411772E+02	*3976542E-04
*1.048425E+01	*5967051E-04	*2784241E+02	*3992911E-04
*1.048425E+01	*6002424E-04	*2751200E+02	*4019291E-04
*1.048425E+01	*6002424E-04	*2751200E+02	*4019291E-04
*1.048425E+01	*6045614E-04	*2718510E+02	*4040650E-04
*1.048425E+01	*6045614E-04	*2686210E+02	*4062590E-04
*1.048425E+01	*6094269E-04	*2645240E+02	*4083389E-04
*1.050281E+01	*6094269E-04	*2622738E+02	*4104798E-04
*1.050281E+01	*6094269E-04	*2591571E+02	*4126128E-04
*1.050281E+01	*6146747E-04	*2560745E+02	*4147497E-04
*1.051142E+01	*6146747E-04	*2530343E+02	*416886E-04
*1.051142E+01	*6201163E-04	*2530343E+02	*416886E-04
*1.051142E+01	*6201163E-04	*2500271E+02	*4190266E-04
*1.051142E+01	*6201163E-04	*2470562E+02	*4240562E+02
*1.051142E+01	*6258734E-04	*2245232E+02	*4382560E-04
*1.05647E+01	*6720299E-04	*2040453E+02	*4553515E-04
*1.05647E+01	*6720299E-04	*1149527E+02	*557926E-04
*1.058151E+01	*6720299E-04	*1044683E+02	*5750052E-04
*1.060275E+01	*7125532E-04	*1695224E+02	*499525E-04
*1.062454E+01	*7125532E-04	*1531521E+02	*506638E-04
*1.0644643E+01	*7451497E-04	*1391831E+02	*5247336E-04
*1.0665659E+01	*7451497E-04	*1264893E+02	*540892E-04
*1.068425E+01	*7705043E-04	*1149527E+02	*557926E-04
*1.070774E+01	*7705043E-04	*7790464E+02	*5750052E-04
*1.072281E+01	*7899123E-04	*9494015E+01	*592156E-04
*1.074384E+01	*8242194E-04	*8426102E+01	*609211E-04
*1.077445E+01	*8045466E-04	*8305904E+01	*741161E+01
*1.0791914E+01	*8045466E-04	*8305904E+01	*741161E+01
*1.0828203E+01	*8353547E+02	*8157954E+01	*745951E+01
*1.0919149E+01	*8353547E+02	*8354200E+01	*7630706E+01
*1.093222E+01	*8353547E+02	*8354200E+01	*7630706E+01
*1.096224E+01	*8354200E+01	*8315655E+01	*7801662E+01
*1.098130E+01	*8354200E+01	*8315655E+01	*7801662E+01
*1.100735E+01	*8390935E+01	*8390935E+01	*8113221E+01
*1.102075E+01	*8390935E+01	*2738421E+01	*8113572E+01
*1.102353E+01	*8414527E+01	*2488660E+01	*8314527E+01
*1.102735E+01	*8414527E+01	*2488660E+01	*8314527E+01
*1.104443E+01	*84306129E+01	*2488660E+01	*84306129E+01

TABLE B.3 OUTPUT FOR A PLANE WAVE BOTTOM REFLECTION

	BOTTOM REFLECTION RUN NUMBER INPUT	DATE 06/29/71	
DEPTH OF WATER IN FT			
DEPTH OF EXPLOSION IN FT			
DEPTH OF GAUGE IN FT			
DEPTH OF GAUGE IN FT			
GEOMETRY CHANGED SO THAT ARRIVAL TIME OF GROUNOWAVE IS Z2 = .1350000E+02			
DEPTH OF EXPLOSION IN FT			
DEPTH OF GAUGE IN FT			
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT			
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT)			
VELOCITY OF SOUND IN WATER IN FT/SEC			
VELOCITY OF SOUND IN BOTTOM IN FT/SEC			
VELOCITY OF SHEARWAVE IN FT/SEC			
VELOCITY OF WATER IN GM/CC			
DENSITY OF BOTTOM IN GM/CC			
COEFFICIENT OF SW PRESSURE FORMULA IN PSI			
PRINT OUT CONTROL PARAMETER (75,GT,0. FOR SHORTER PRINT OUT)			
EXPONENT OF SW PRESSURE FORMULA			
COEFFICIENT OF SW PRESSURE FORMULA			
COEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS			
EXPOENT OF SW TIME CONSTANT FORMULA			
NUMBER OF SUBDIVISIONS OF THETA			
DURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA			
DESIRRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE			
SCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH)			
SCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH)			
SLOPE OF ROTATION IN DEGREES			
PARAMETER THAT SELECTS THEORY			
APPAL TIME OF GROUND WAVE IN MICROSECONDS			
PLOT CONTROL PARAMETER (Z3 = 0. MEANS PLOTS ARE WANTED)			
CYLINDER RADIUS IN FT			
PRINT CONTROL PARAMETER (FULL PRINT OUT IN SUBROUTINE PTV IF APRINT.LE.0.)			
CHARACTERISTIC MAGNITUDES			
ANGLE OF INCIDENT WAVE IN DEGREES			
VELOCITY OF STONLEY WAVE IN FT/SEC			
POISSON RATIO			
REDUCED TIME OF SURFACE REFLECTION			
Critical angle of compression wave in degrees			
Angle of pressure wave in bottom in degrees			
Reflection coefficient			
Angle of shearwave in bottom in degrees			
Angle of phaseshift in degrees			
Reduced time of precursor arrival			
Reduced time of peak of bottom reflected wave			
Slant distance between charge and gauge=characteristic length in ft.			
Reduced slant distance (RACTU/WCH*1/3)			
Characteristic time=RACTU/WATER IN SECONDS			
Characteristic time=RACTU/WATER IN SECONDS			
Characteristic pressure=free water sw peak pressure in psi			
Reduced time constant of incident wave			
Actual sw time constant in milliseconds			
Reduced time constant of bottom reflected wave			
Bottom reflected wave time constant in milliseconds			
CONSTANTS OF THE CALCULATION			
SMALLH	DEZERO	02	COSTH
			SINTH
•1167134E+01	-•1050868E+01	•3133967E+00	•5582754E+00
			•308447E+00
			•9512423E+00

TABLE B.3 CONTINUED

REDUCED TIME <i>t</i>	BOTTOM REFLECTION P-BOT	PLANE WAVE APPROXIMATION ARONS-YENNIE APPROACH NON-RIGID BOTTOMS			TOTAL PRESSURE P (PSI)
		SHOCKWAVE P-D	SURFACE REFLECTION P-S	TIME SECONDS	
*9768287E-00	*3569761E+02	0	0	-0.1948450E-04	*3559761E+02
*9794437E+00	*3741375E+02	0	0	-0.172855E-04	*3741375E+02
*9b2059E+00	*394024E+02	0	0	-0.150652E-04	*394024E+02
*9b46739E+00	*411822E+02	0	0	-0.128876E-04	*411822E+02
*9b72889E+00	*4430085E+02	0	0	-0.106886E-04	*4430085E+02
*9b9939E+00	*475831E+02	0	0	-0.848966E-05	*4725831E+02
*9b5190E+00	*50854E+02	0	0	-0.629072E-05	*506854E+02
*945149E+00	*541418E+02	0	0	-0.409173E-05	*547148E+02
*977791E+00	*593483E+02	0	0	-0.189276E-05	*5953483E+02
*1000000E+01	*64532371E+02	*25244175E+03	0	0	*31691754E+03
*1002615E+01	*711971E+02	*2292158E+03	0	0	*2198967E-05
*1005210E+01	*819632E+02	*2081269E+03	0	0	*439793E-05
*1007845E+01	*945649E+02	*1889783E+03	0	0	*65690E-05
*101060E+01	*1131444E+03	*1715914E+03	0	0	*879586E-05
*1013075E+01	*149189E+03	*1550042E+03	0	0	*109948E-04
*1013598E+01	*1618262E+03	*1528255E+03	0	0	*114654E-04
*101421E+01	*170567E+03	*1499037E+03	0	0	*118744E-04
*1014644E+01	*197137E+03	*1470379E+03	0	0	*123142E-04
*1015167E+01	*2293034E+03	*1442268E+03	0	0	*127540E-04
*1015890E+01	*293076E+03	*1416694E+03	0	0	*131938E-04
*1016554E+01	*2479390E+03	*1395609E+03	0	0	*135000E-04
*1016577E+01	*306459E+03	*1369124E+03	0	0	*139397E-04
*101710E+01	*2459064E+03	*1342949E+03	0	0	*147959E-04
*1017623E+01	*212422E+03	*1317274E+03	0	0	*148193E-04
*1018146E+01	*162524E+03	*1292090E+03	0	0	*152591E-04
*1018659E+01	*203590E+03	*1267388E+03	0	0	*158971E-04
*1019192E+01	*1466344E+03	*1243158E+03	0	0	*161387E-04
*1019716E+01	*1320582E+03	*1219391E+03	0	0	*165785E-04
*1020239E+01	*1113354E+03	*1196079E+03	0	0	*238943E+03
*1022284E+01	*7229964E+02	*1086034E+03	0	0	*192173E-04
*1025469E+01	*422667E+02	*9861135E+02	0	0	*214162E-04
*1028084E+01	*203590E+02	*853865E+02	0	0	*236152E-04
*1030399E+01	*403798E+01	*8130067E+02	0	0	*256142E-04
*103334E+01	*3868950E+02	*4556126E+02	0	0	*657176E+02
*1049004E+01	*4149596E+02	*4136941E+02	0	0	*412069E+02
*1035929E+01	*4330813E+02	*3756323E+02	0	0	*434054E+02
*1038544E+01	*254457E+02	*6086182E+02	0	0	*574488E+01
*1041159E+01	*3120062E+02	*526225E+02	0	0	*104488E+02
*1043774E+01	*352645E+02	*5017786E+02	0	0	*4560492E+02
*1046389E+01	*45551E+02	*2811990E+02	0	0	*4780388E+02
*104904E+01	*4588286E+02	*2552274E+02	0	0	*500285E+02
*1051619E+01	*4582262E+02	*2318361E+02	0	0	*522018E+02
*10538237E+01	*4538237F+02	*205061E+02	0	0	*544007E+02
*1062019E+01	*4498166E+02	*1911385E+02	0	0	*5659975E+02
*1069924E+01	*4498166E+02	*1911385E+02	0	0	*587987E+02

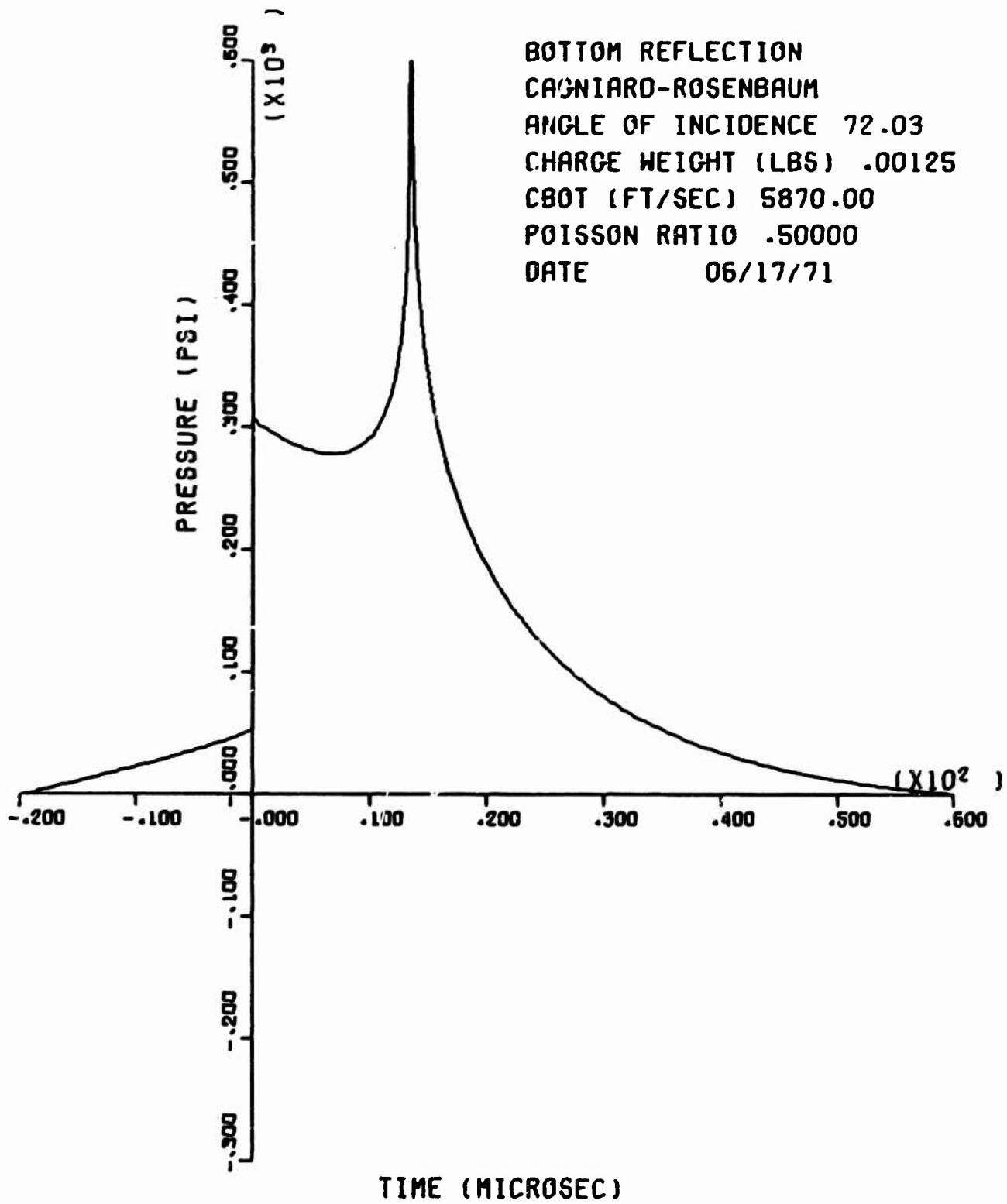


FIG. 2 SAMPLE CALCOMP PLOT OF PRESSURE-TIME HISTORY

APPENDIX C

TWO SPECIAL INPUT OPTIONS

Two special input options for altering the input geometry are provided in the code BOTREF using the variables THOVAL and Z2. The first option allows the programmer to specify an incident angle θ which is expressed as the ratio of the incident angle θ to the critical angle θ_{cr} . This is accomplished by setting THOVAL = $\theta/\theta_{cr} > 0$. If THOVAL ≤ 0 , this option is ignored. The programmer also must supply the water depth H, the horizontal range r, and the gauge depth d_g . If possible, the code calculates the required charge depth d keeping d_g fixed. Otherwise, d is set to H, and a new value of d_g is determined. Thus this option permits the user to calculate bottom reflections for a range of incident angles without first having to determine the exact geometry that is required.

The second option using Z2 > 0 provides an alternative means of specifying the reflection geometry. The arrival time Z2 (in microseconds) of the bottom reflected wave after the direct wave is exceedingly sensitive to the geometry which often cannot be measured with the necessary accuracy. This time, which can be accurately measured, provides the means which can be used to correct the input geometry. The geometry is changed by altering d and d_g and holding r fixed so that the incident angle θ and the bottom reflected slant range R_r are unchanged. Typically, the direct and reflected pulses are only slightly altered, but the change in their sum may be significant.